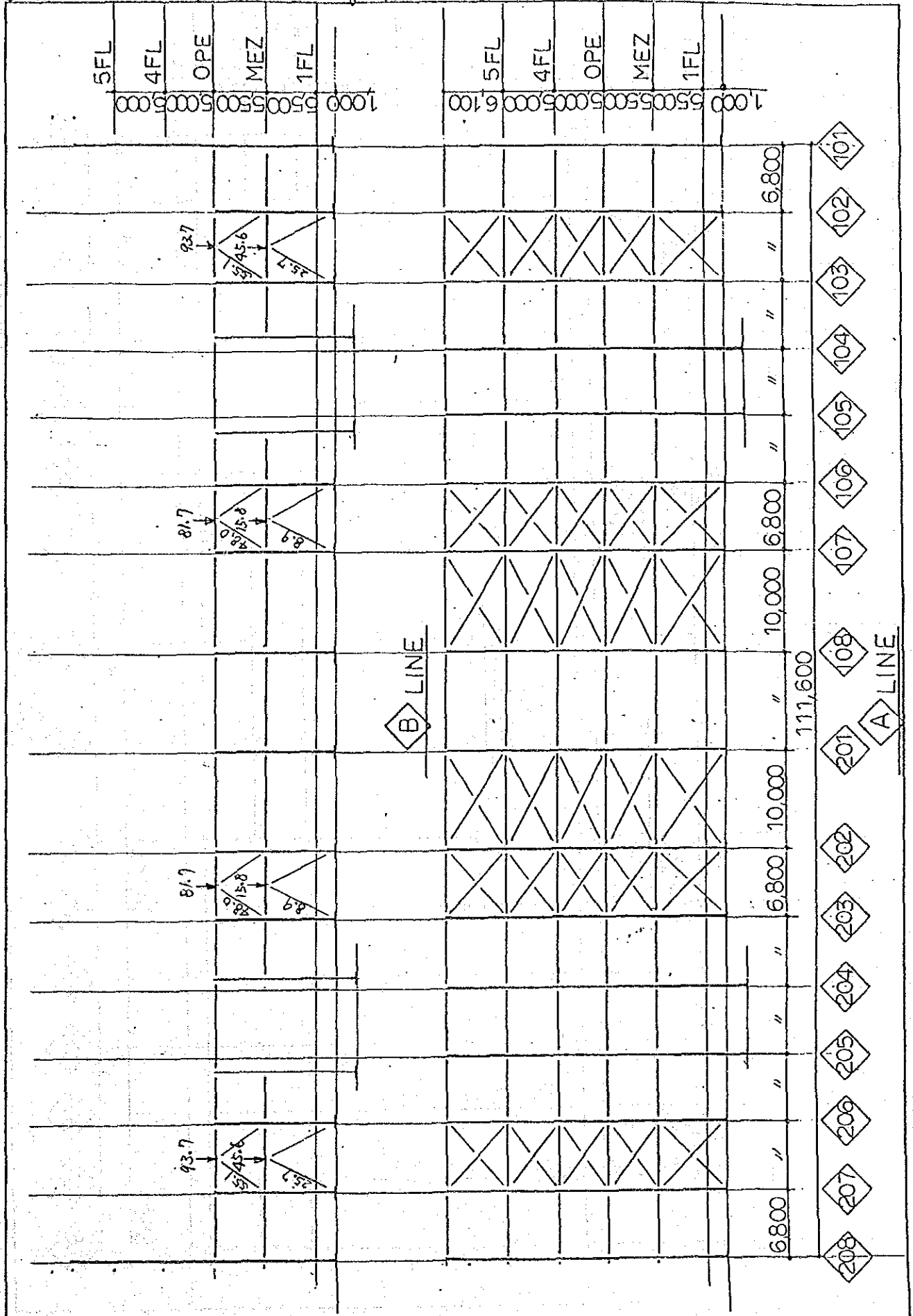


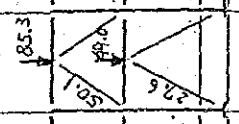
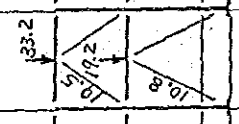
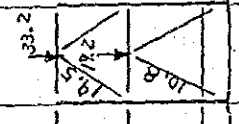
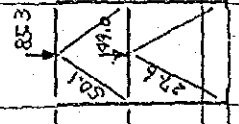
3. Structural Analysis of Frames

3.1 Calculation of vertical bracing by vertical load



5FL	1,000
4FL	550
OPE	500
MEZ	500
1FL	500

5FL	1,000
4FL	550
OPE	500
MEZ	500
1FL	500

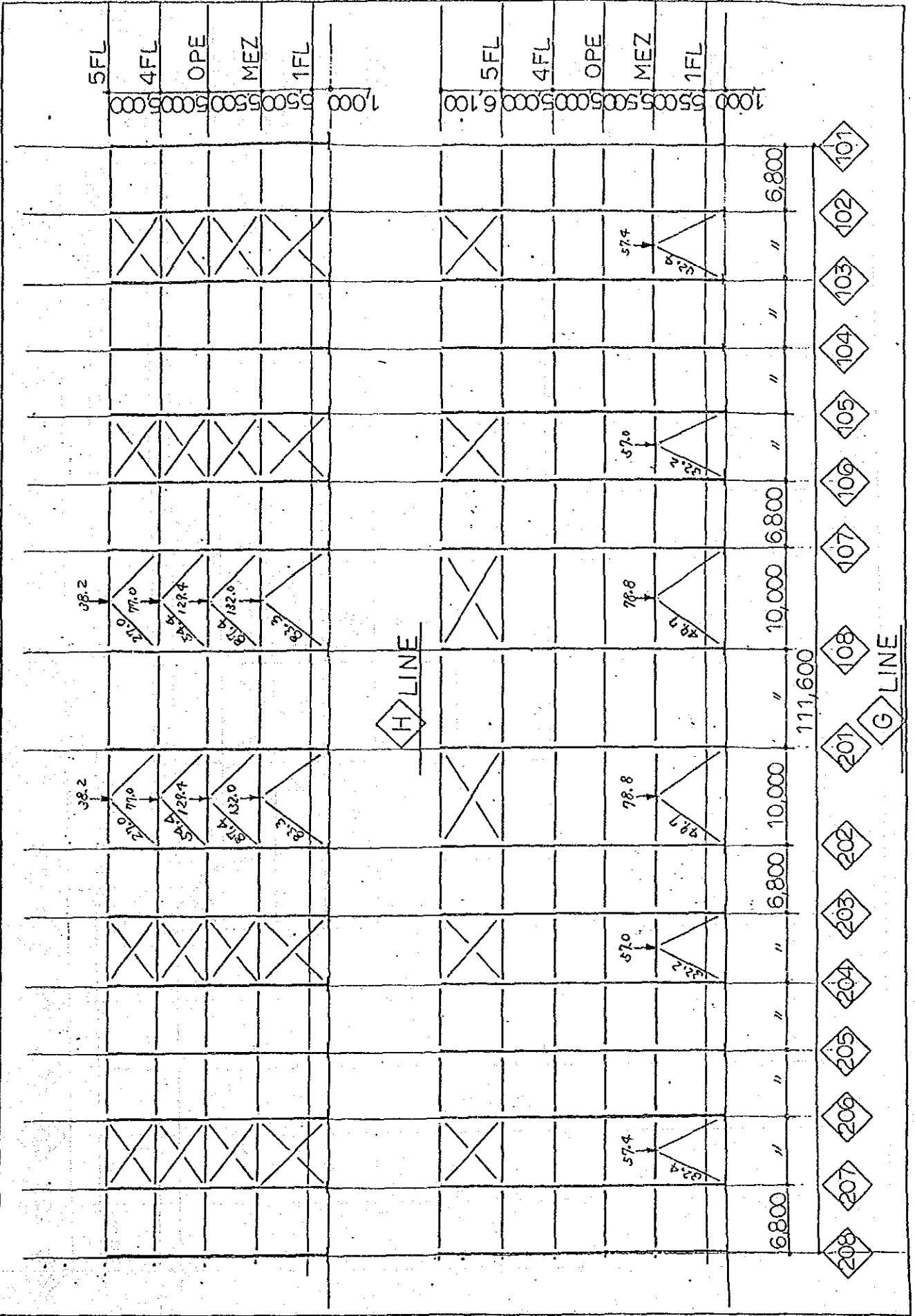


F LINE

6,800	"	"	6,800	10,000	"	10,000	6,800	"	6,800
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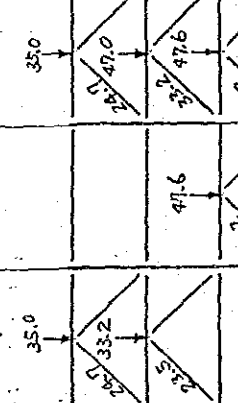
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D LINE



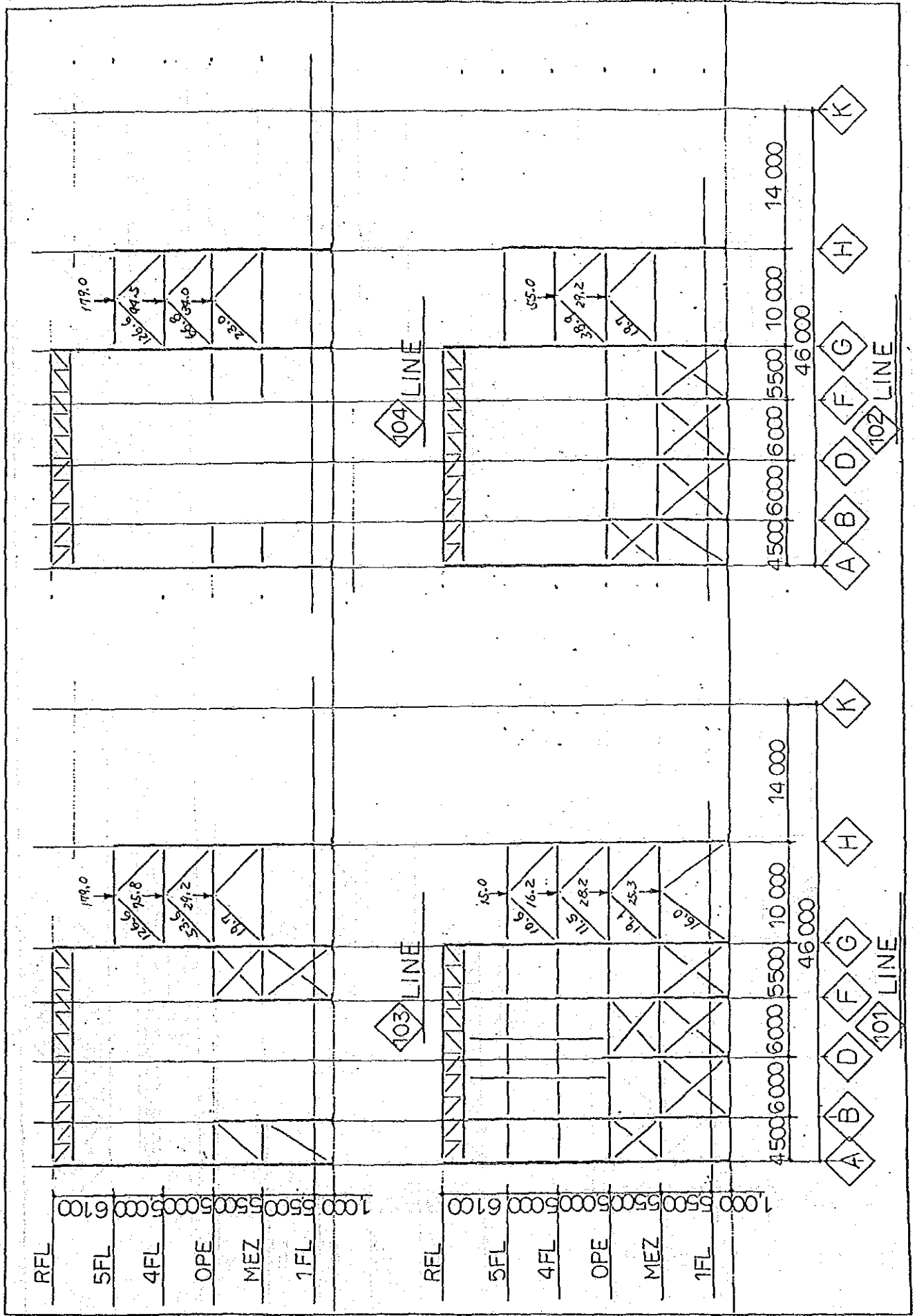
80

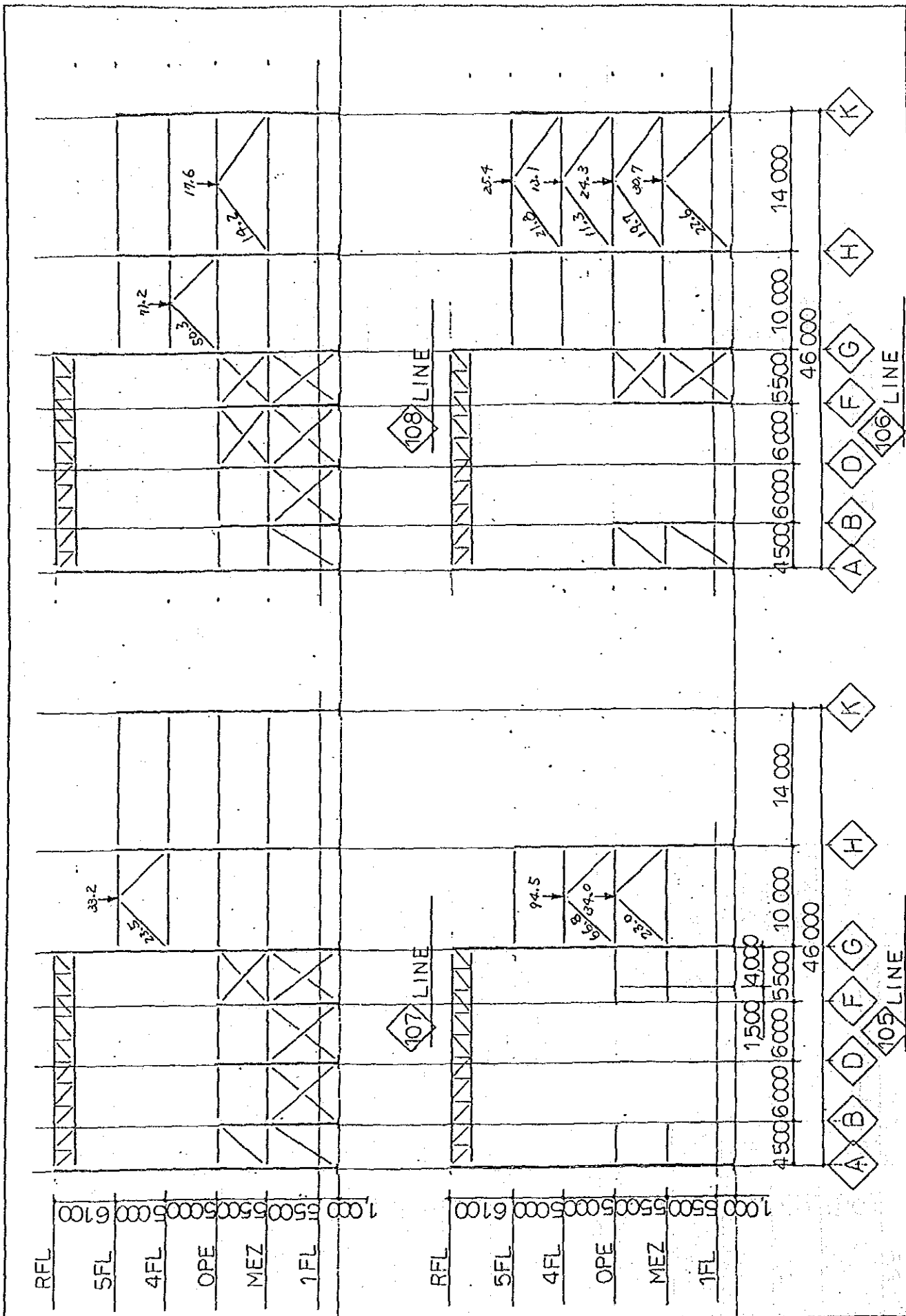
5FL	4FL	OPE	MEZ	1FL	5FL	4FL	OPE	MEZ	1FL											
1000	550	500	500	500	1000	550	500	500	500											

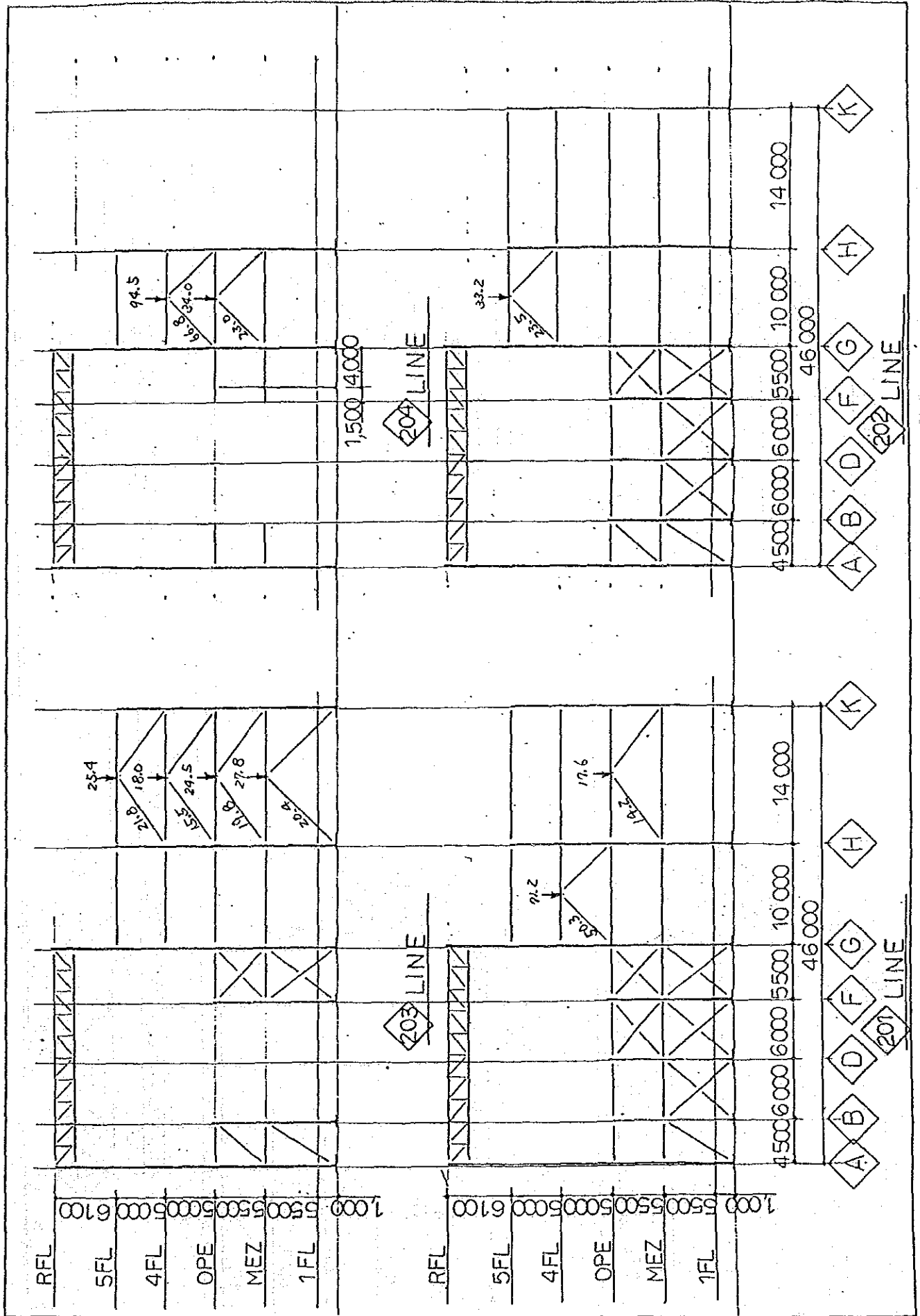


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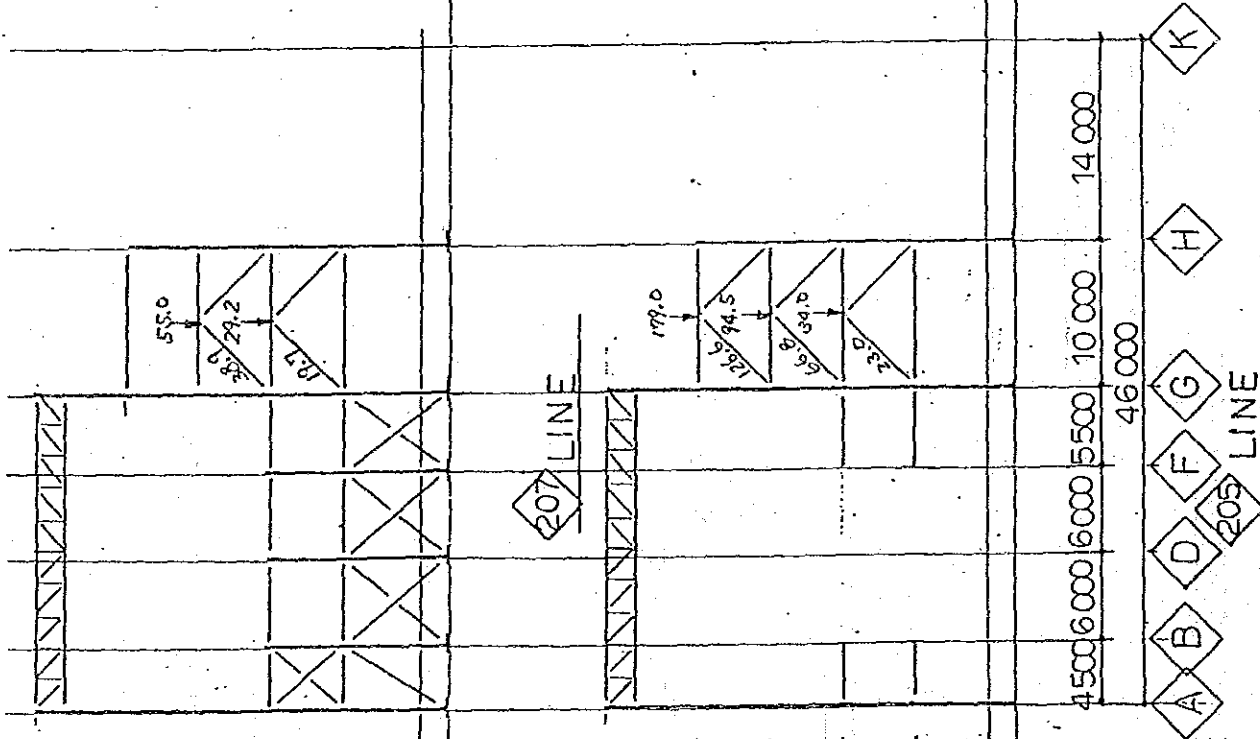
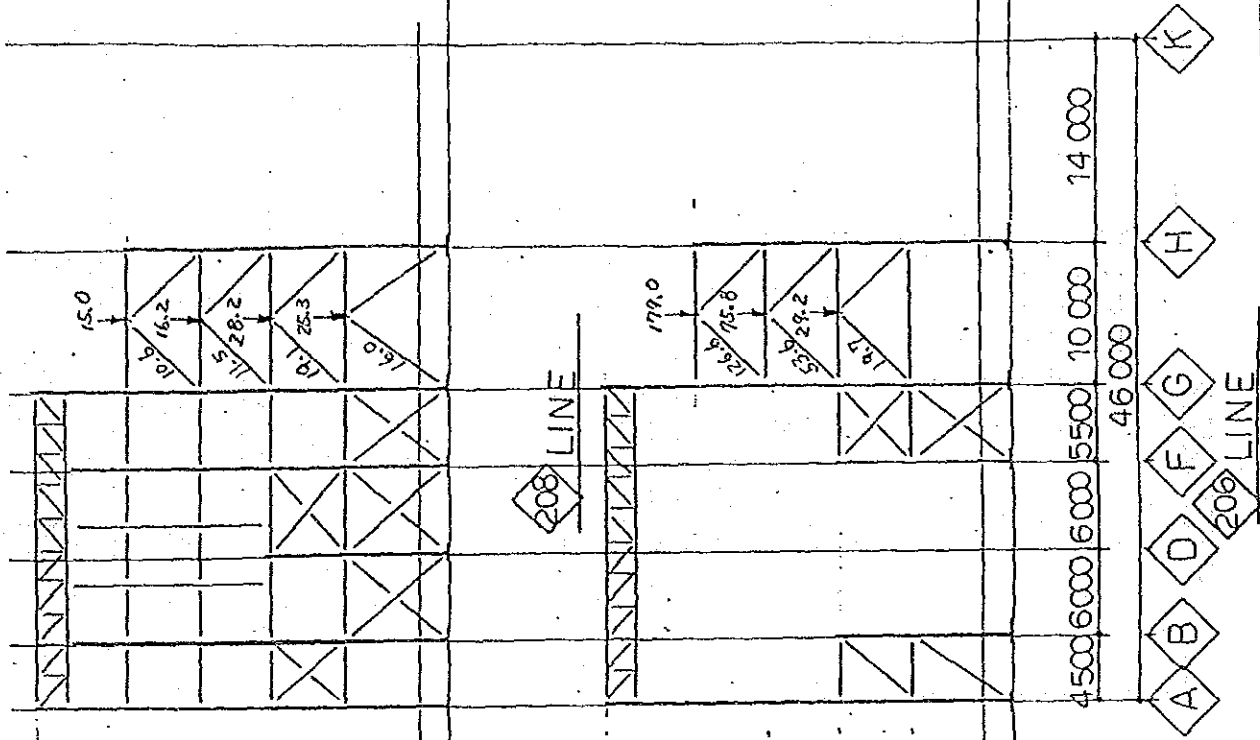
K LINE





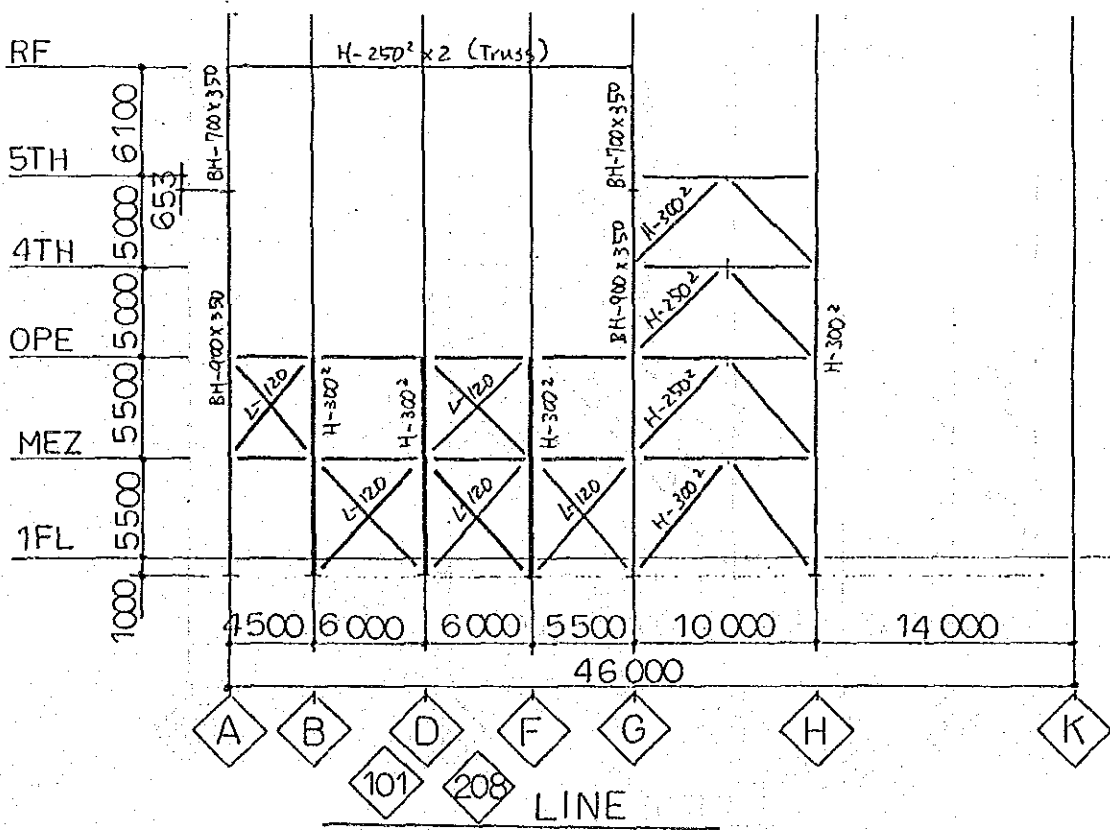
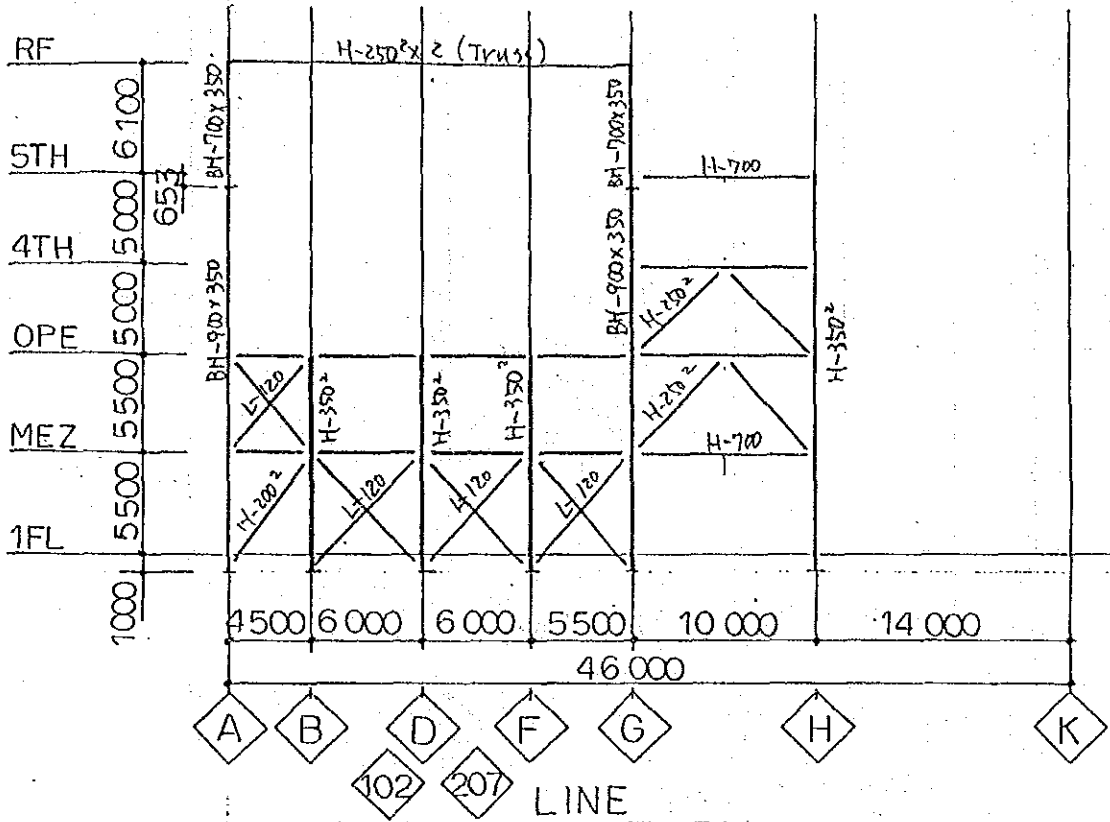


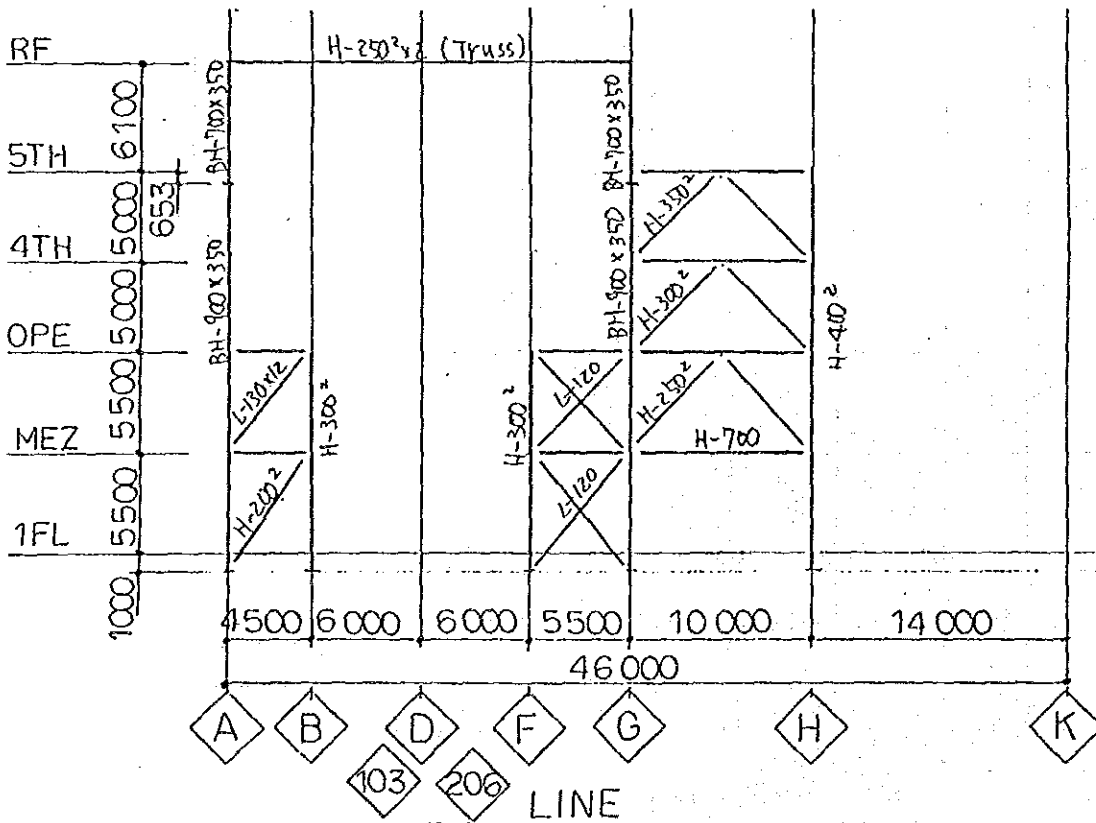
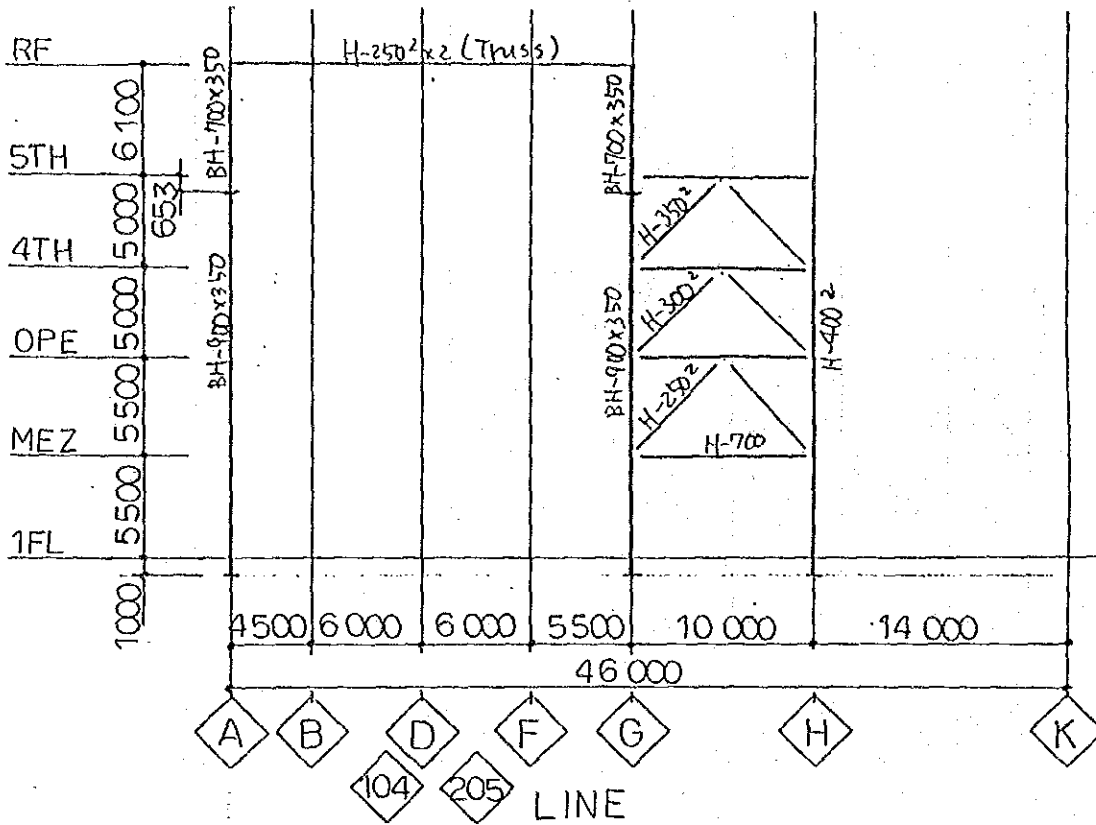
7X

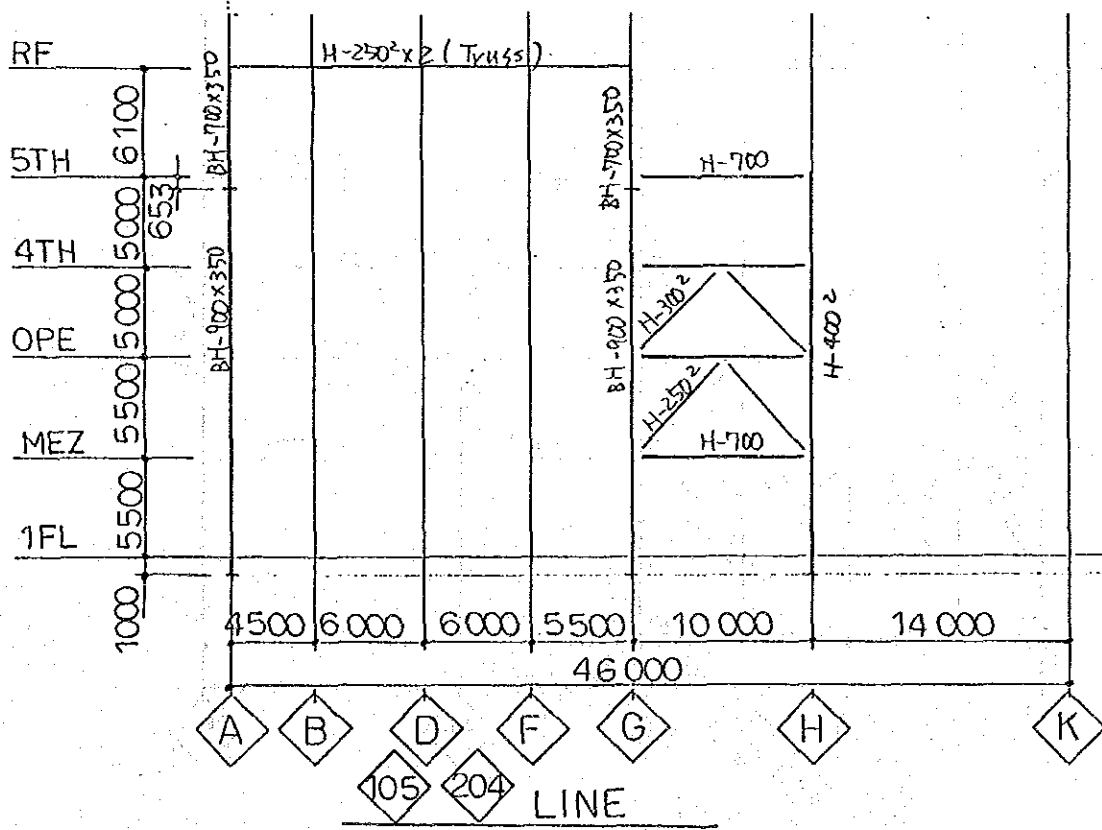
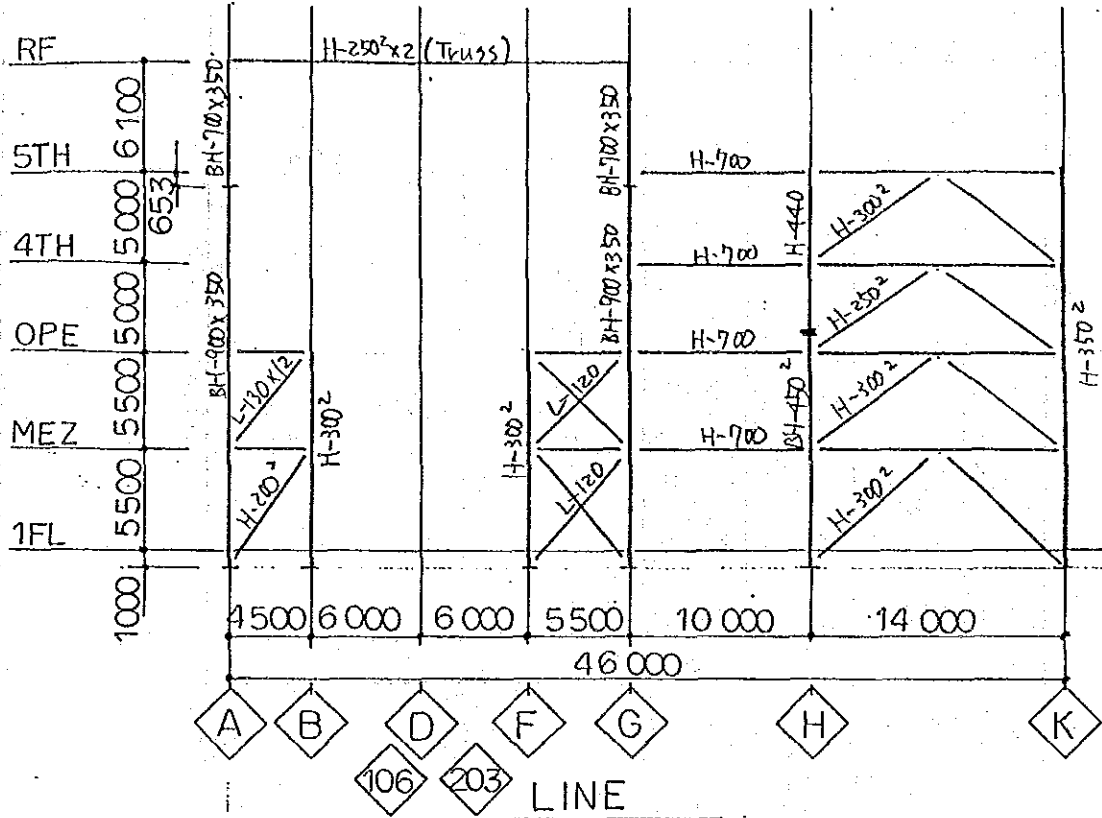


RFL	1,000	5,500	5,000	5,000	6,100
SFL	1,000	5,500	5,000	5,000	6,100
4FL	1,000	5,500	5,000	5,000	6,100
OPE	1,000	5,500	5,000	5,000	6,100
MEZ	1,000	5,500	5,000	5,000	6,100
1FL	1,000	5,500	5,000	5,000	6,100

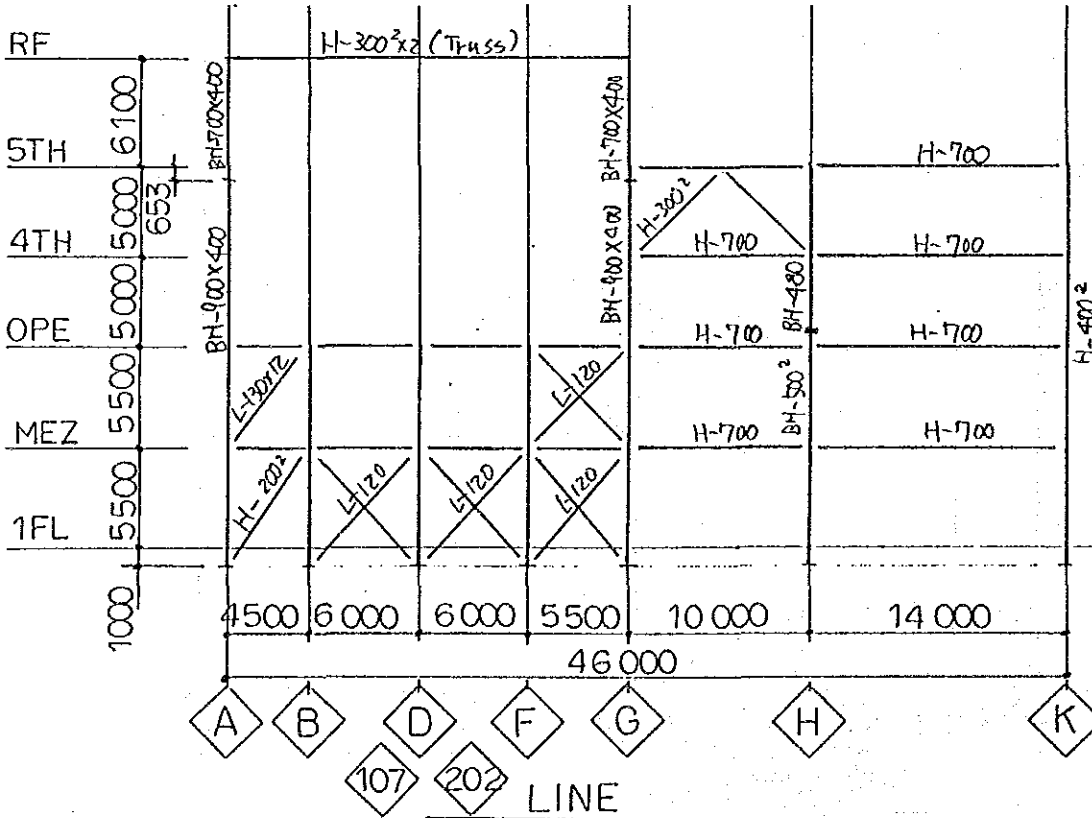
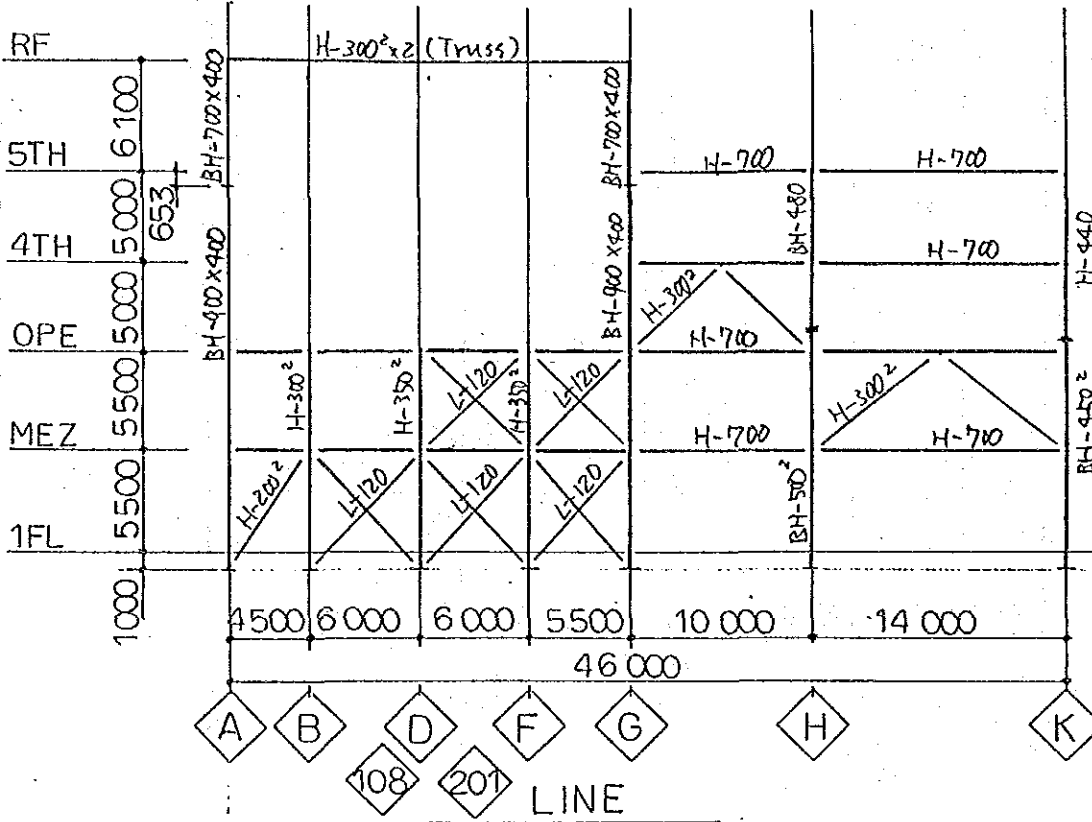
3.2 Input data



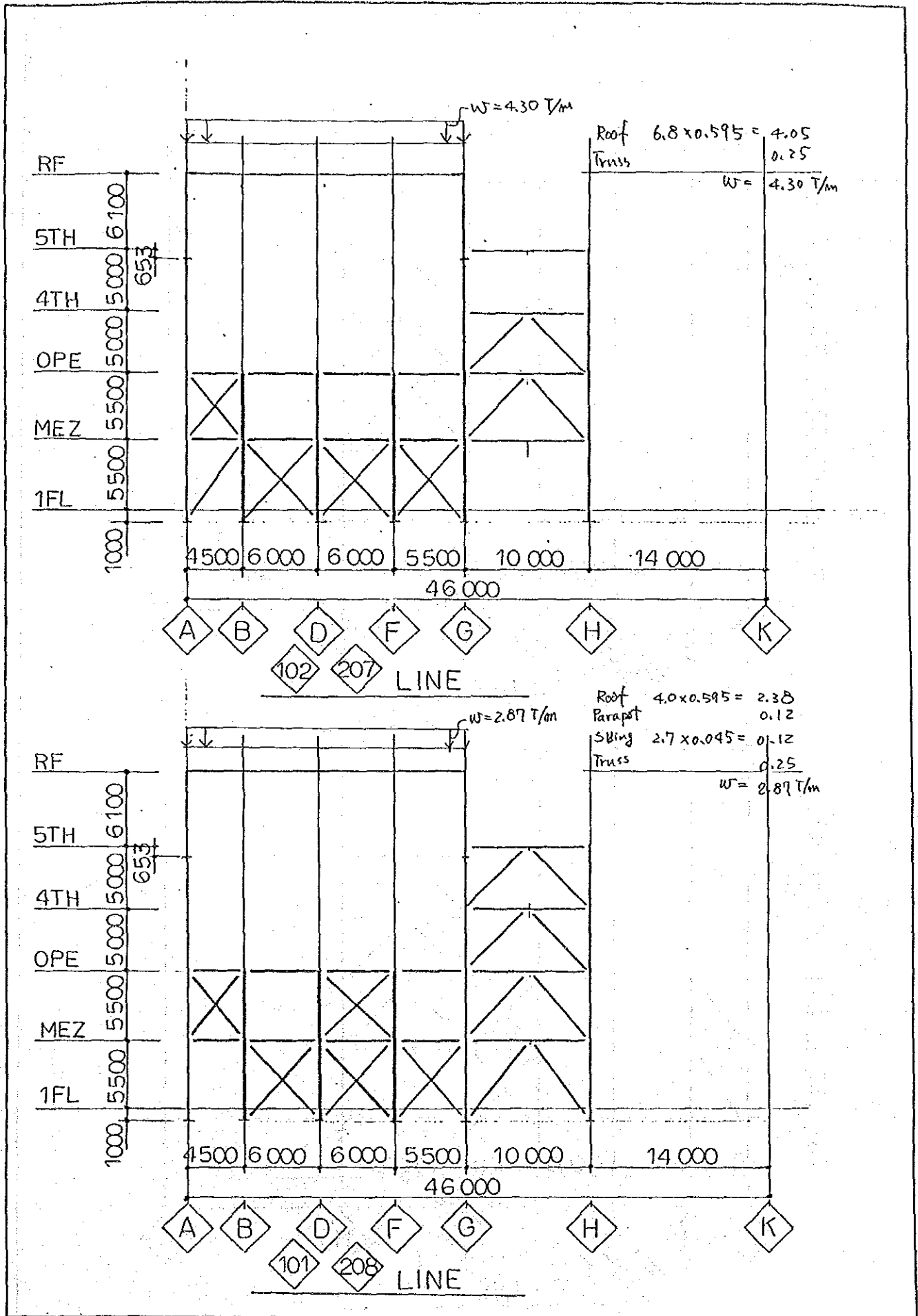


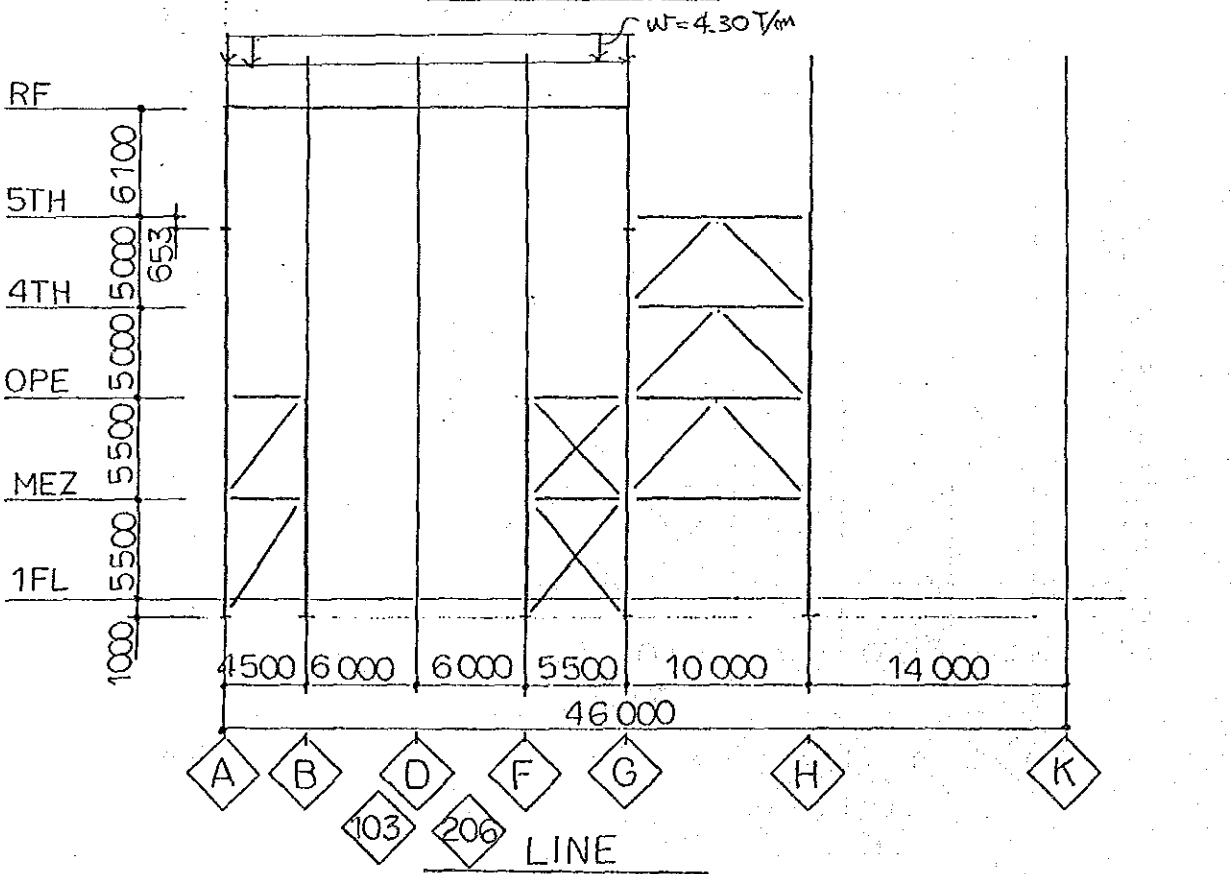
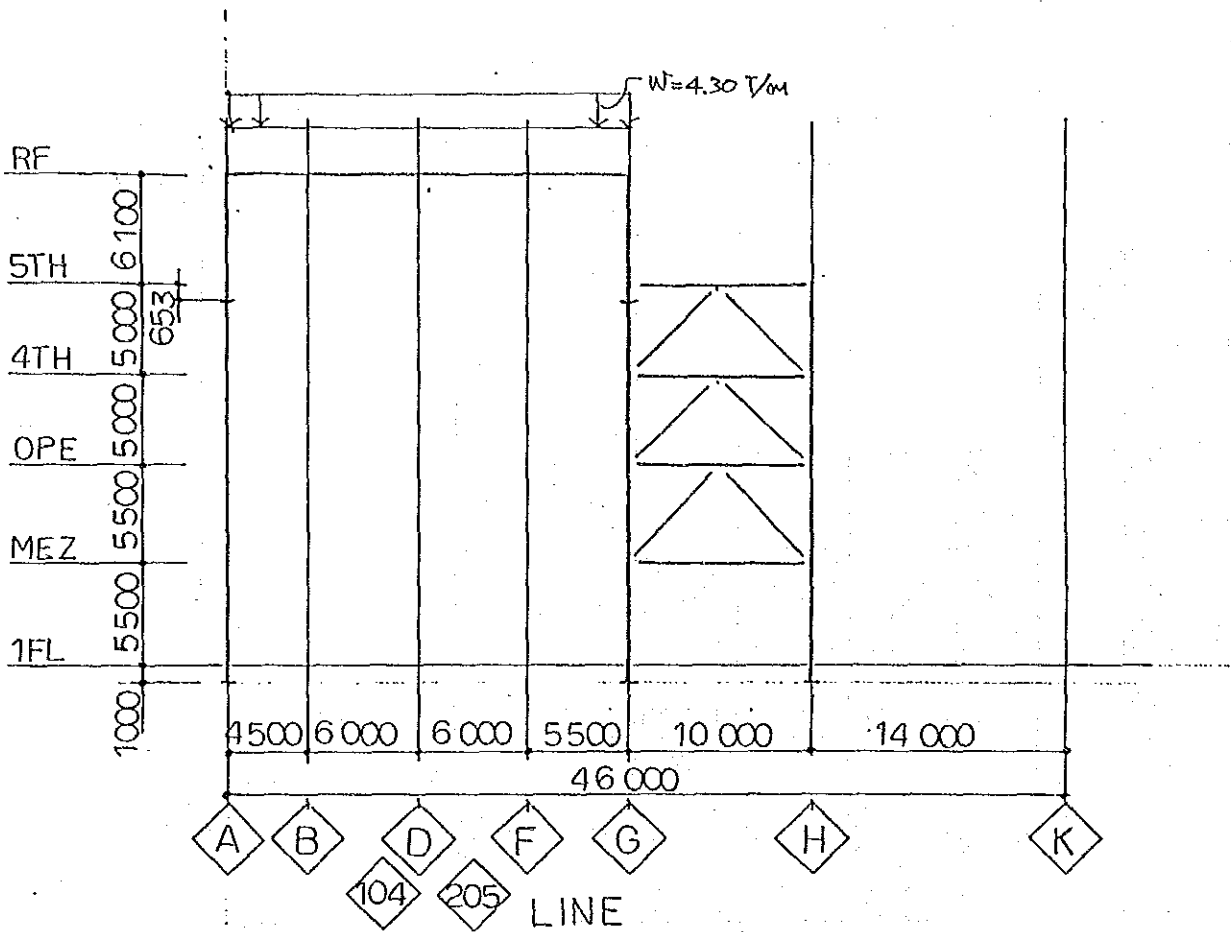


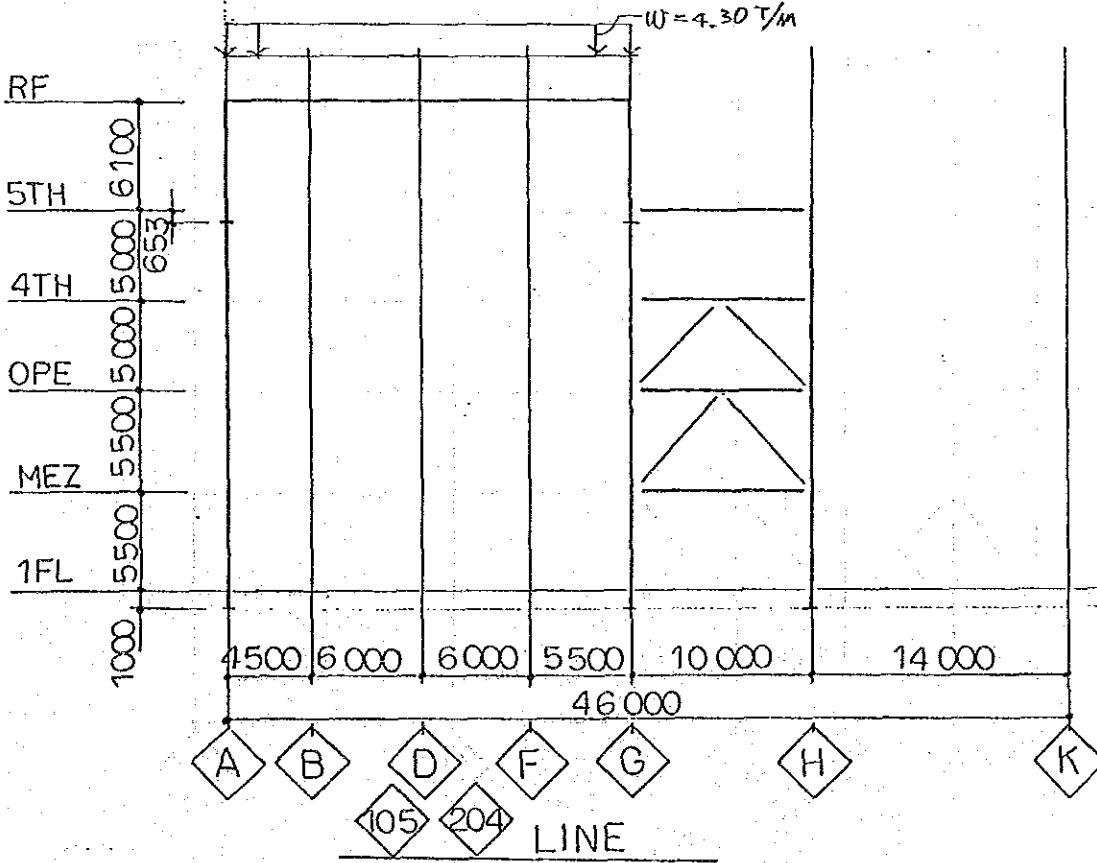
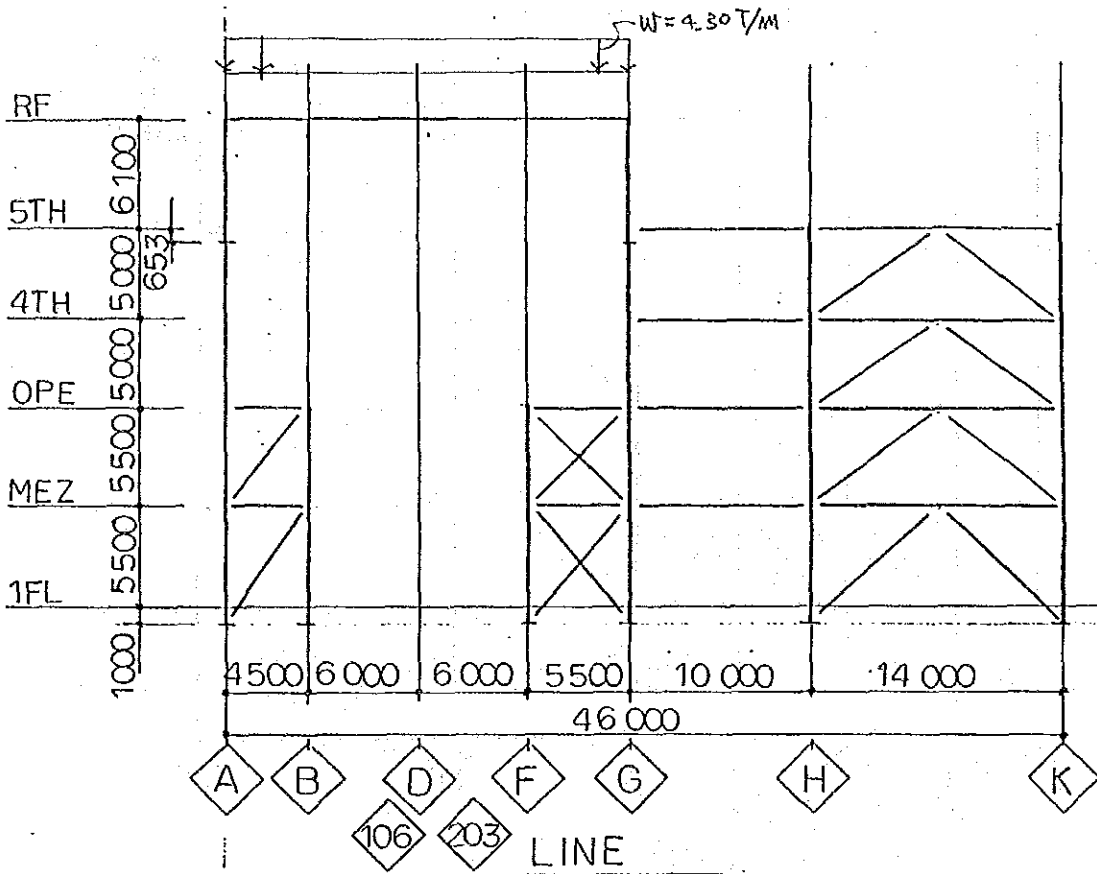
PA

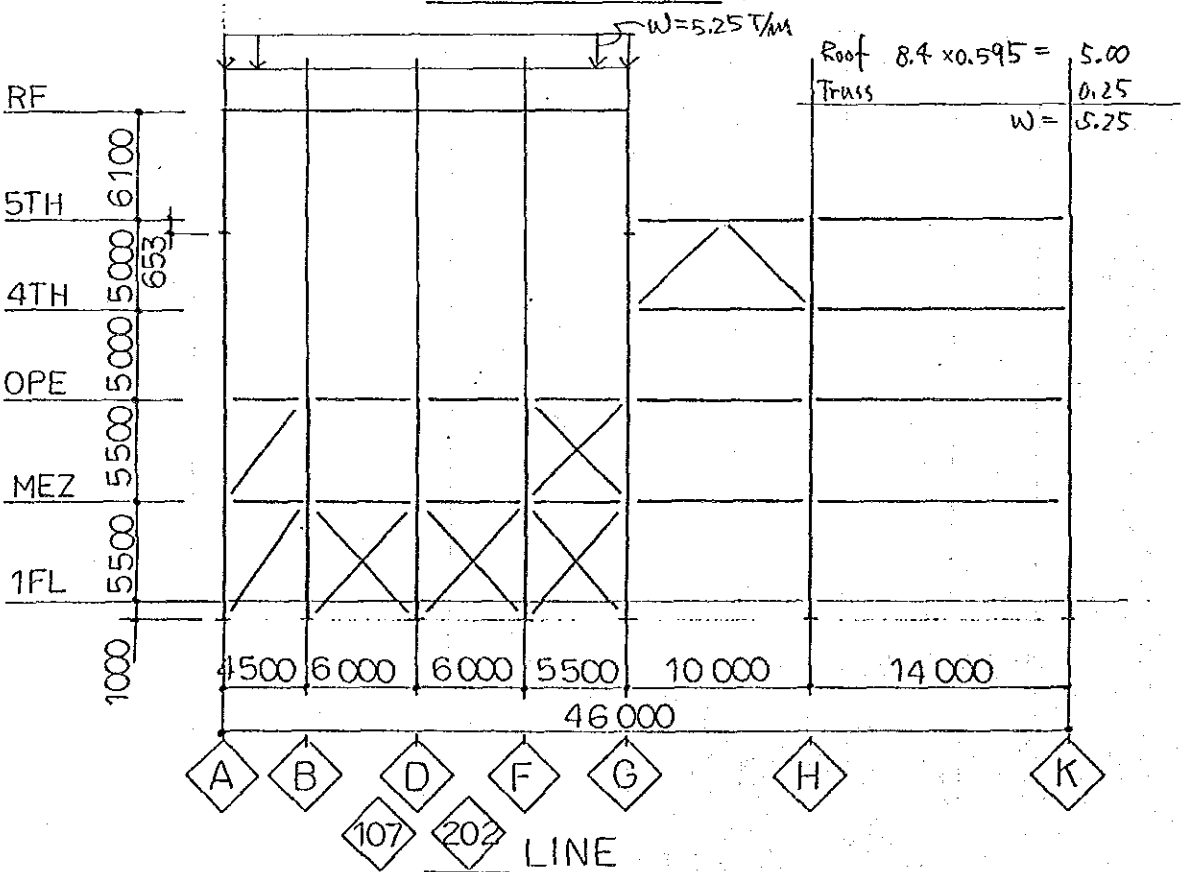
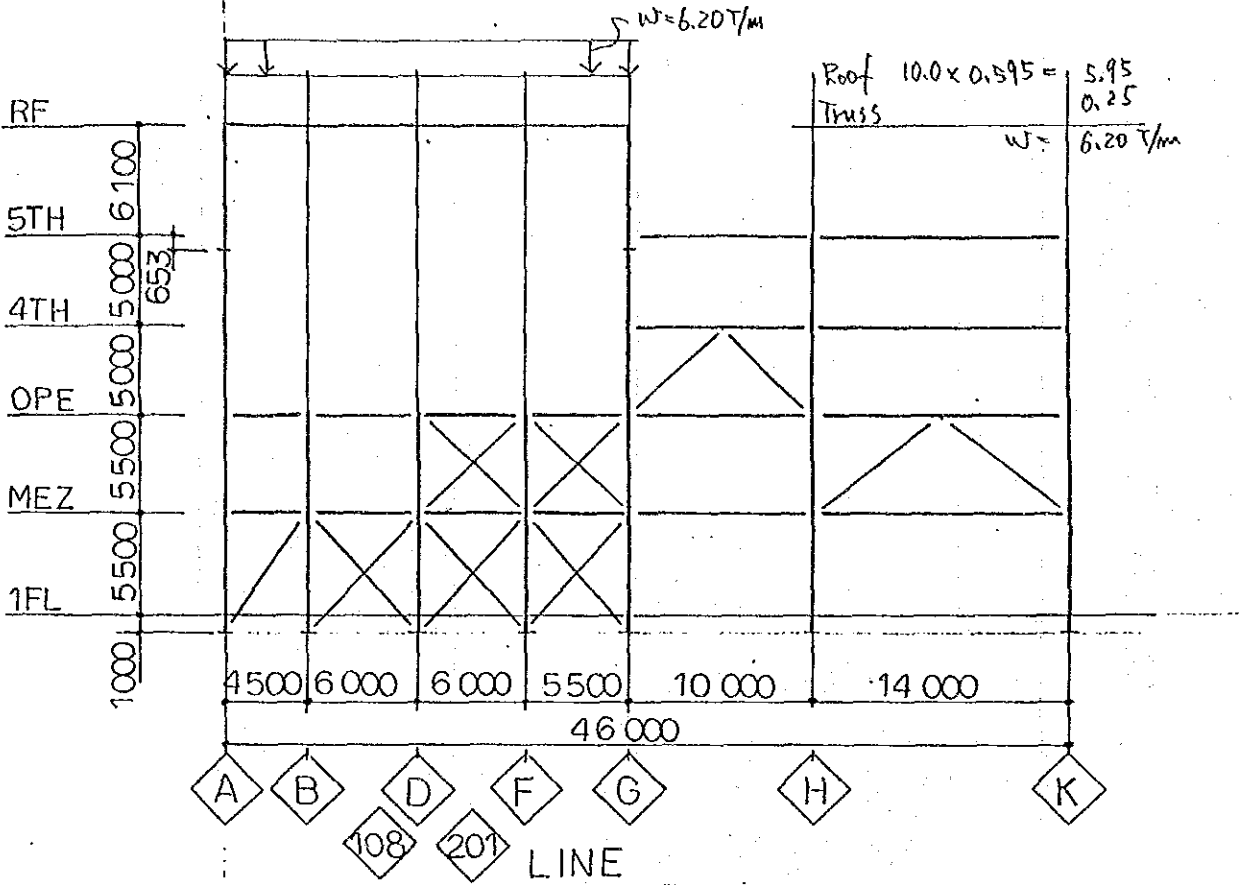


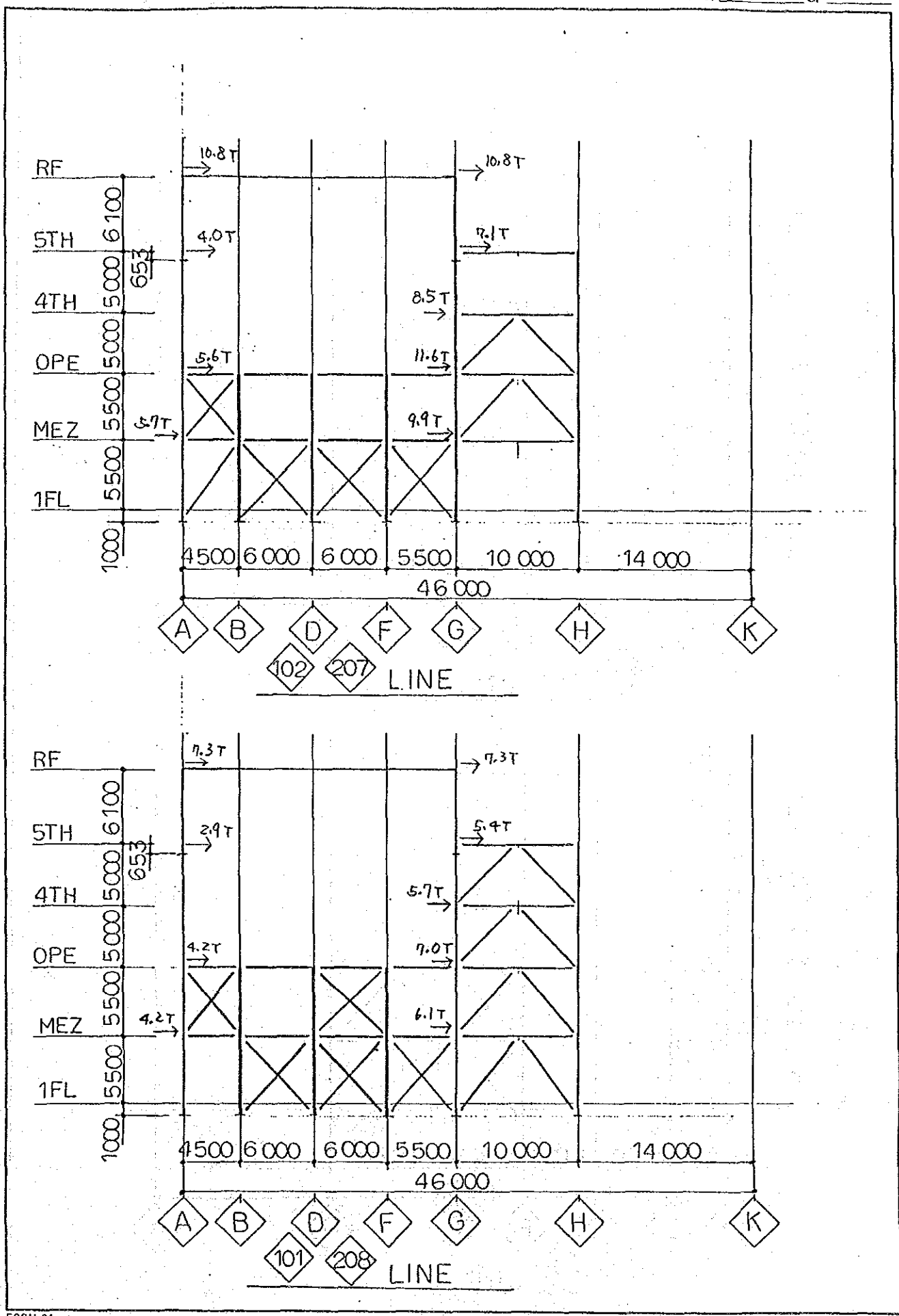
01



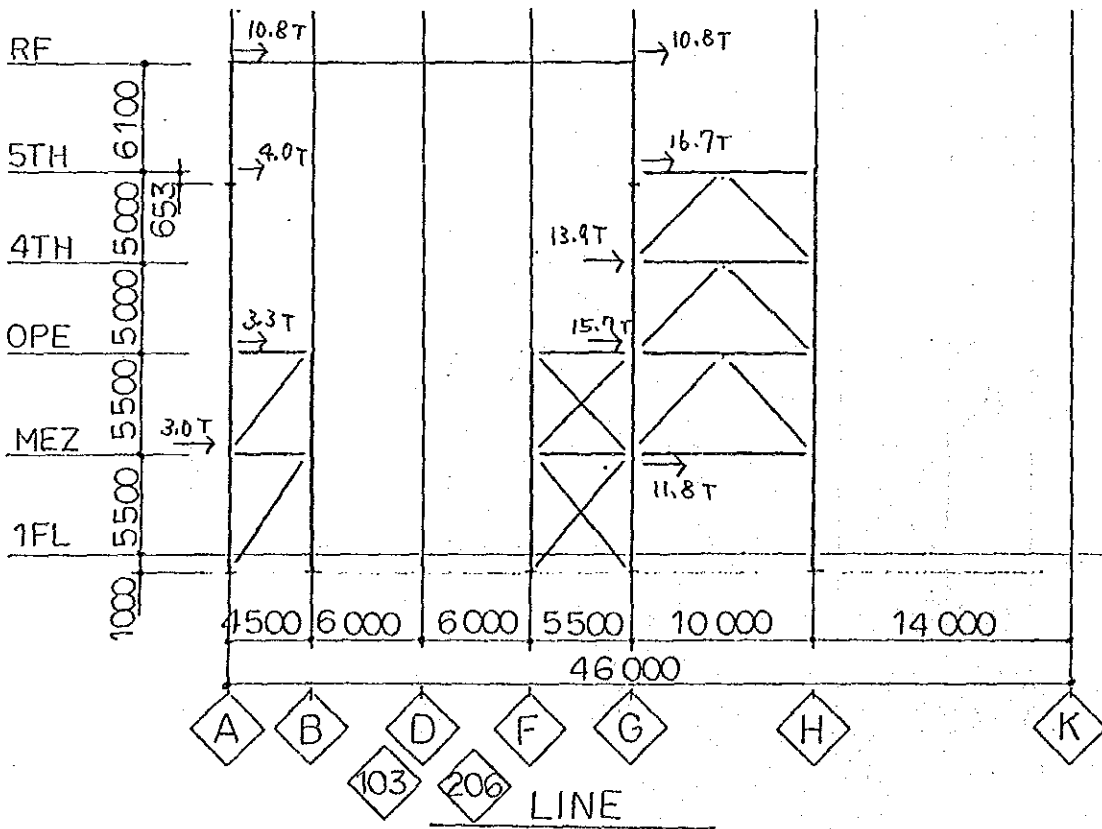
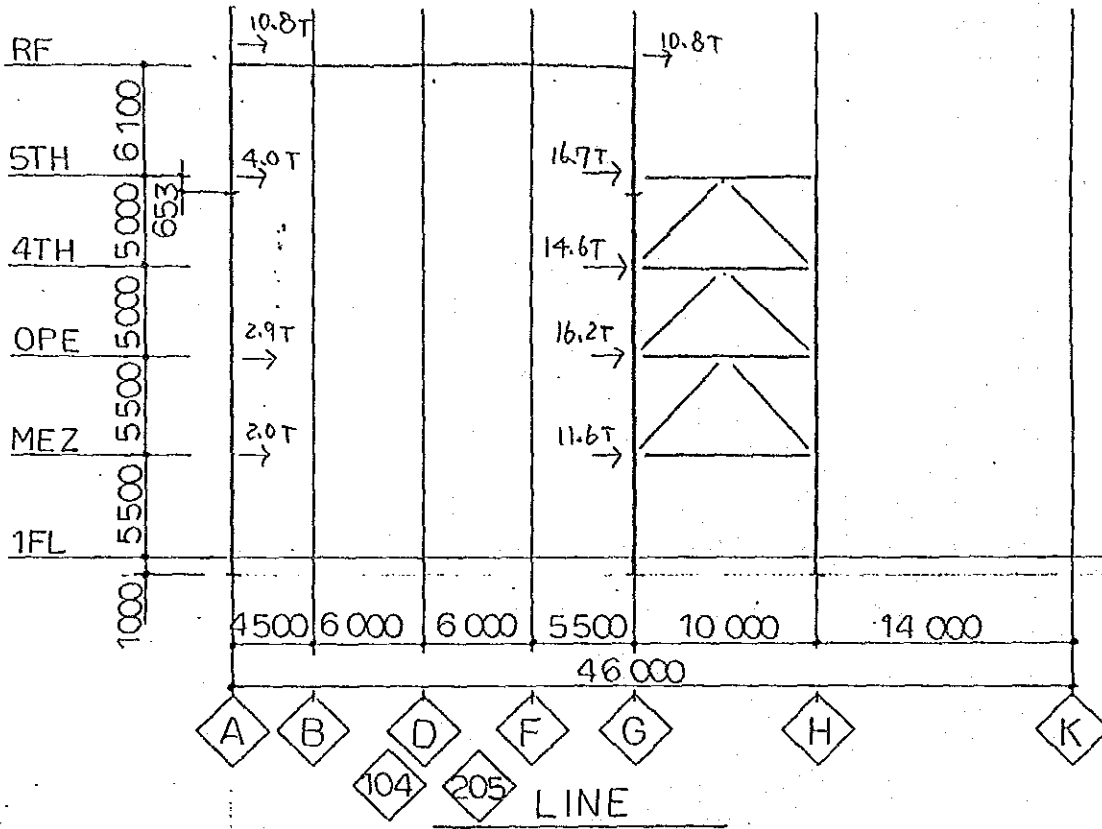


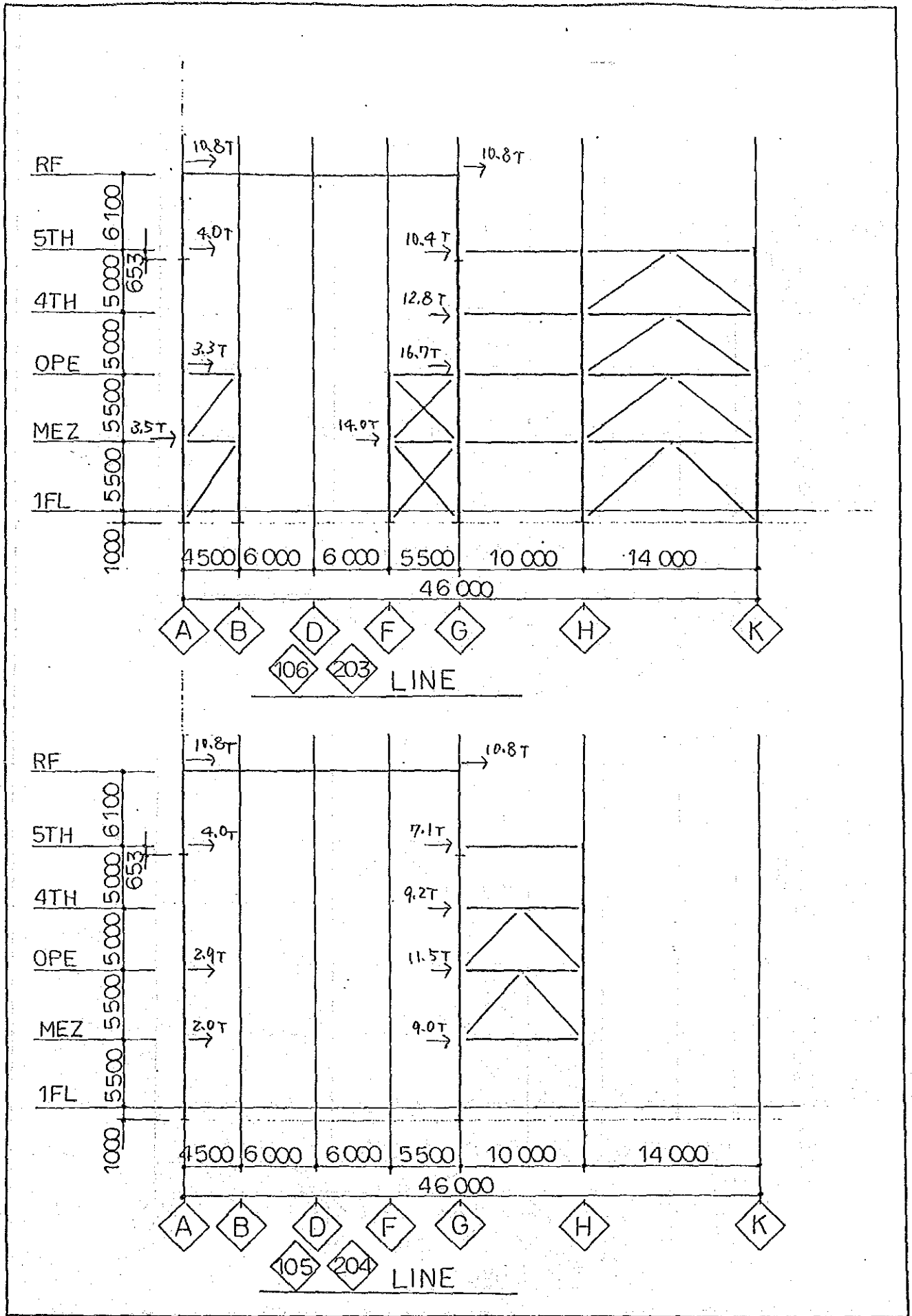


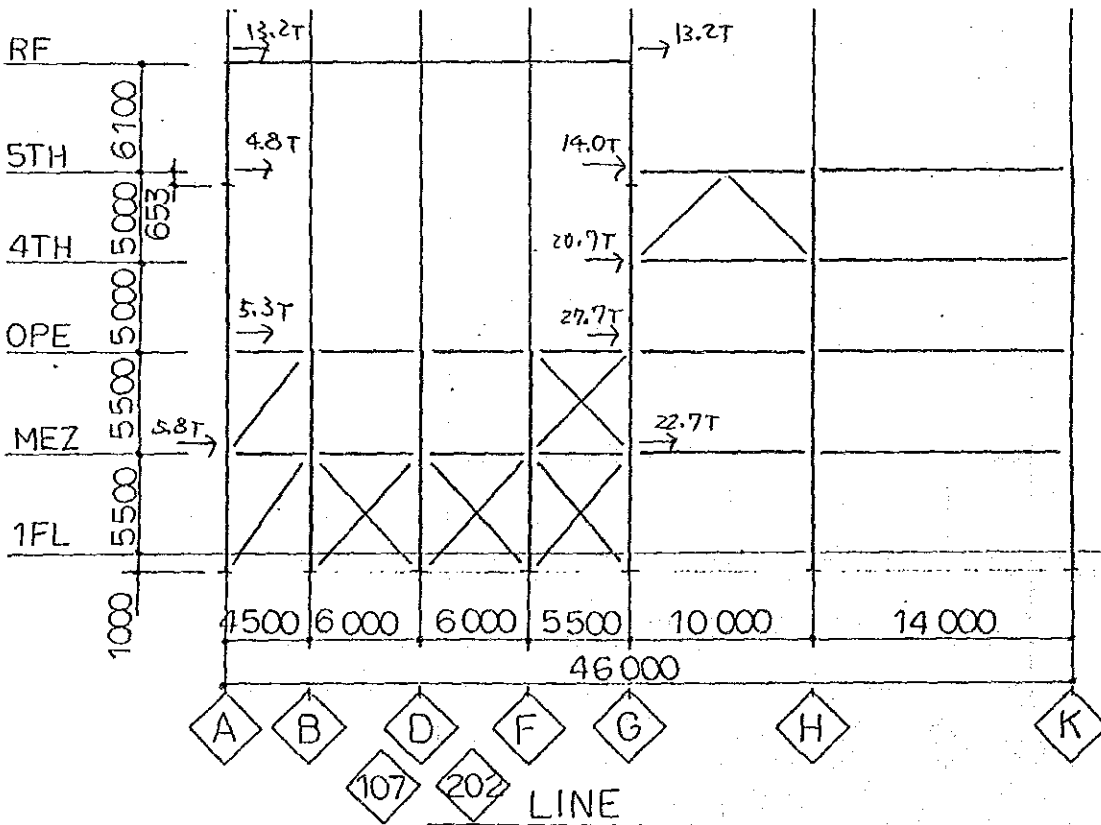
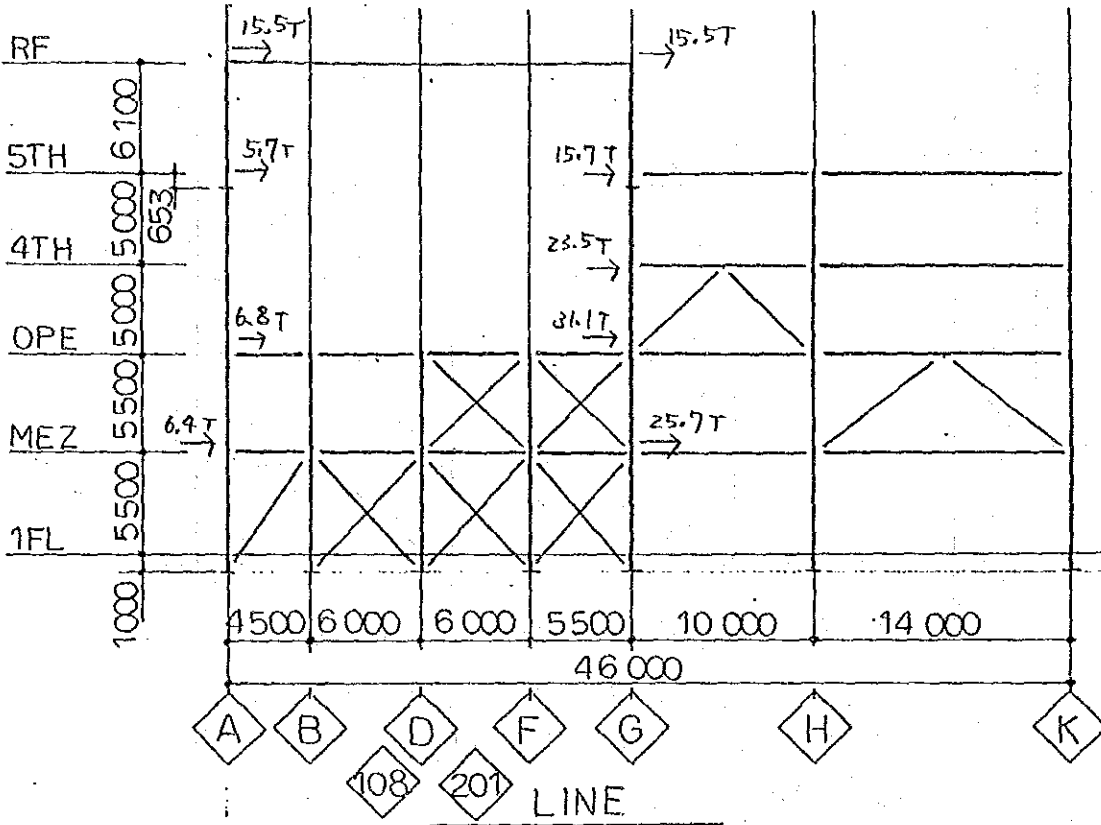


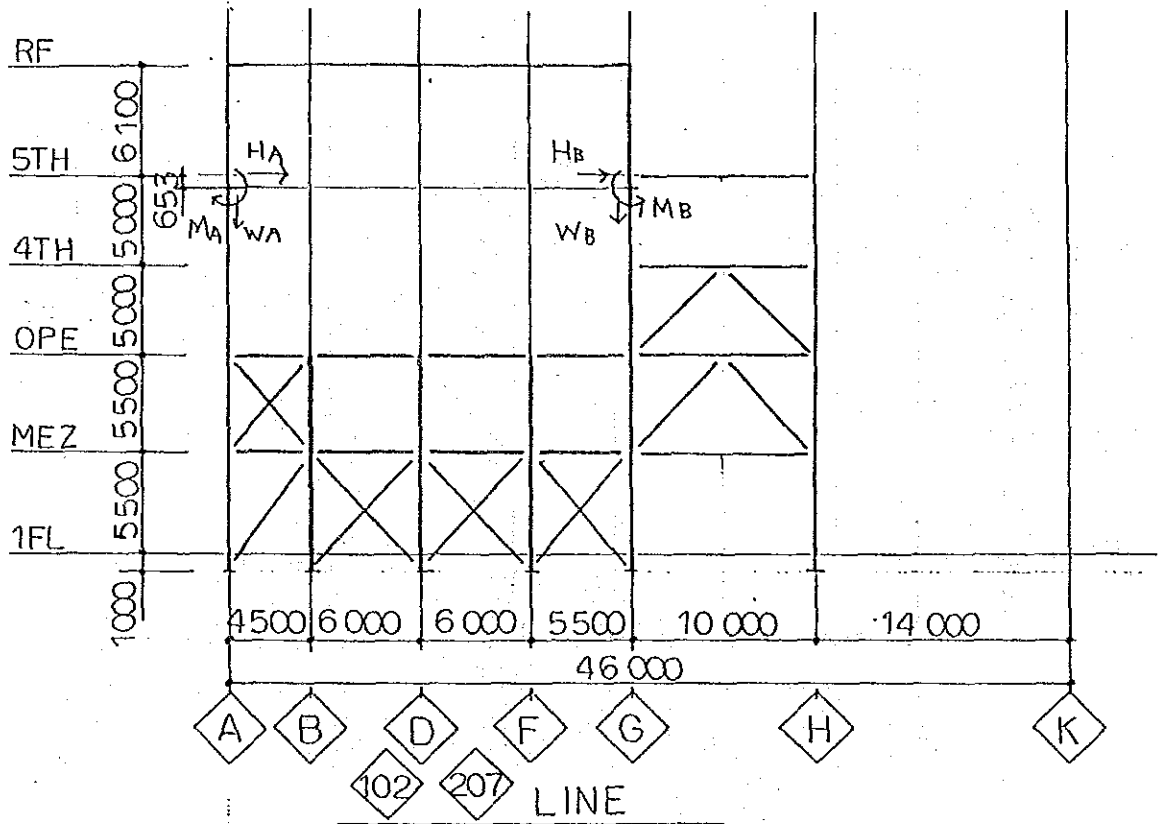


94





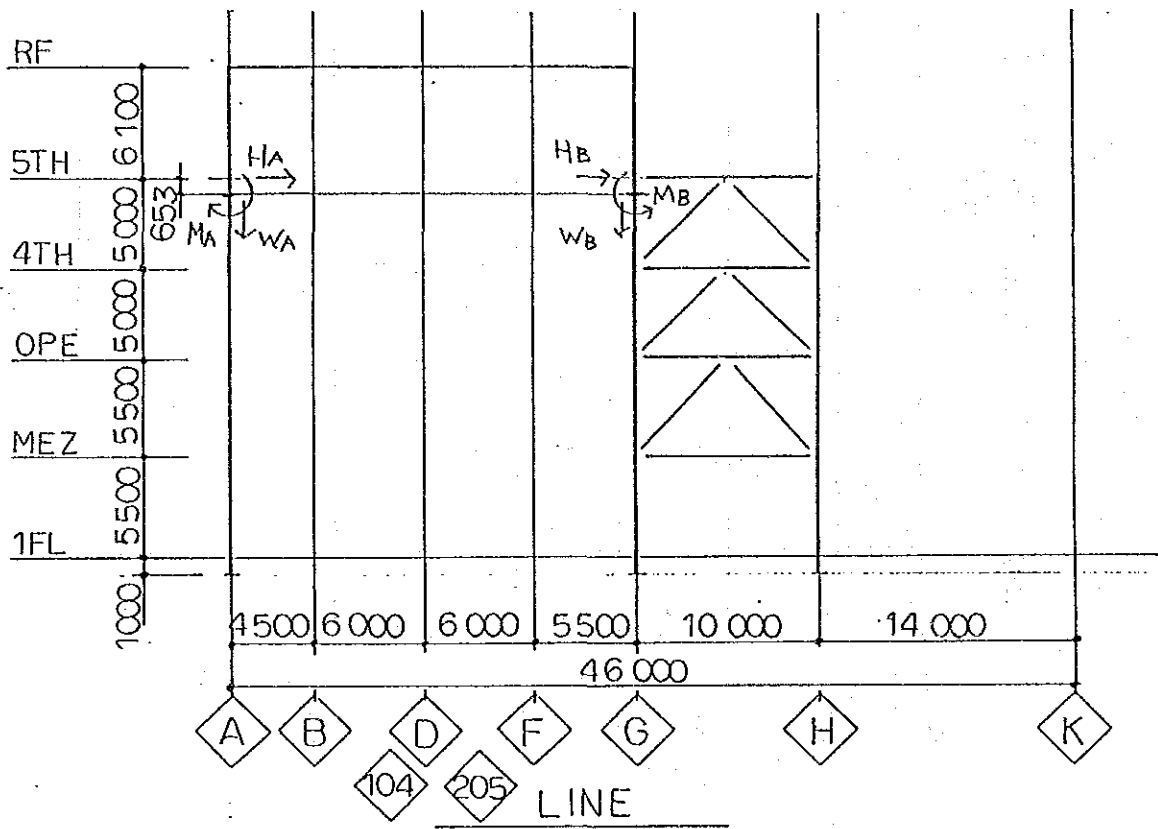




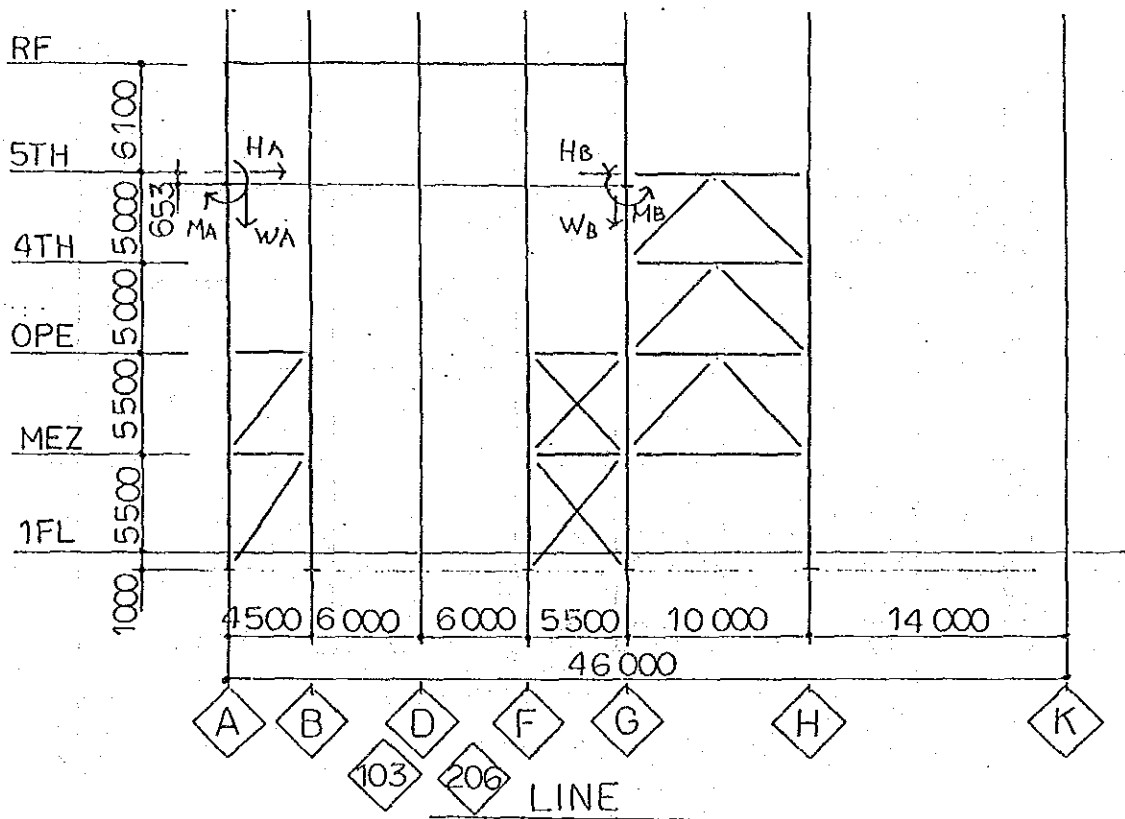
C.D.L. $W_A = 24.8 T$ $W_B = 13.3 T$
 $H_A = 2.1 T$ $H_B = 1.1 T$
 $M_A = 19.8 T.m$ $M_B = 10.6 T.m$

C.L.L. $W_A = 71.2 T$ $W_B = 17.8 T$
 $H_A = 5.9 T$ $H_B = 1.5 T$
 $M_A = 57.0 T.m$ $M_B = 14.2 T.m$

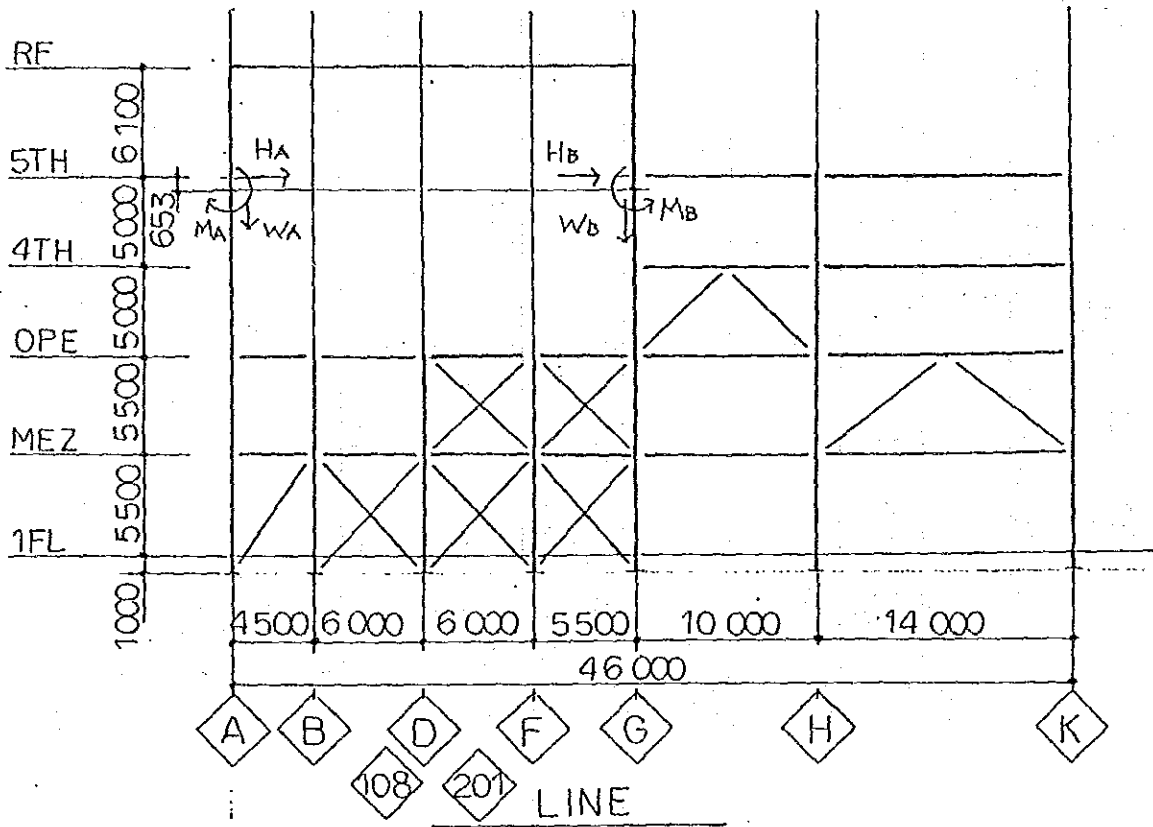
C.H.L. $H_A = 4.2 T$ $H_B = 2.3 T$
 (K=0.17)



C.D.L.	$W_A = 24.8T$	$W_B = 13.3T$
	$H_A = 2.1T$	$H_B = 1.1T$
	$M_A = 19.8T.m$	$M_B = 10.6T.m$
C.L.L.	$W_A = 71.2T$	$W_B = 17.8T$
	$H_A = 5.9T$	$H_B = 1.5T$
	$M_A = 57.0T.m$	$M_B = 14.2T.m$
C.H.L.	$H_A = 4.2T$	$H_B = 2.3T$

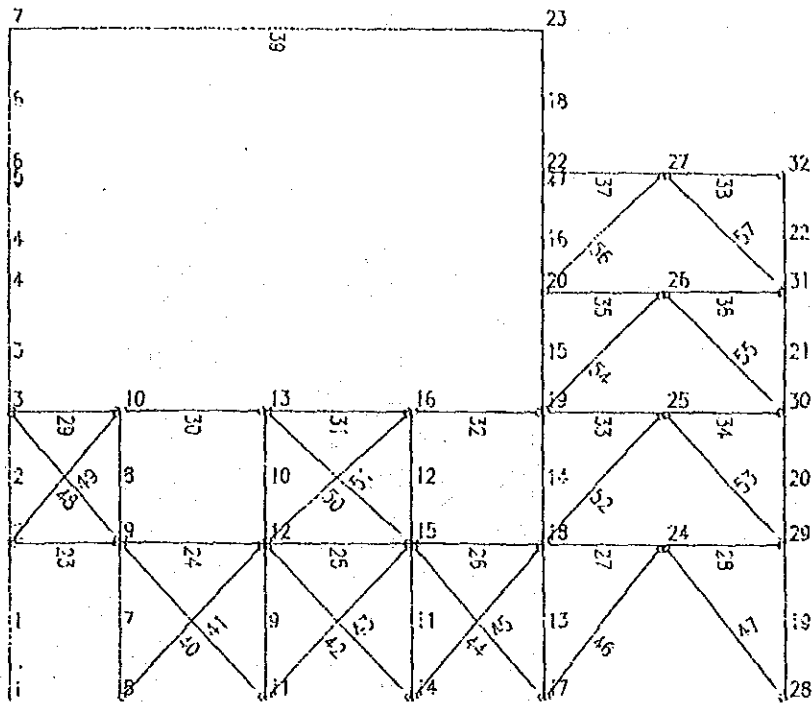


C.D.L.	WA =	24.8 T	WB =	13.3 T
	HA =	2.1 T	HB =	1.1 T
	MA =	19.8 T.m	MB =	10.6 T
C.L.L.	WA =	71.2 T	WB =	17.8 T
	HA =	5.9 T	HB =	1.5 T
	MA =	57.0 T.m	MB =	14.2 T.m
C.H.L.	HA =	4.2 T	HB =	2.3 T



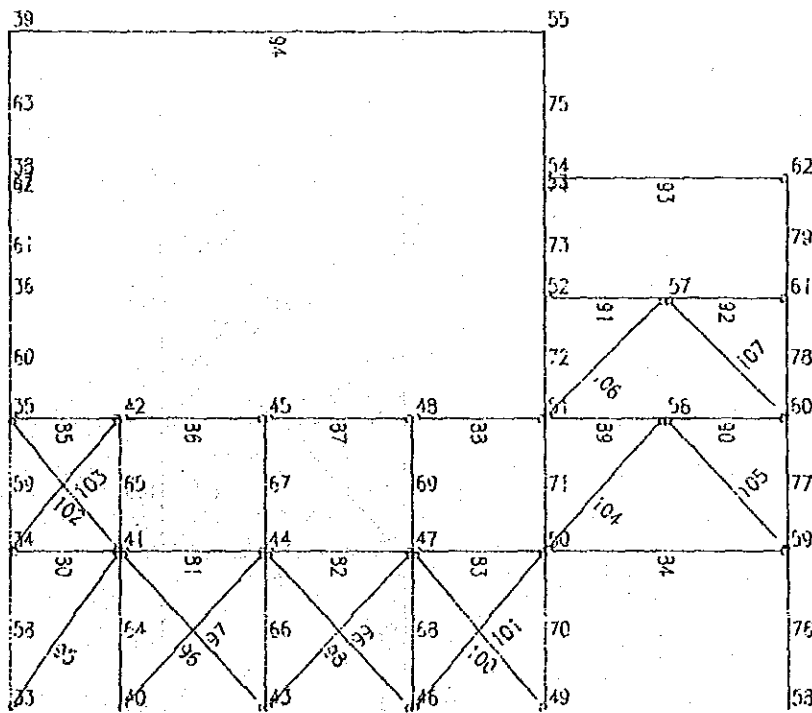
C.D.L.	WA = 28.1 T	WB = 15.1 T
	HA = 2.3 T	HB = 1.3 T
	MA = 22.5 T.m	MB = 12.1 T.m
C.L.L.	WA = 80.6 T	WB = 20.2 T
	HA = 6.7 T	HB = 1.7 T
	MA = 64.5 T.m	MB = 16.2 T.m
C.H.L.	HA = 4.8 T	HB = 2.6 T

101



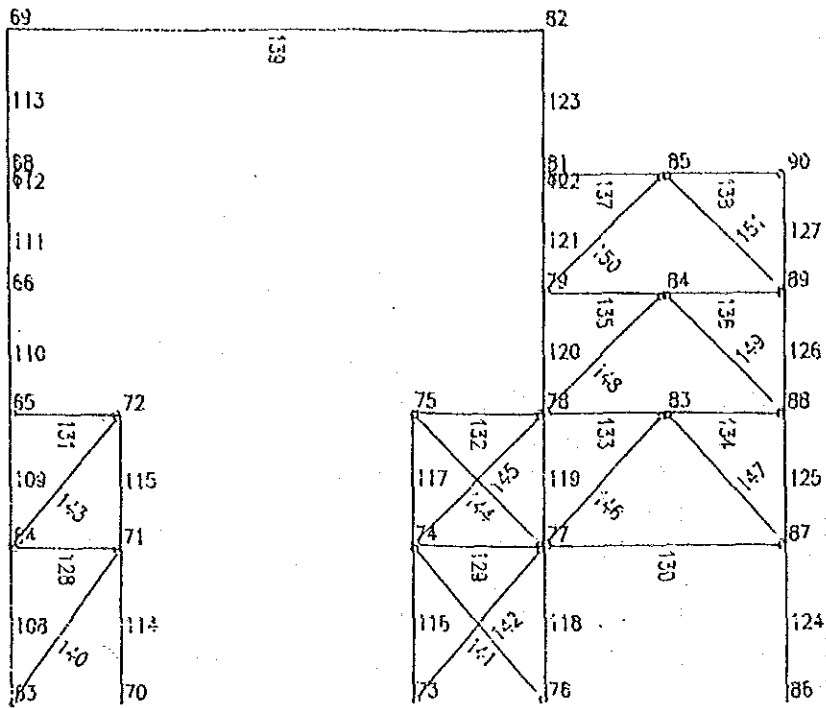
フレーム モデル 図

(101 - LINE SCALE 1/300)



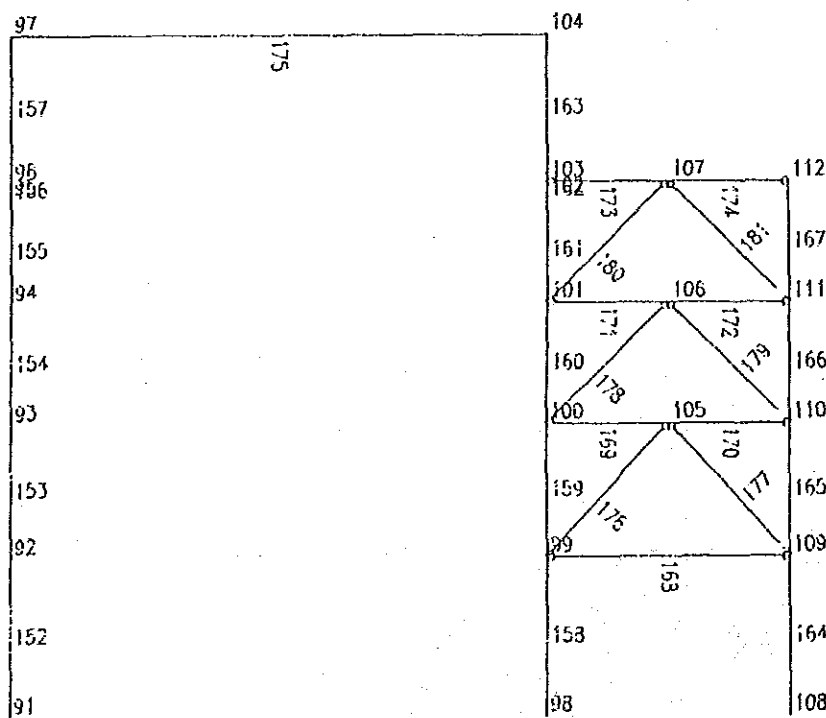
フレーム モデル 図

(102 - LINE SCALE 1/300)



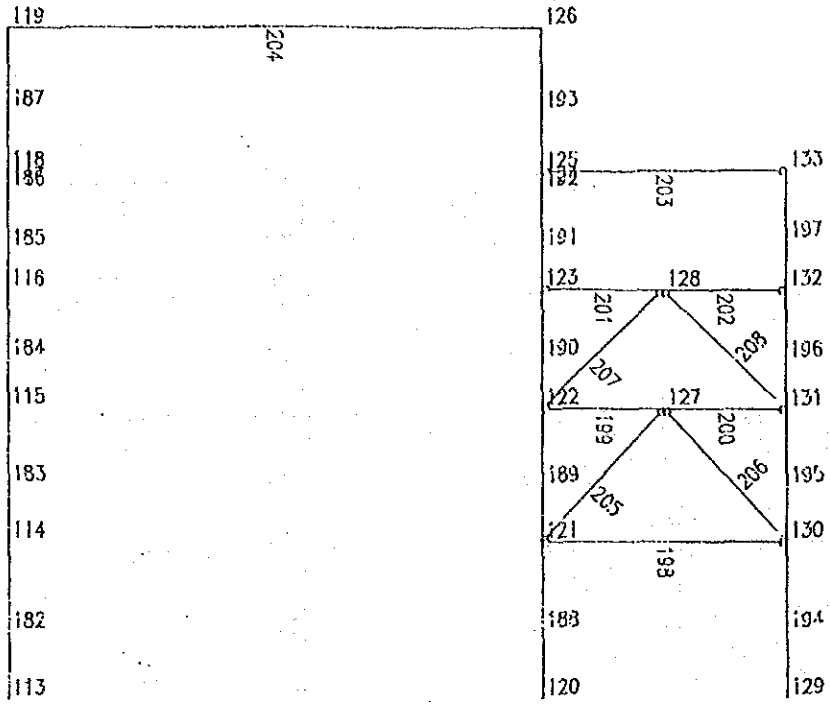
フレーム モデル 図

(103 - LINE SCALE 1/300)



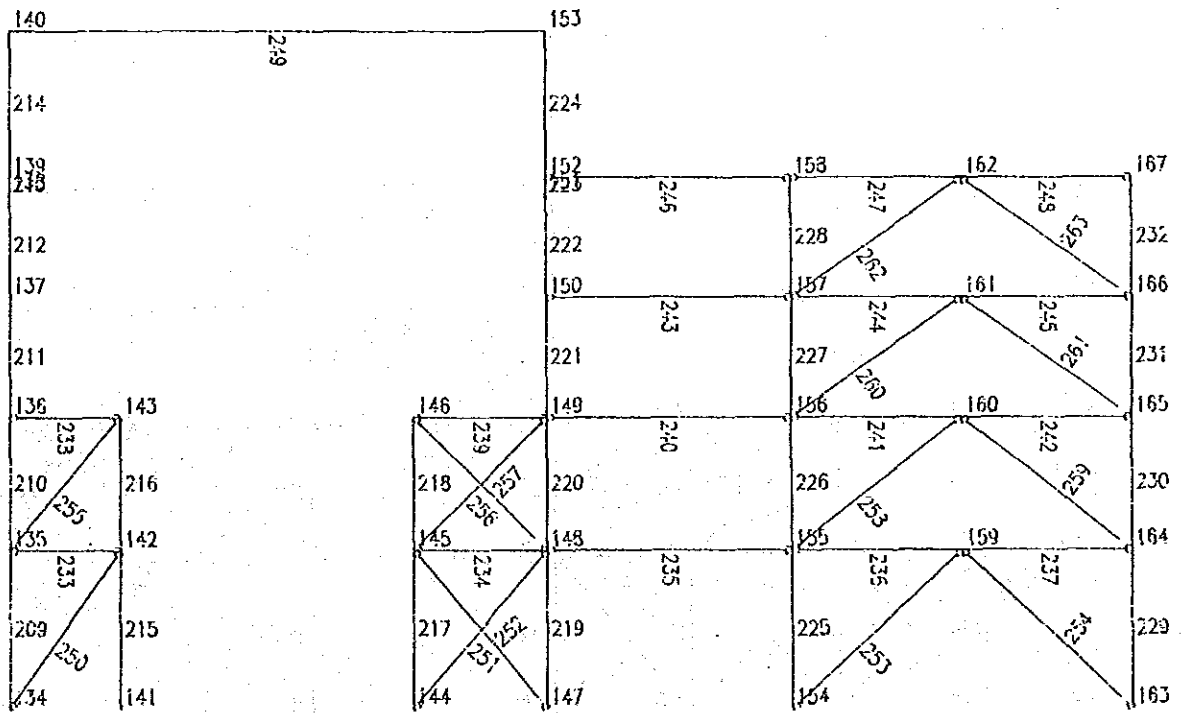
フレーム モデル 図

(104 - LINE SCALE 1/300)



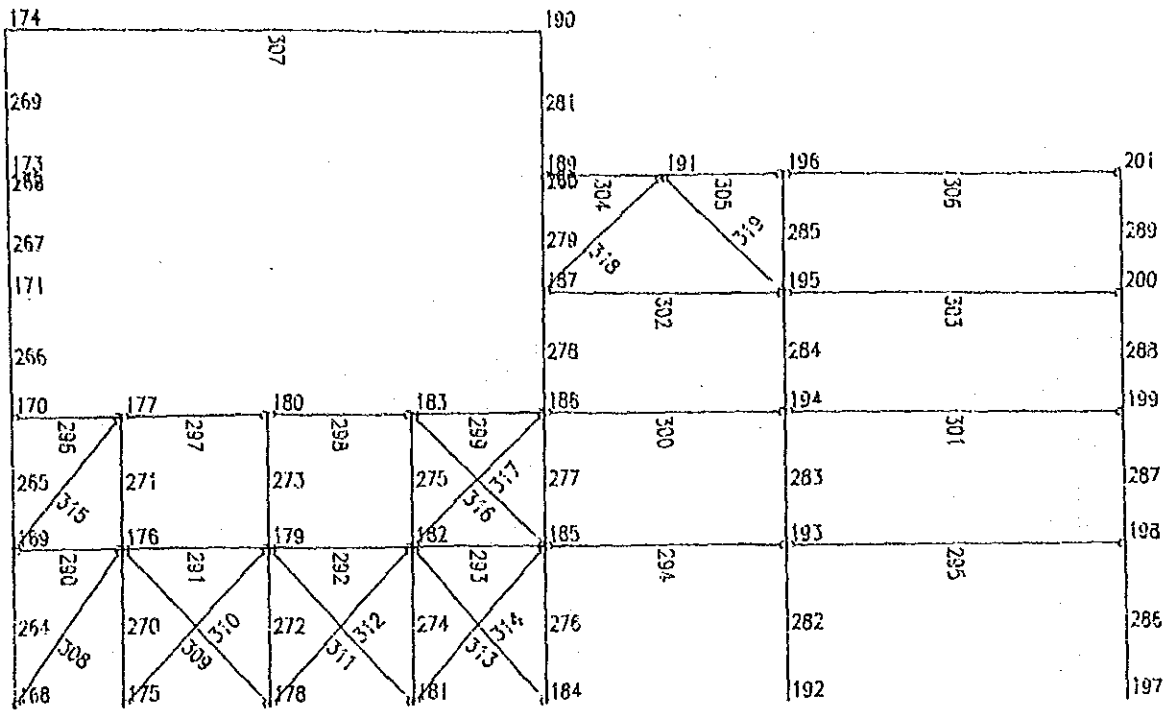
フレーム モデル 図

(105 - LINE. SCALE. 1/300)

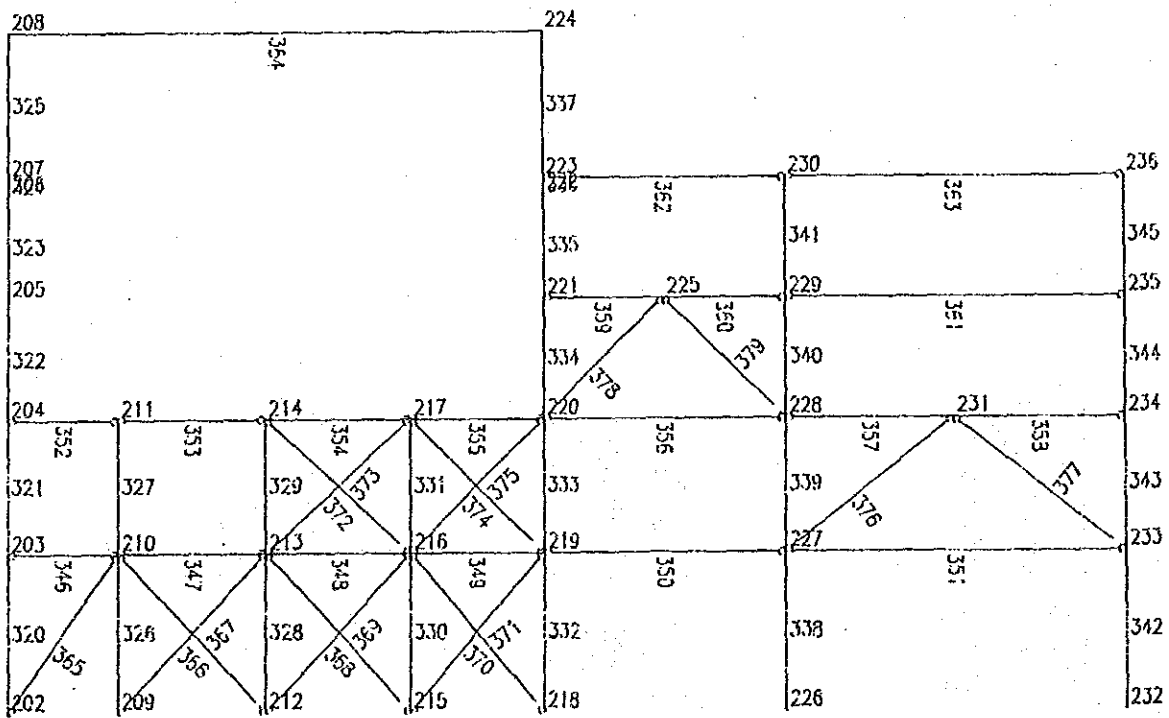


フレーム モデル 図

(106 - LINE. SCALE. 1/300)

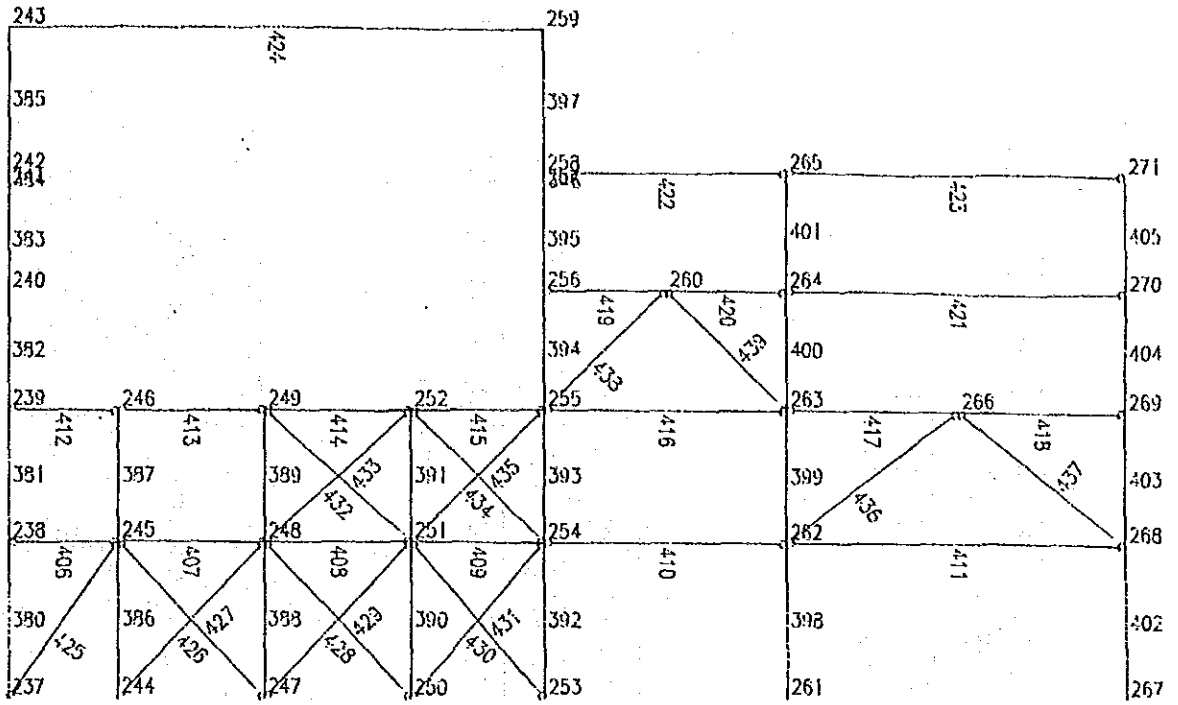


フレーム モデル ☒
 (107 -- LINE. SCALE 1/300)

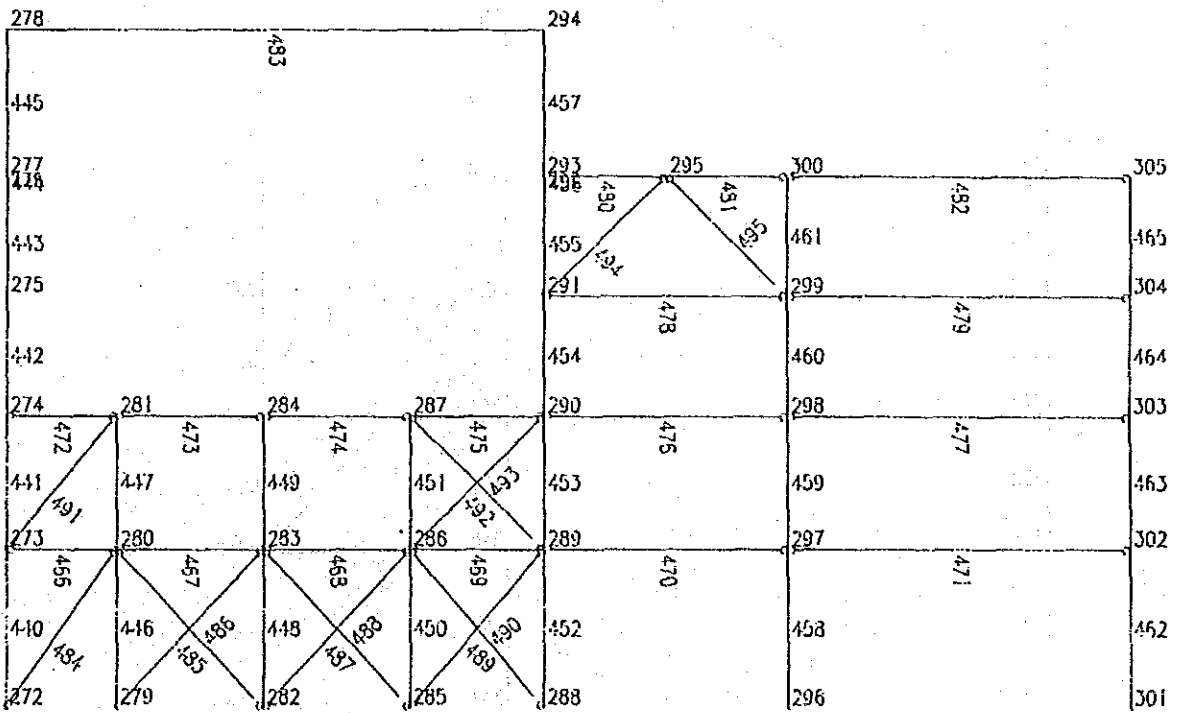


フレーム モデル ☒
 (108 -- LINE. SCALE 1/300)

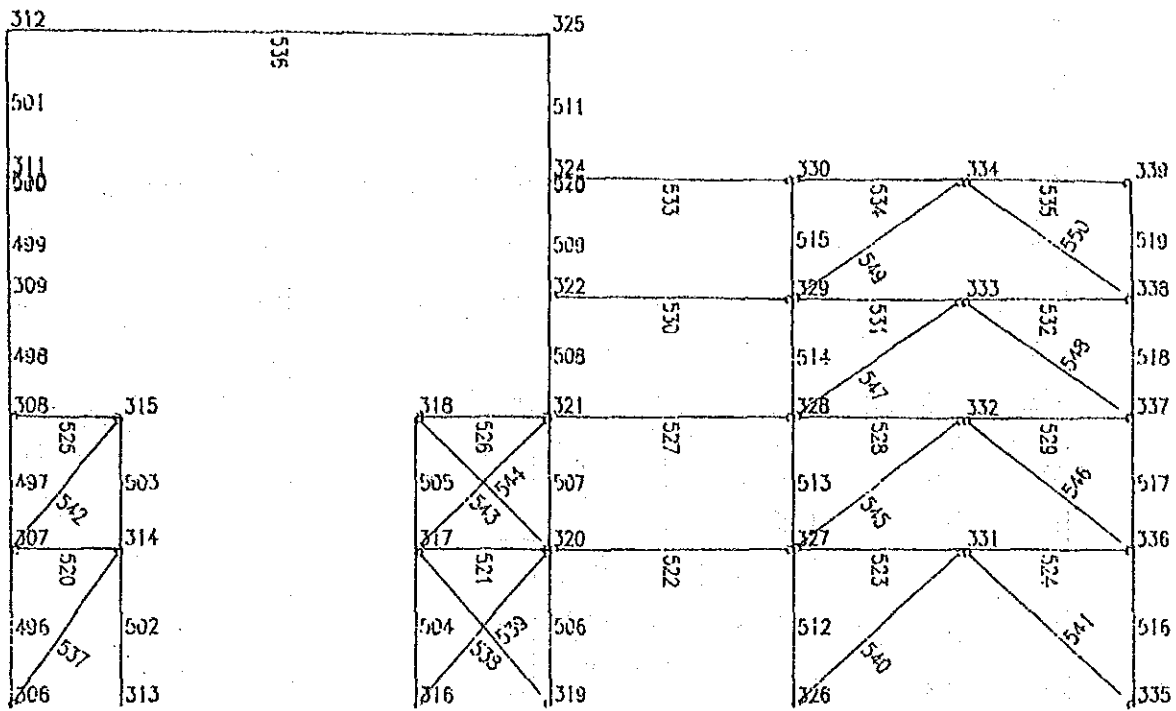
af



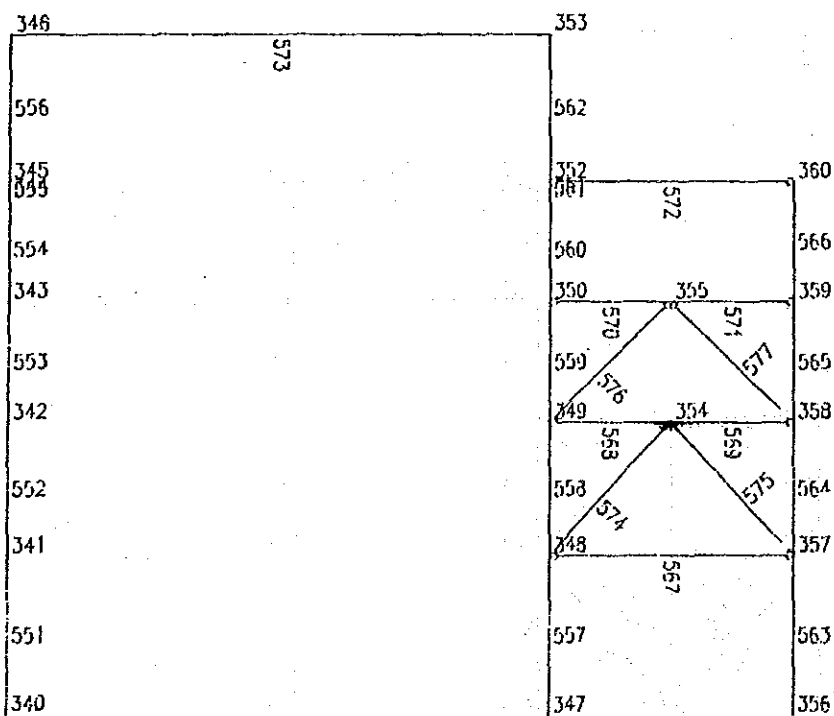
フレーム モデル 図
 (201 - LINE. SCALE. 1/300)



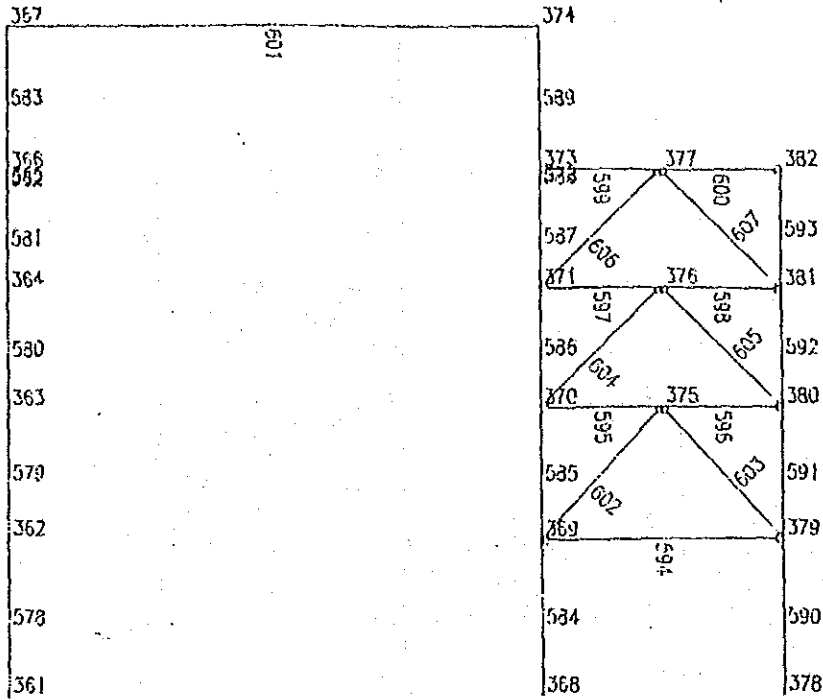
フレーム モデル 図
 (202 - LINE. SCALE. 1/300)




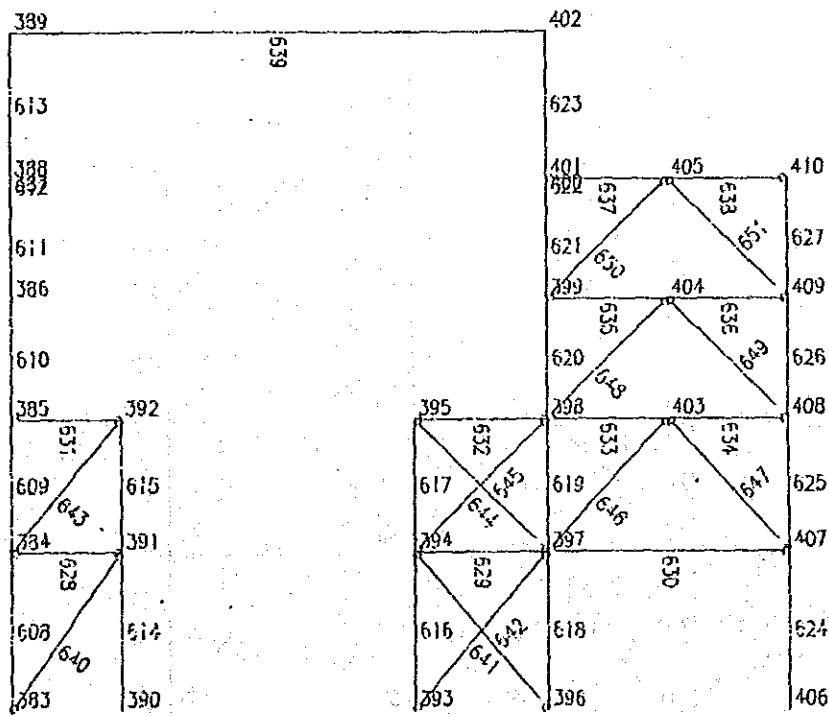
フレーム モデル ☒
 (203 - LINE SCALE 1/300)




フレーム モデル ☒
 (204 - LINE SCALE 1/300)

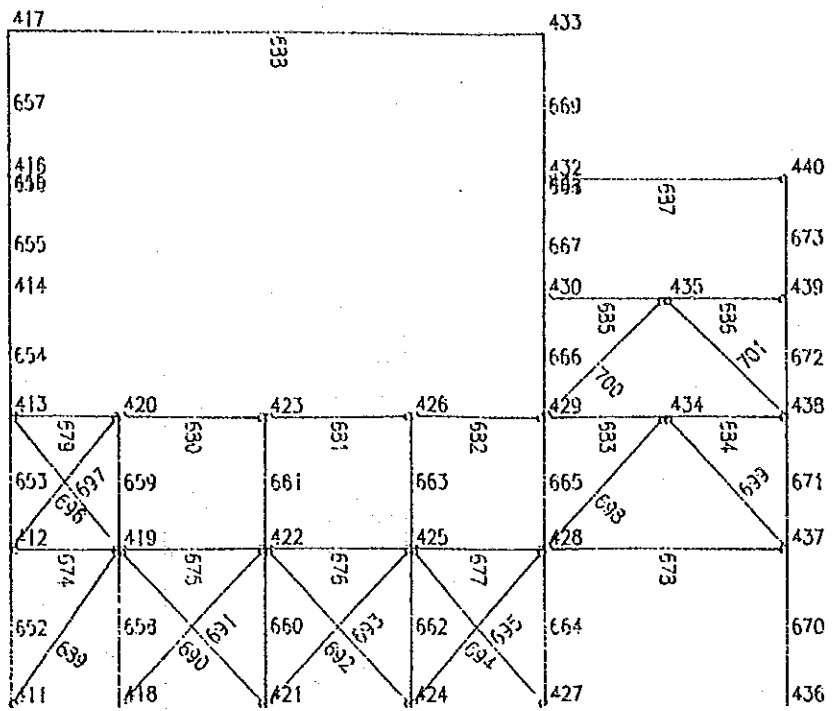


フレーム モデル 
 (205 - LINE SCALE 1/300)

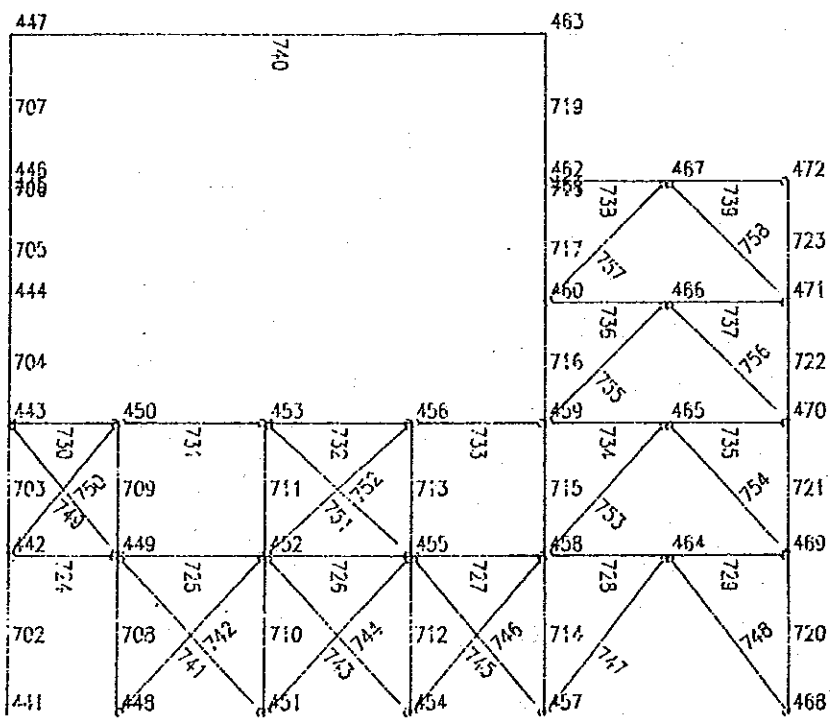


フレーム モデル 
 (206 - LINE SCALE 1/300)

10A

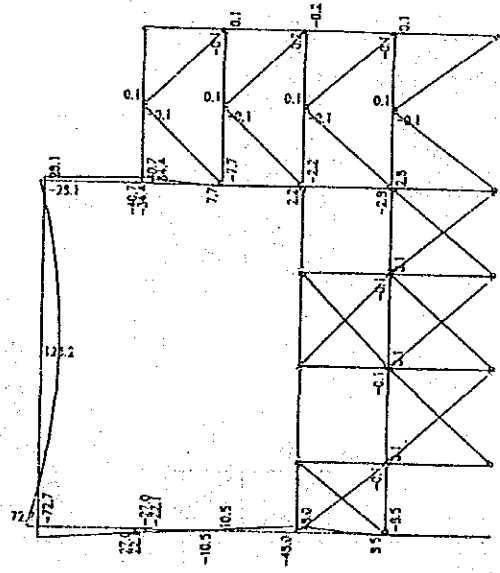
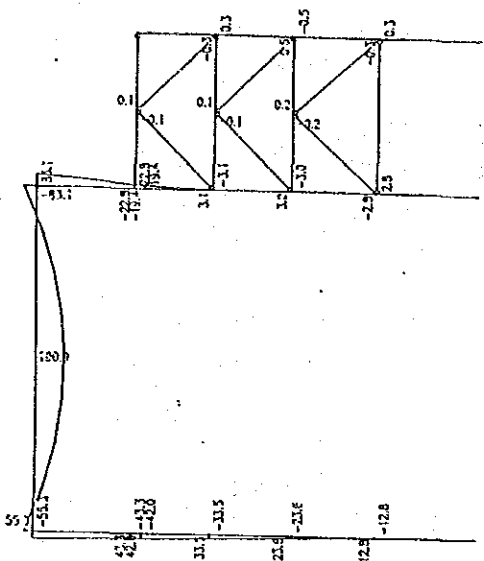
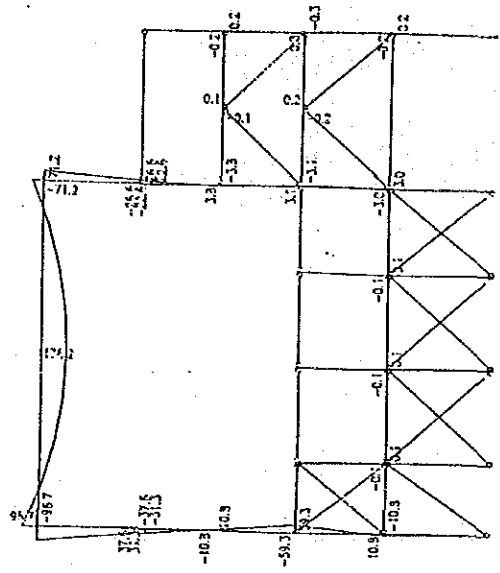


フレーム モデル ☒
 (207 -- LINE SCALE 1/300)

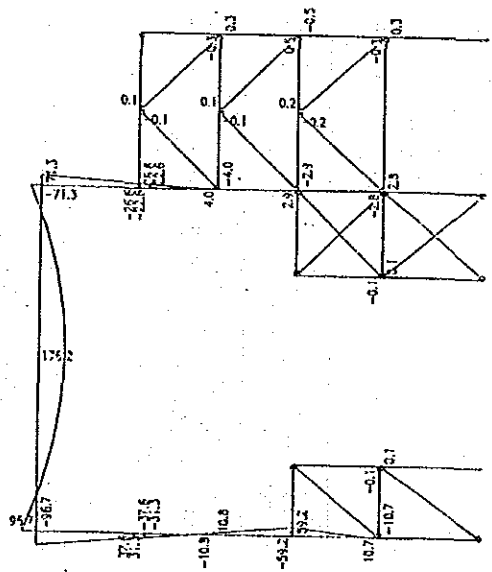


フレーム モデル ☒
 (208 -- LINE SCALE 1/300)

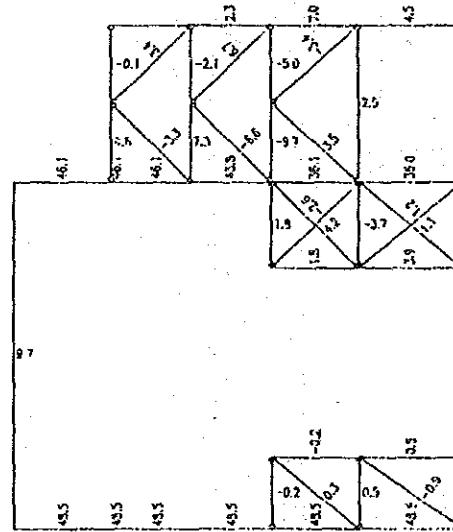
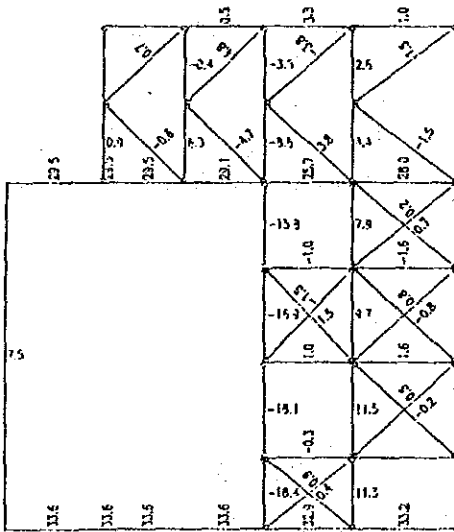
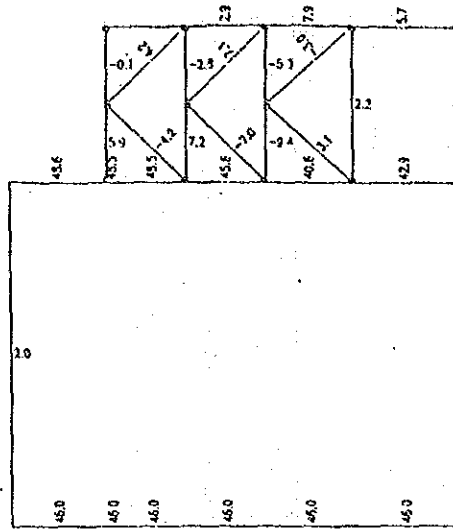
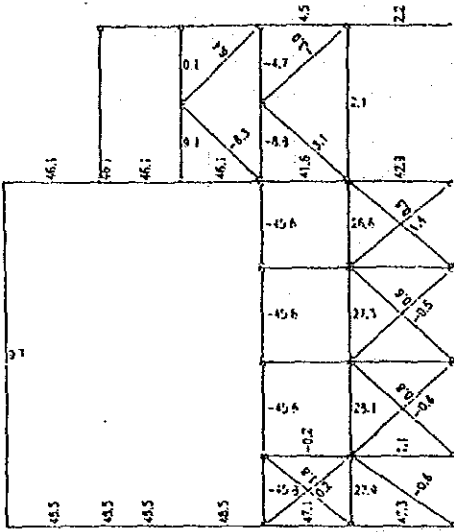
3.3 Stress



荷重ケース 1 B.M.D. 曲げモーメント図
(101-102 - LINE)

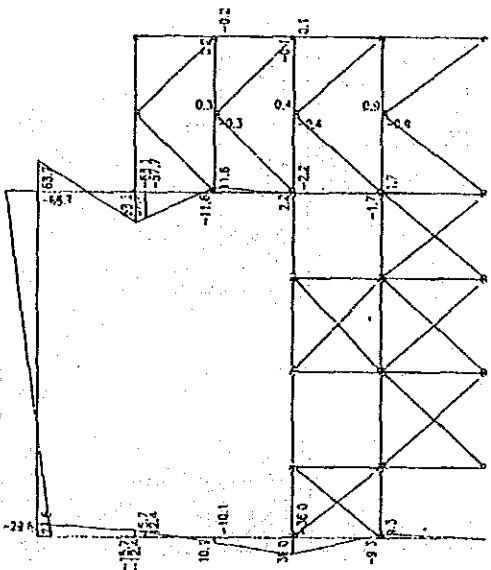
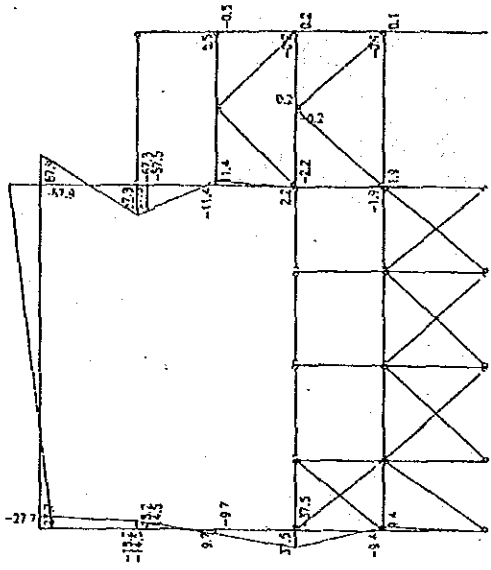


荷重ケース 1 B.M.D. 曲げモーメント図
(103-104 - LINE)

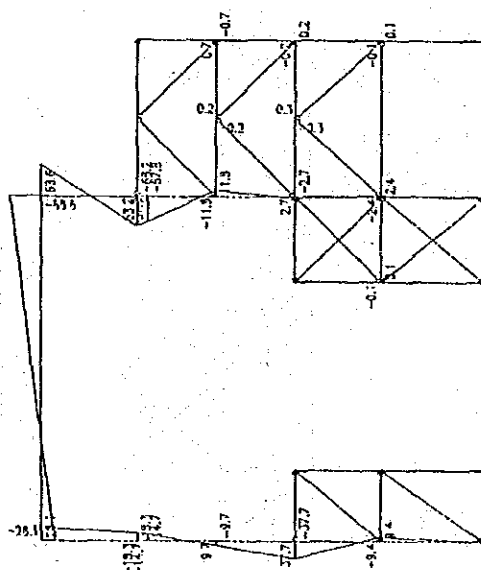
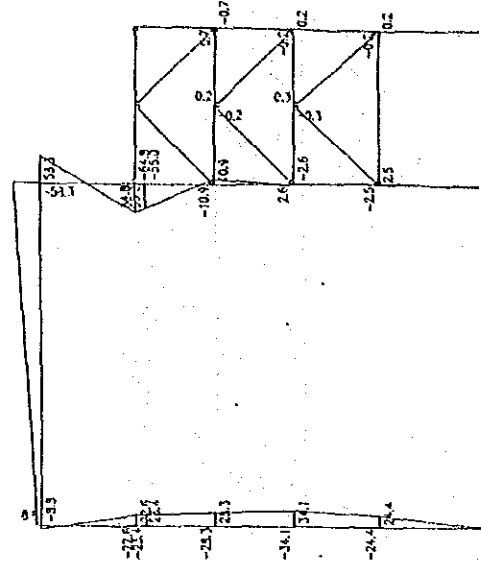


荷重ケース 1 軸力図
(101-102 - LINE)

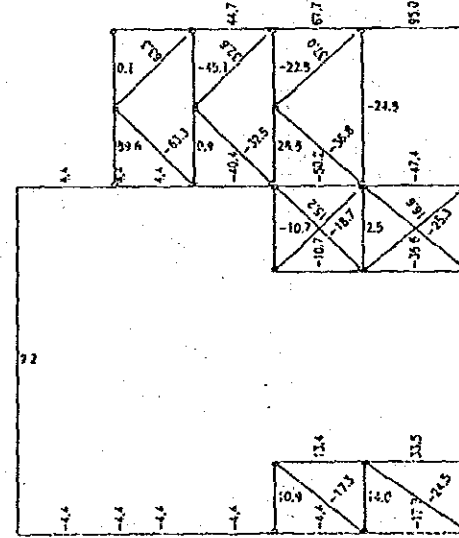
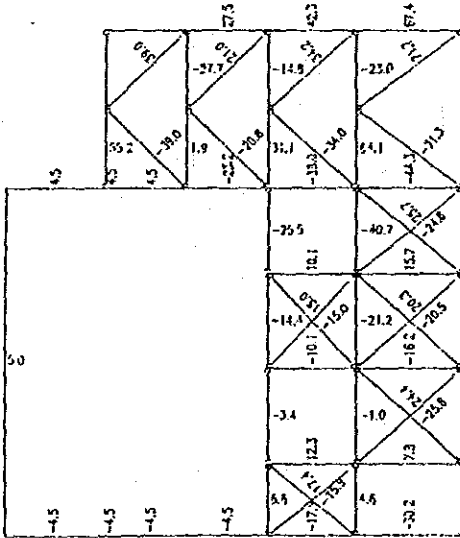
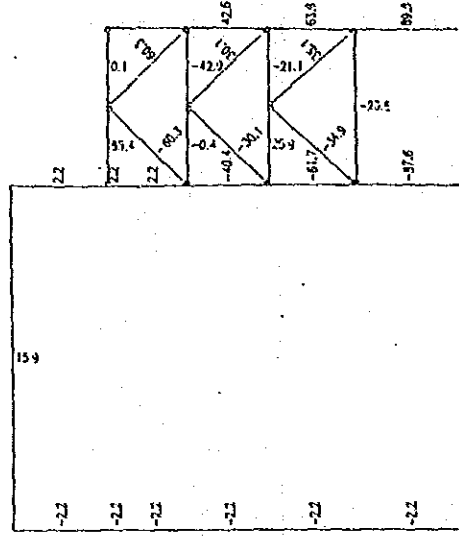
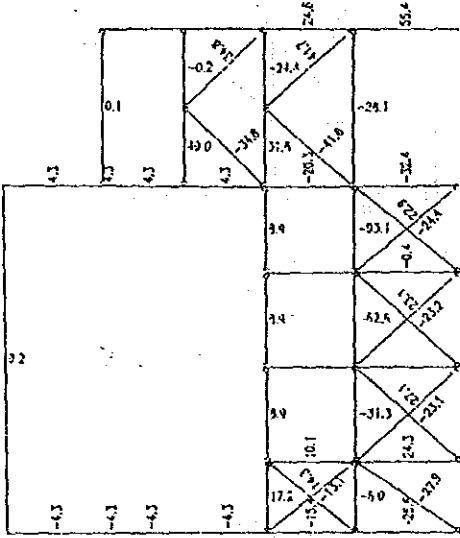
荷重ケース 1 軸力図
(103-104 - LINE)



荷重ケース 2 B.M.D. 曲げモーメント図
(101-102 - LINE)

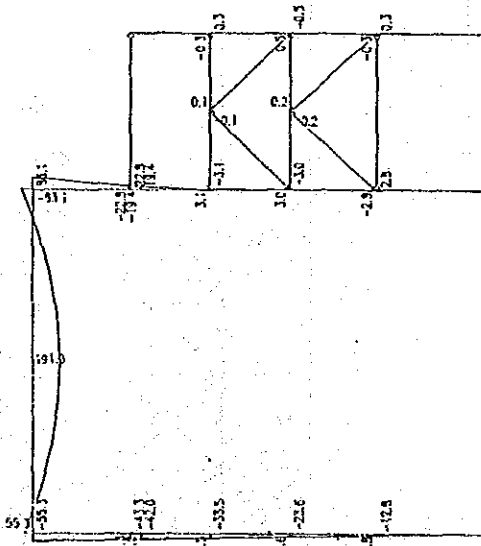
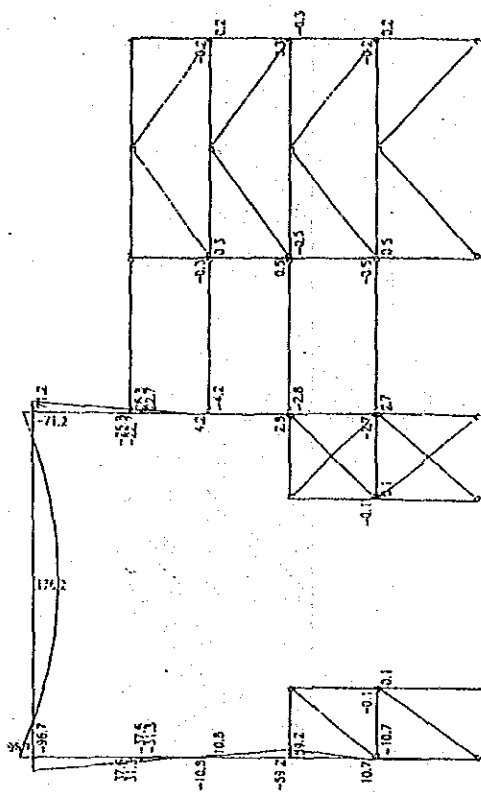


荷重ケース 2 B.M.D. 曲げモーメント図
(103-104 - LINE)

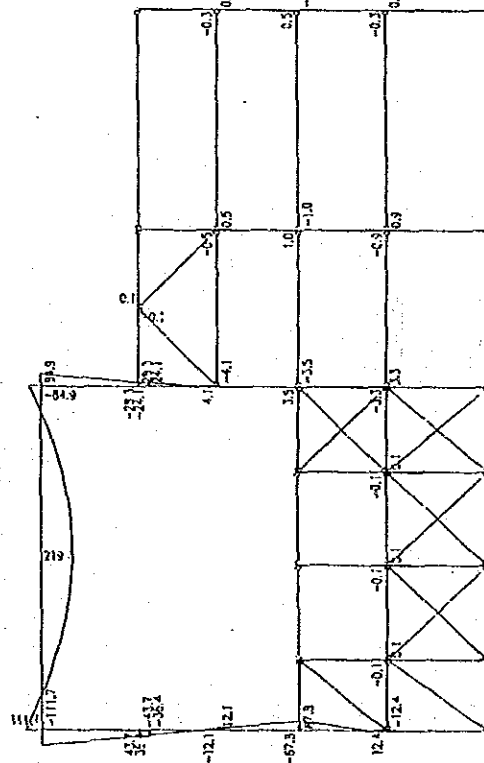
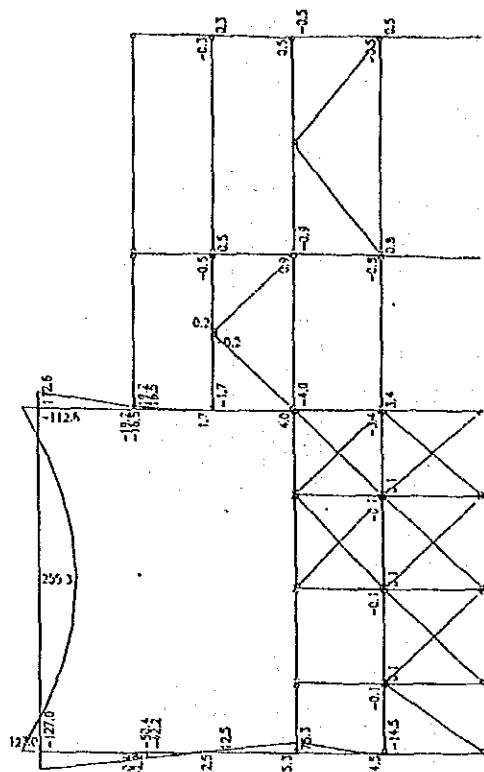


A.F.D. 図
荷重ケース 2 軸 力 図
(101-102 - LINE)

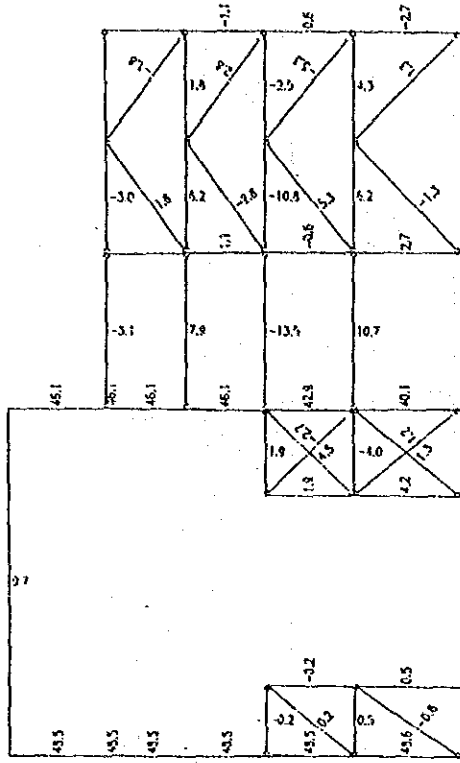
A.F.D. 図
荷重ケース 2 軸 力 図
(103-104 - LINE)



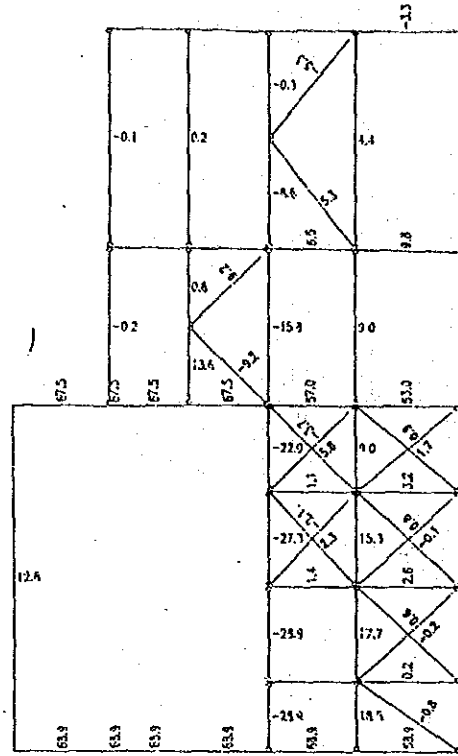
荷重ケース 1 B.M.D. 曲げモーメント図
(105-106 - LINE)



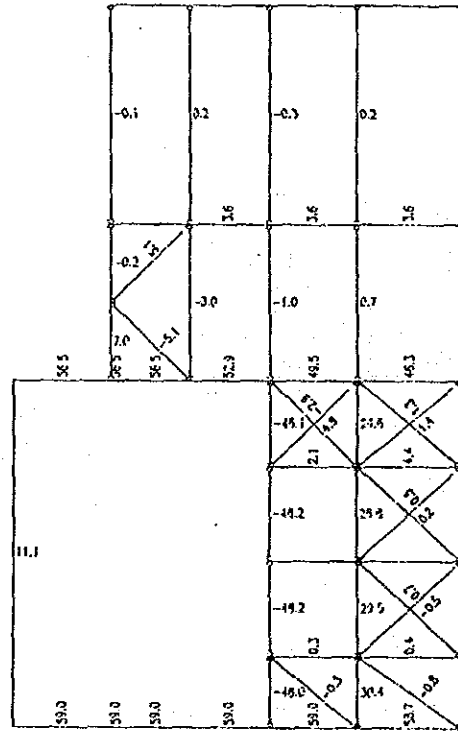
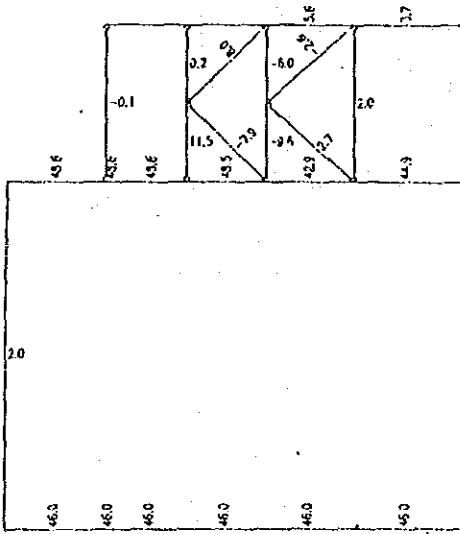
荷重ケース 1 B.M.D. 曲げモーメント図
(107-103 - LINE)



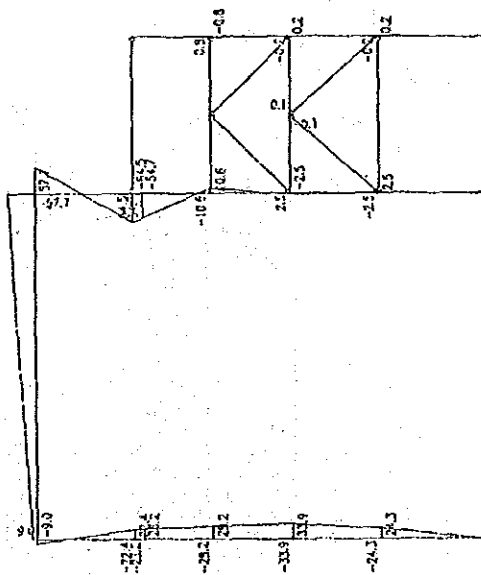
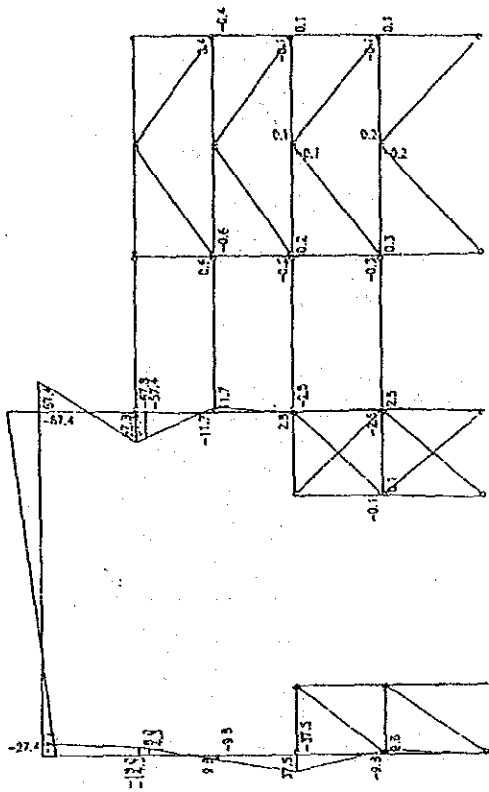
荷重ケース 1 軸 A.F.D. 図
(105-106 - LINE)



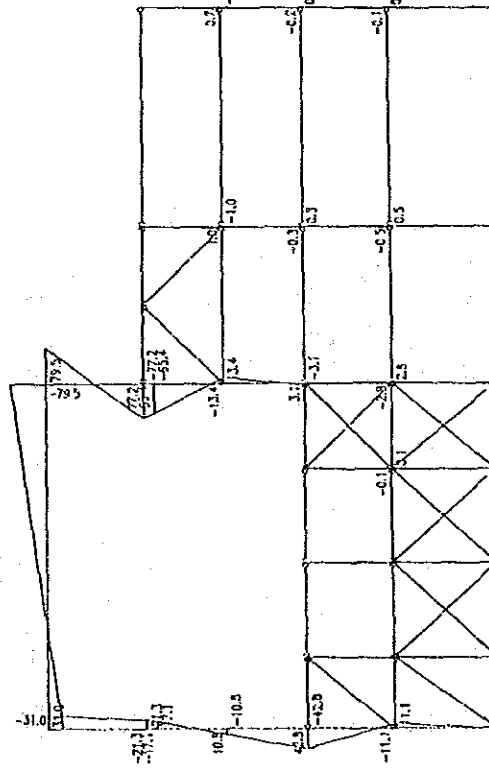
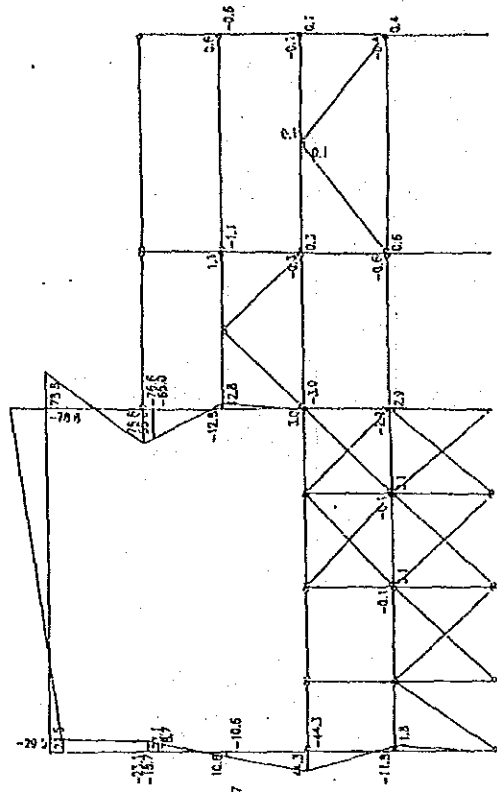
荷重ケース 1 軸 A.F.D. 図
(107-108 - LINE)



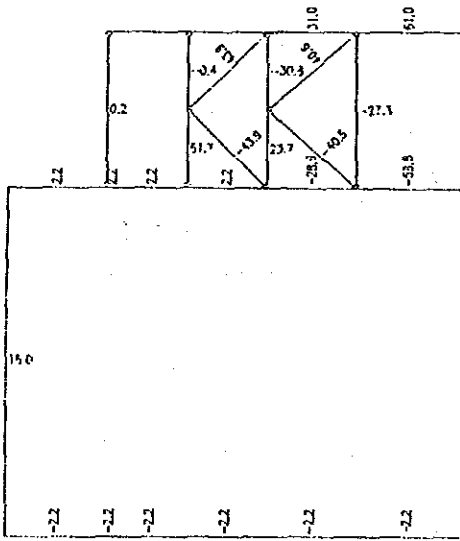
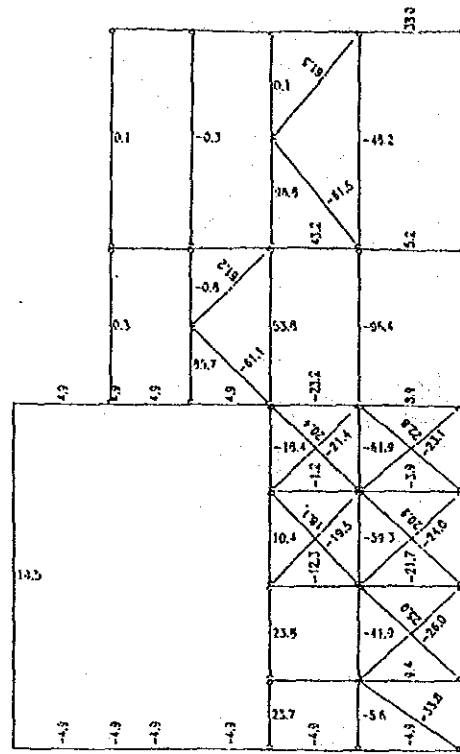
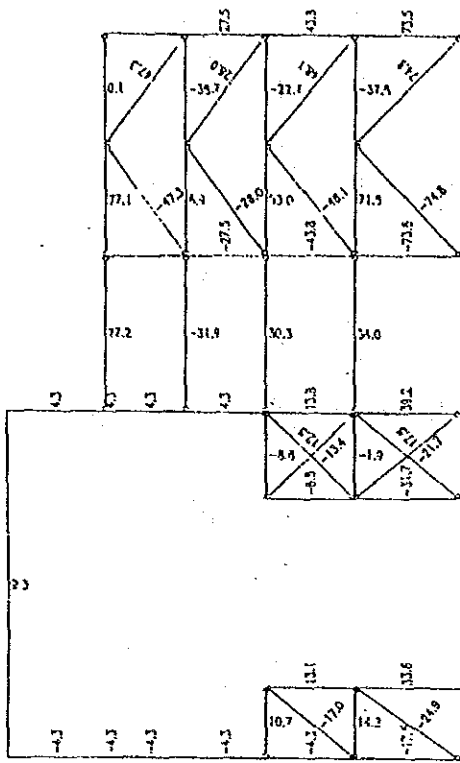
112



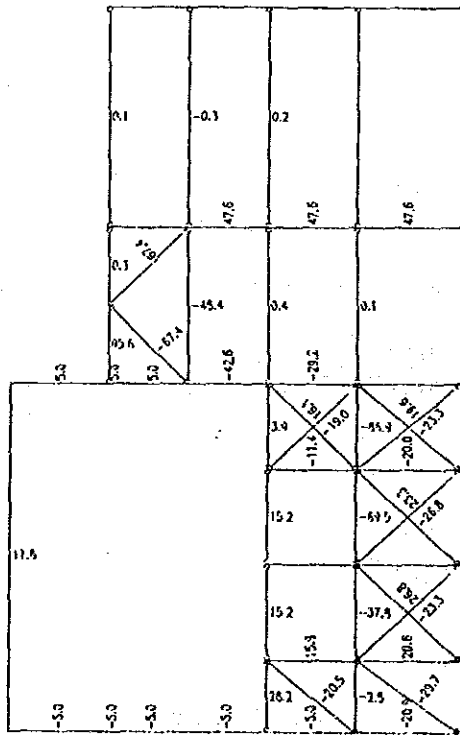
荷重ケース 2 曲げモーメント図
(105-106 - LINE)



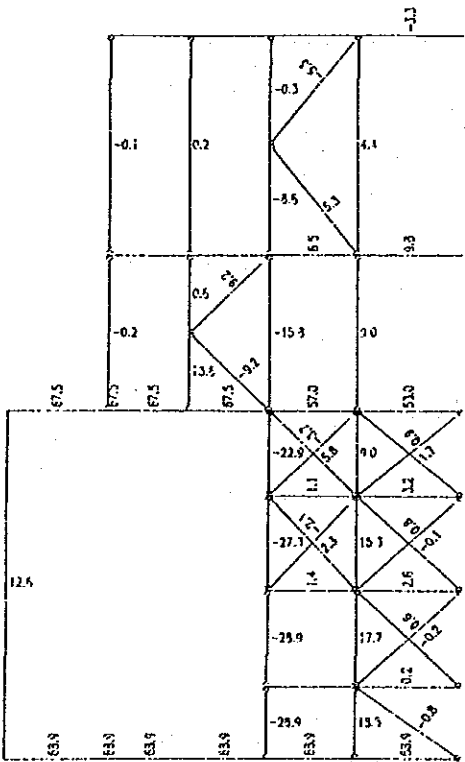
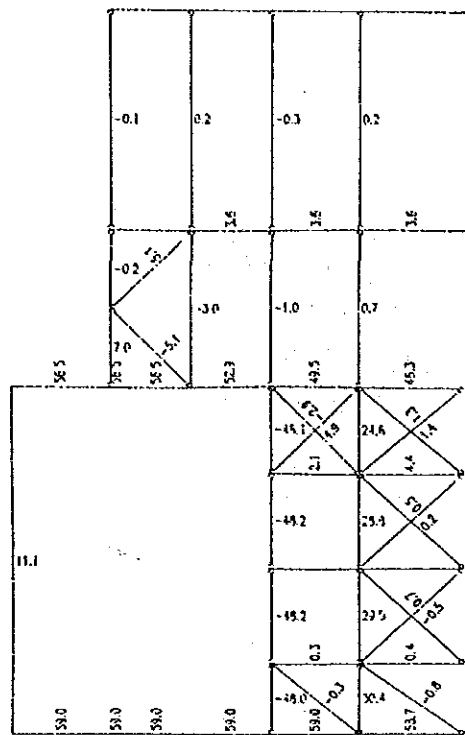
荷重ケース 2 曲げモーメント図
(107-108 - LINE)



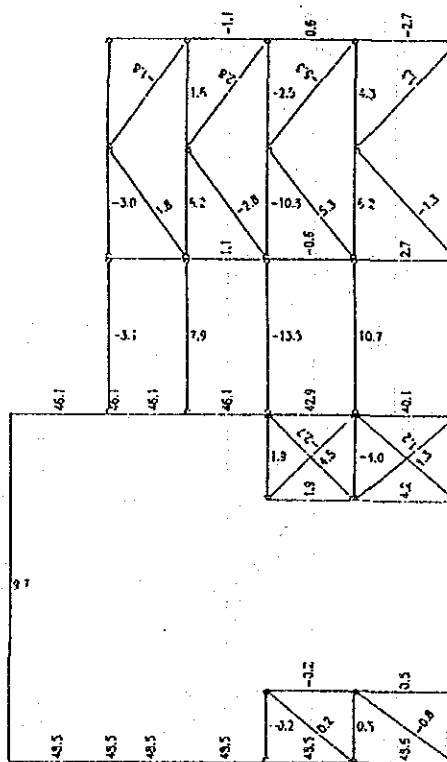
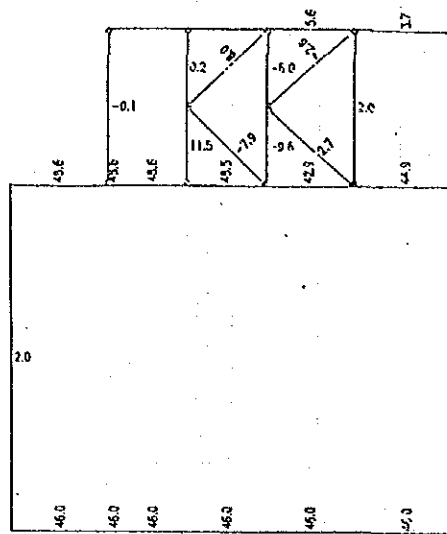
荷重ケース 2 軸 A.F.D. 図
(105-106 - LINE)



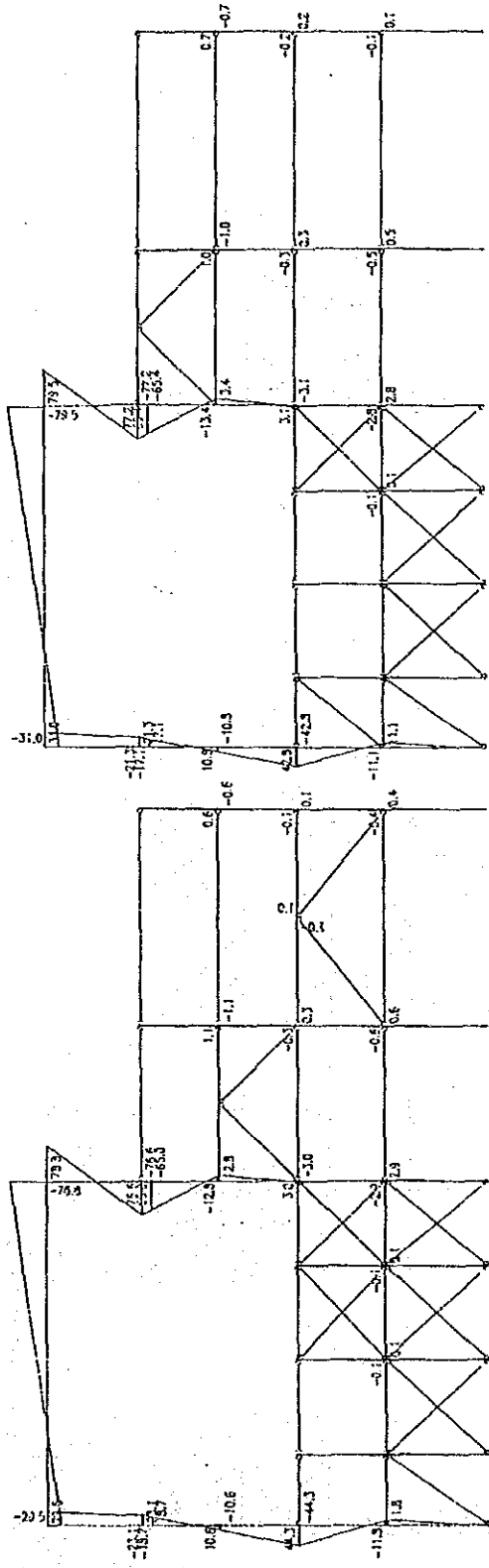
荷重ケース 2 軸 A.F.D. 図
(107-108 - LINE)



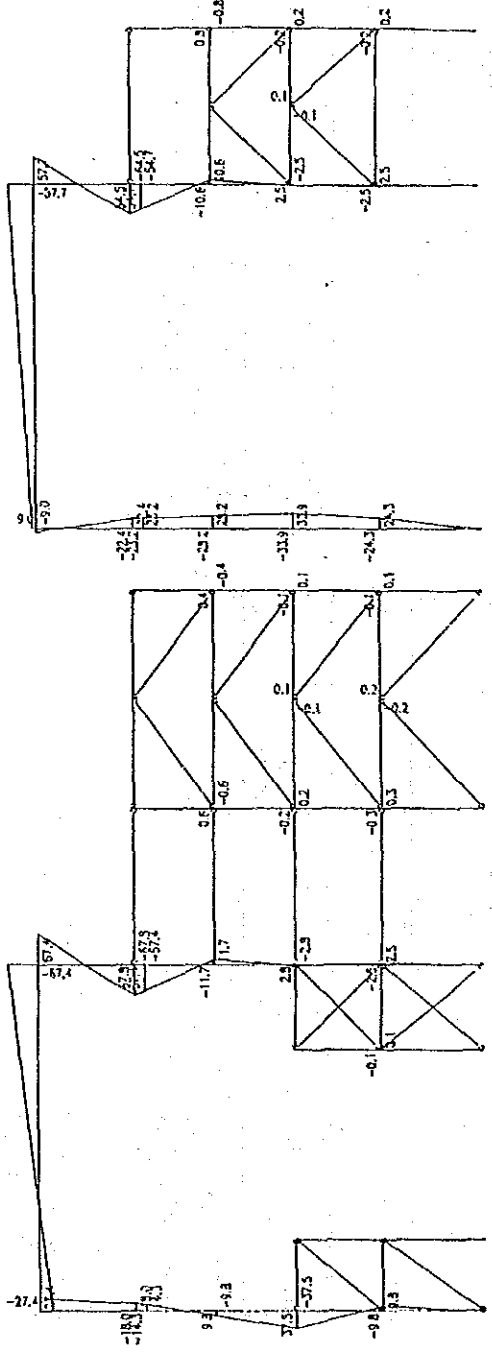
荷重ケース 1 軸力図 (A.F.D.)
(201-202 - LINE)



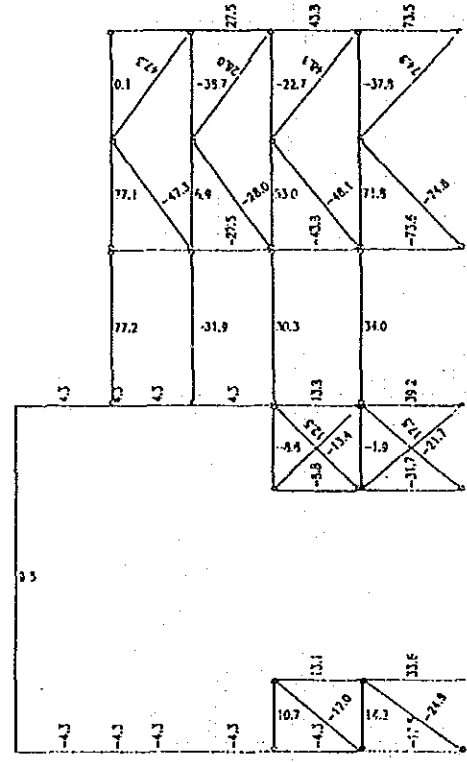
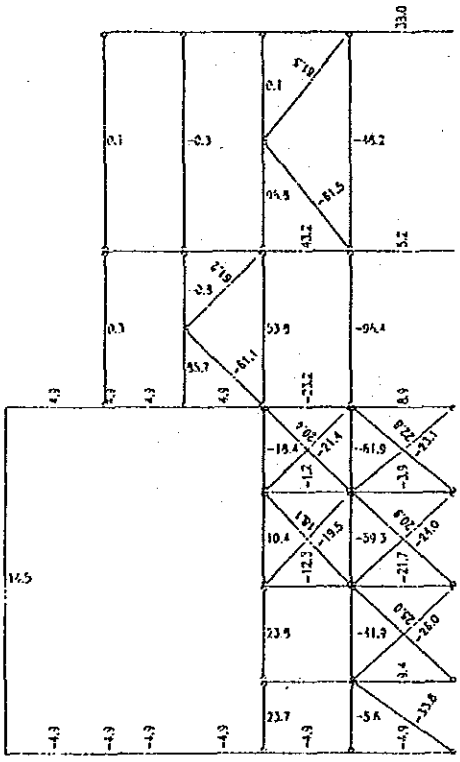
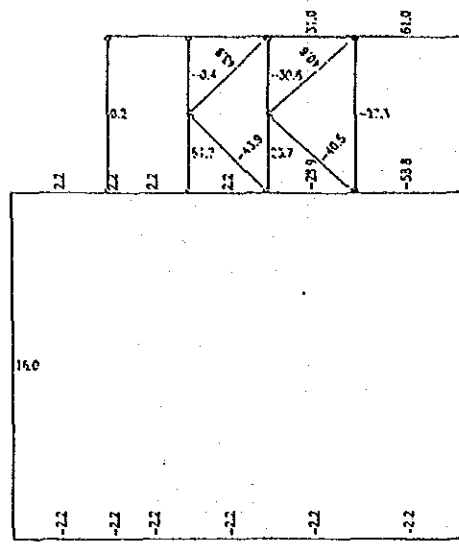
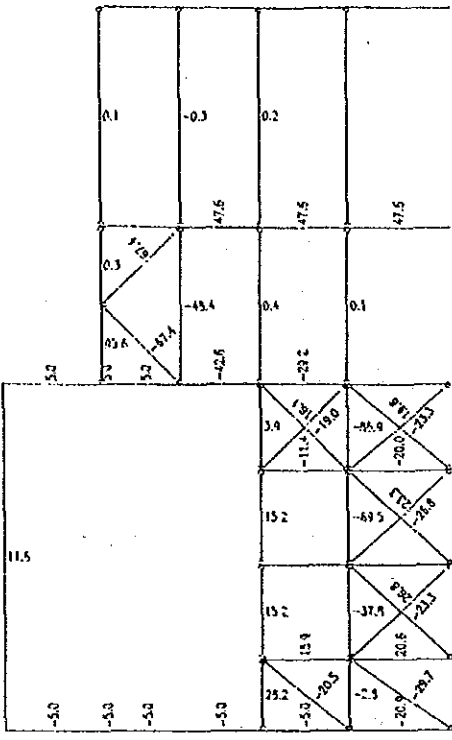
荷重ケース 1 軸力図 (A.F.D.)
(203-204 - LINE)



荷重ケース 2 B.M.D. 曲げモーメント図
(201-202 - LINE)

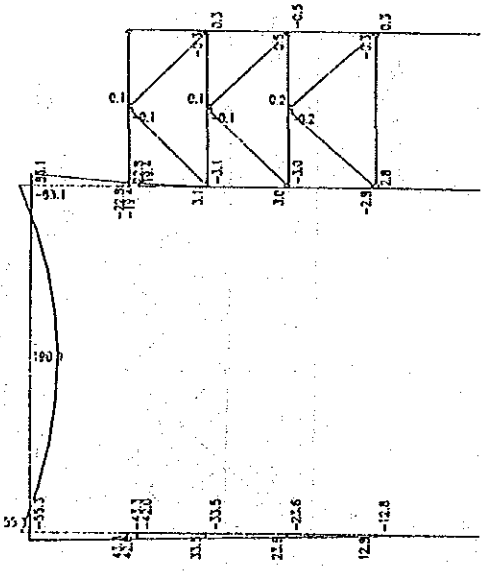
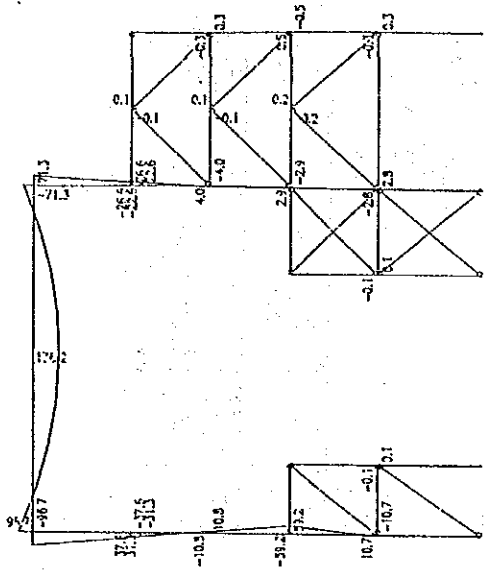


荷重ケース 2 B.M.D. 曲げモーメント図
(203-204 - LINE)

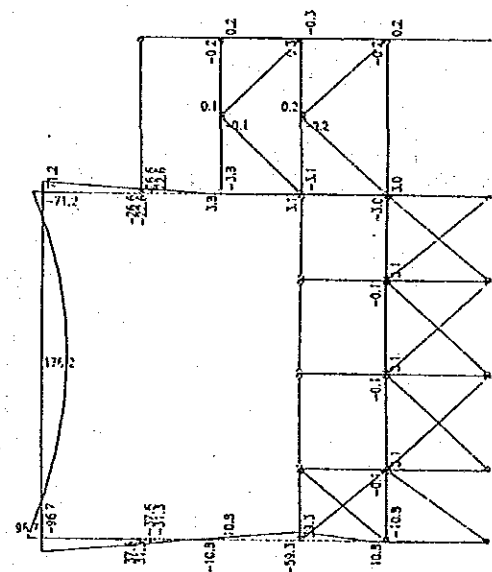
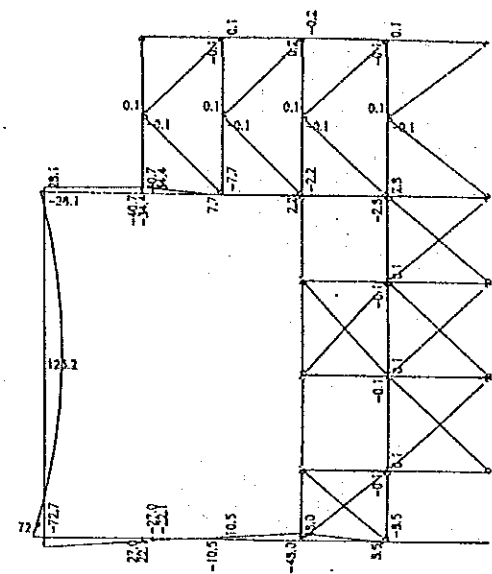


荷重ケース 2 軸力図
(201-202 - LINE)

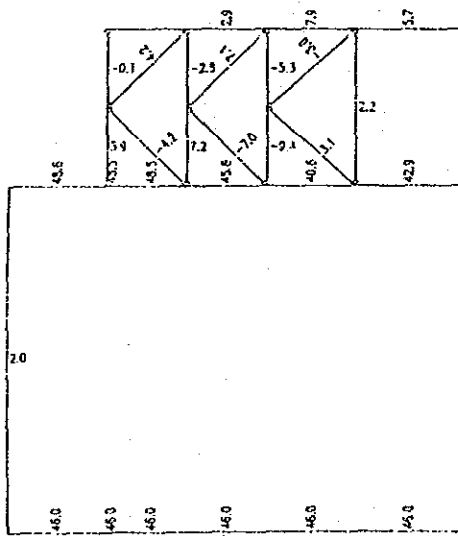
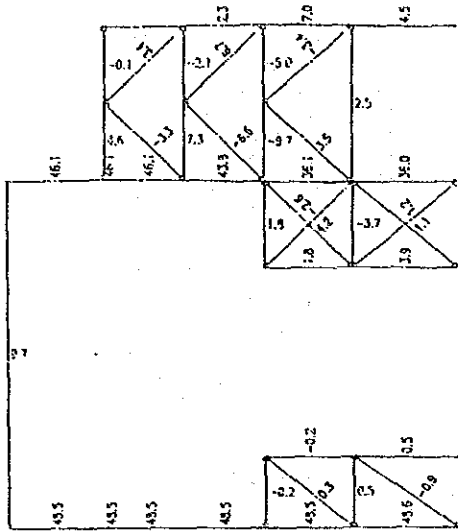
荷重ケース 2 軸力図
(203-204 - LINE)



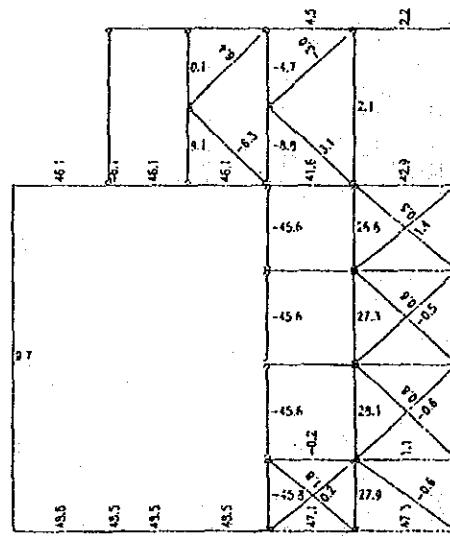
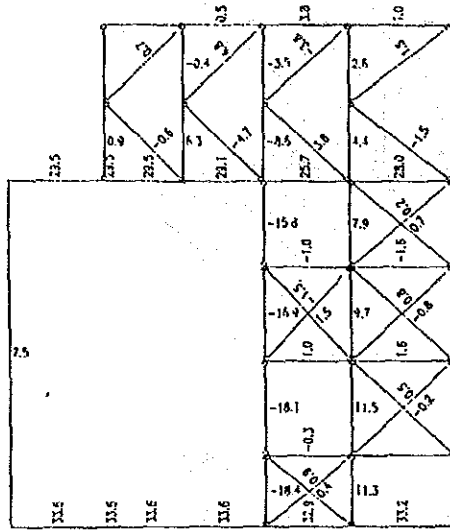
荷重ケース 1 B.M.D. 曲げモーメント図
(205-206 - LINE)



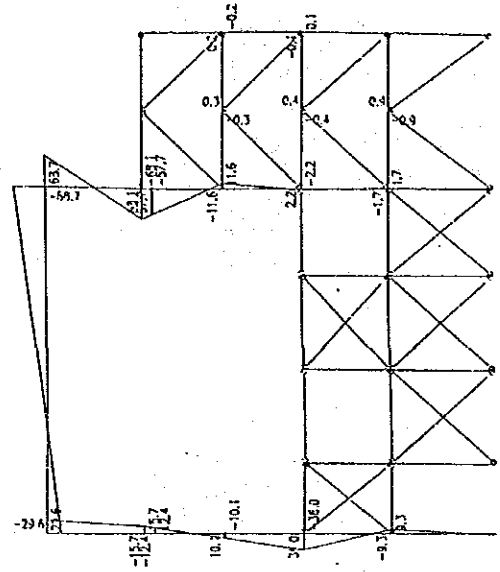
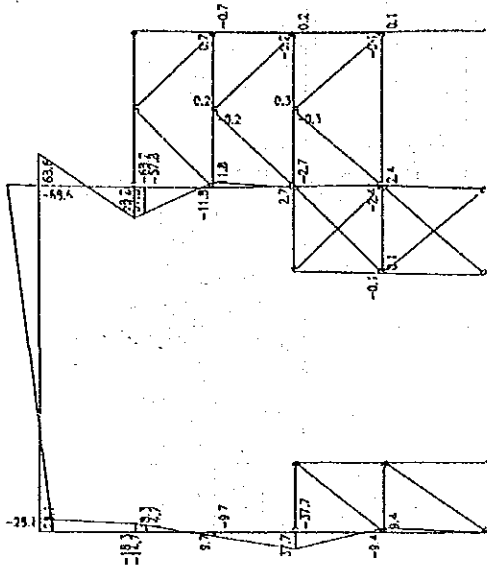
荷重ケース 1 B.M.D. 曲げモーメント図
(207-208 - LINE)



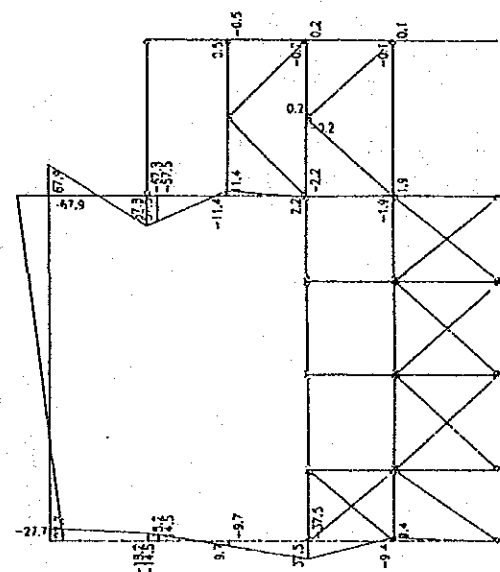
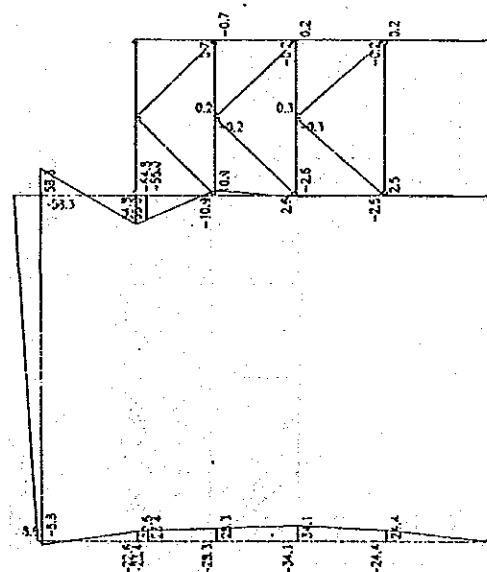
荷重ケース 1 軸力図
(205-206 - LINE)



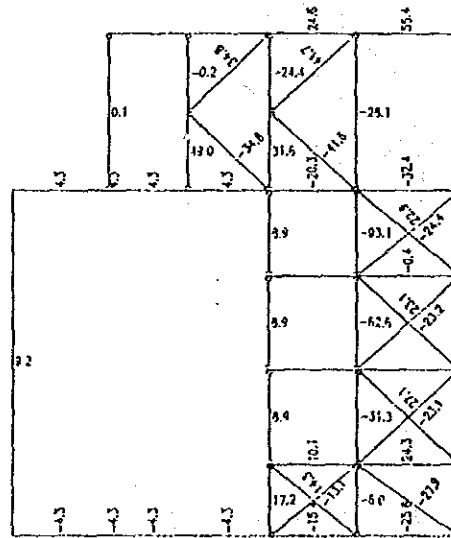
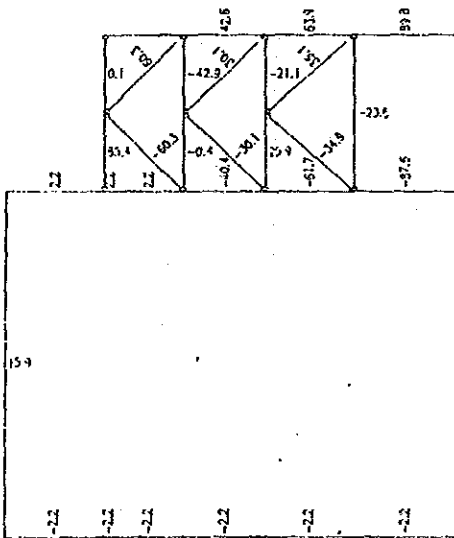
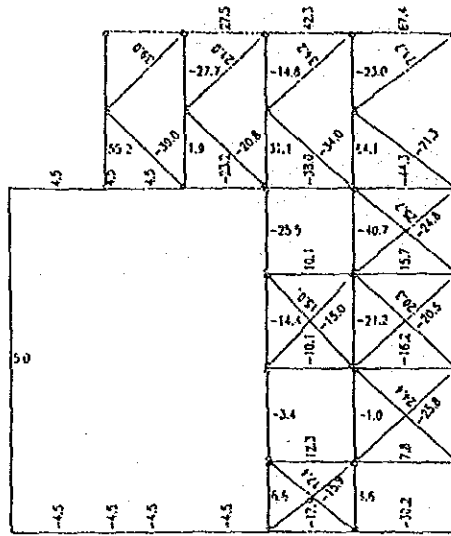
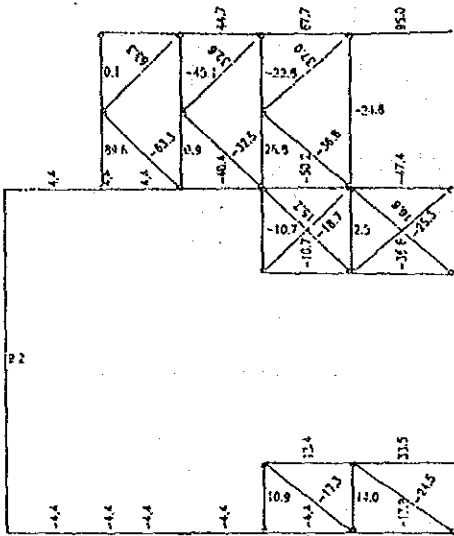
荷重ケース 1 軸力図
(207-208 - LINE)



荷重ケース 2 B.M.D. 曲げモーメント図
(205-206 - LINE)



荷重ケース 2 B.M.D. 曲げモーメント図
(207-208 - LINE)

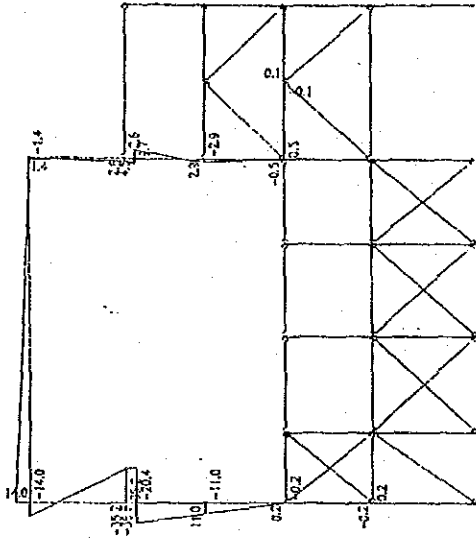


A.F.D. 図

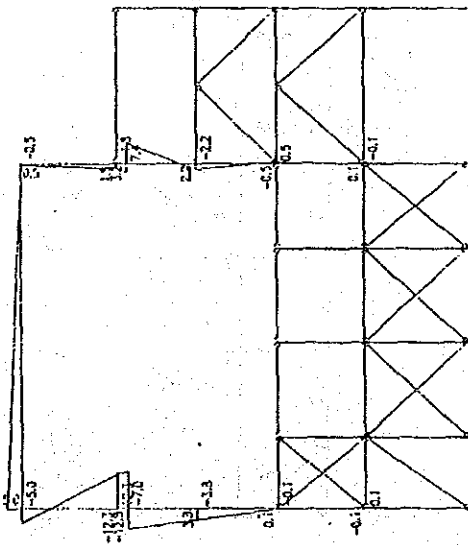
荷重ケース 2 軸
(205-206 - LINE)

A.F.D. 図

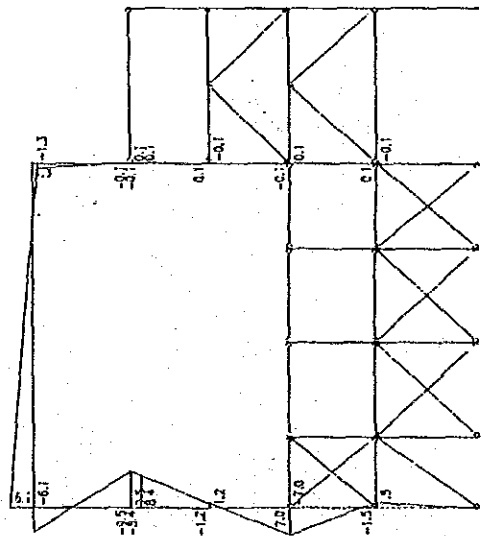
荷重ケース 2 軸
(207-208 - LINE)



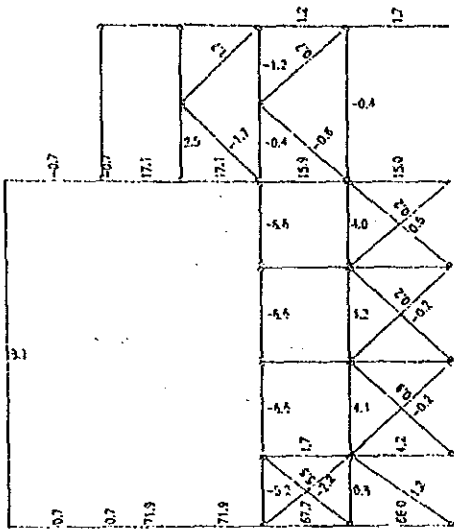
C.I.L.L. B.M.D. 曲げモーメント図
荷重ケース 2 (102 - LINE)



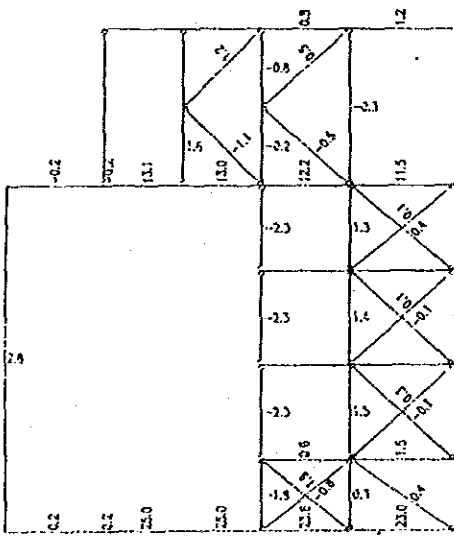
C.I.L.L. B.M.D. 曲げモーメント図
荷重ケース 1 (102 - LINE)



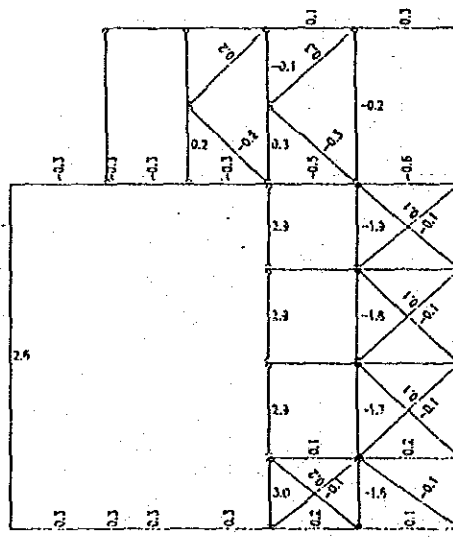
C.I.L.L. B.M.D. 曲げモーメント図
荷重ケース 3 (102 - LINE)



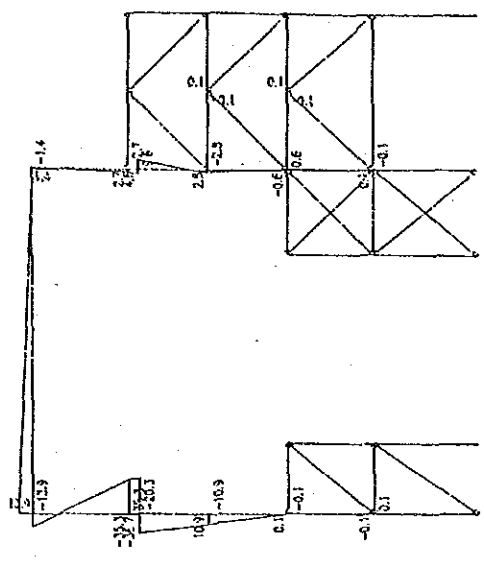
C.L.L. 2 軸 A.F.D. 図
荷重ケース 2 (102-LINE)



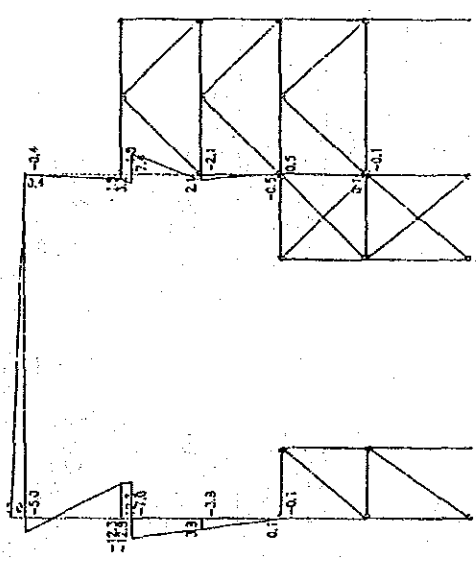
C.L.L. 1 軸 A.F.D. 図
荷重ケース 1 (102-LINE)



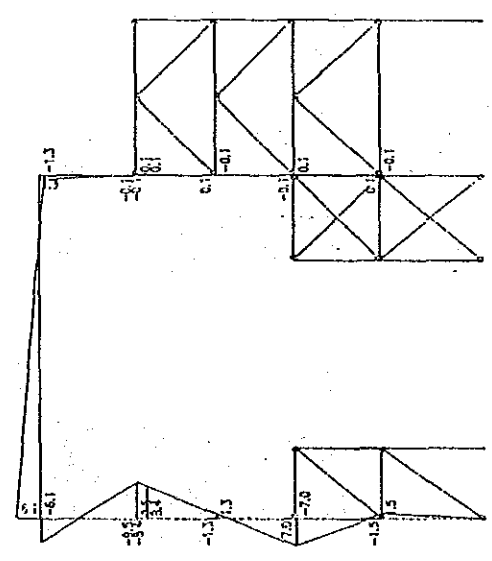
C.H.L. 3 軸 A.F.D. 図
荷重ケース 3 (102-LINE)



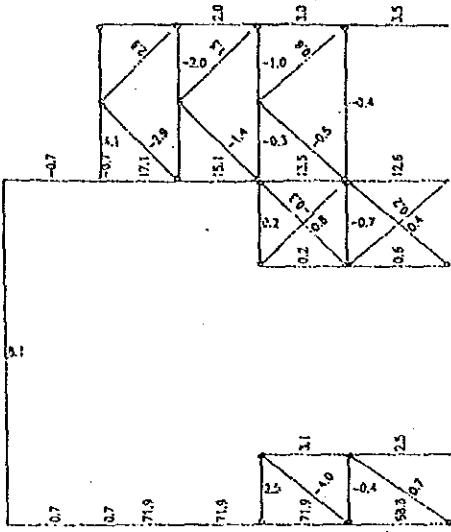
C.L.L. B.M.D.
荷重ケース 2 曲げモーメント図
(103 -- LINE)



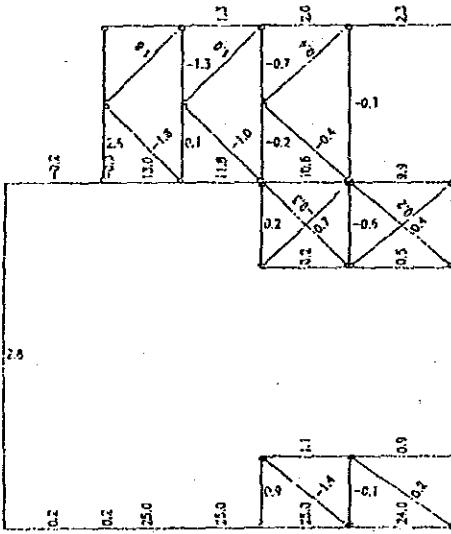
C.D.L. B.M.D.
荷重ケース 1 曲げモーメント図
(103 -- LINE)



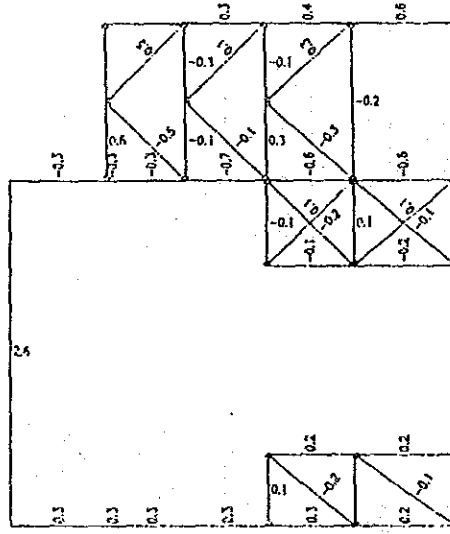
C.H.L. B.M.D.
荷重ケース 3 曲げモーメント図
(103 -- LINE)



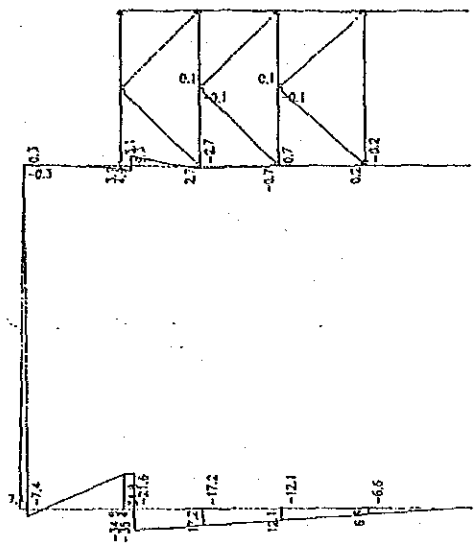
C.L.L. A.F.D. 軸 力 図
荷重ケース 2 (103 - LINE)



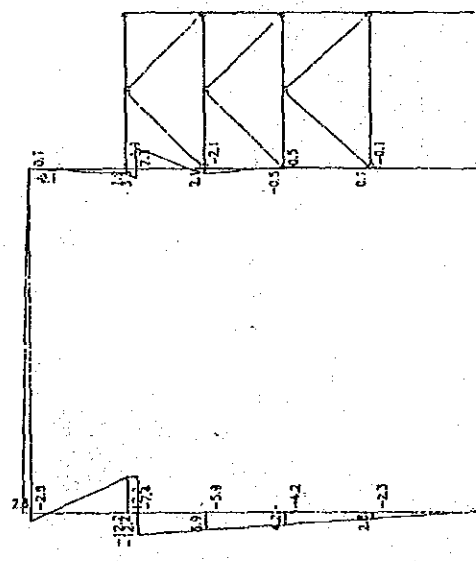
C.L.L. A.F.D. 軸 力 図
荷重ケース 1 (103 - LINE)



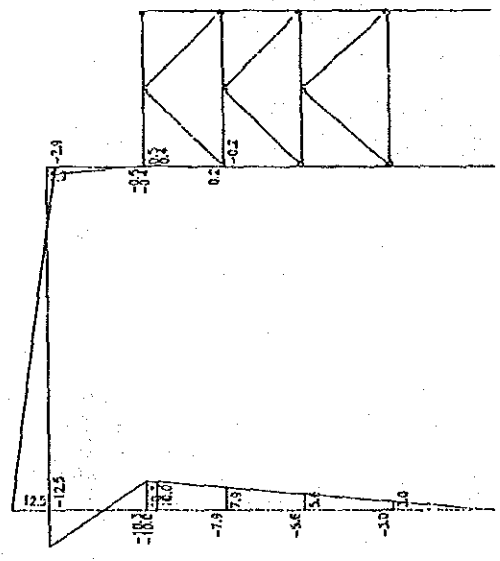
C.L.L. A.F.D. 軸 力 図
荷重ケース 3 (103 - LINE)



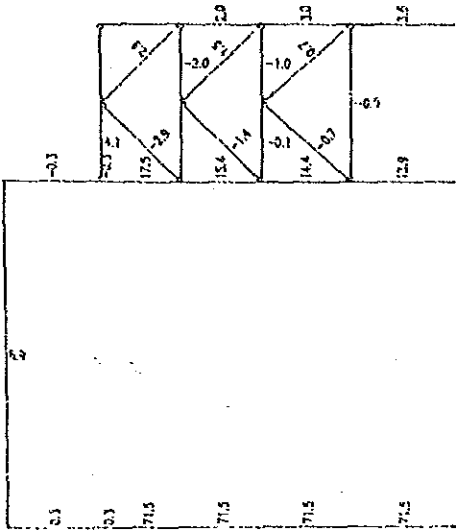
C.L.L. 荷重ケース 2 (10t-LINE) B.M.D. 曲げモーメント図



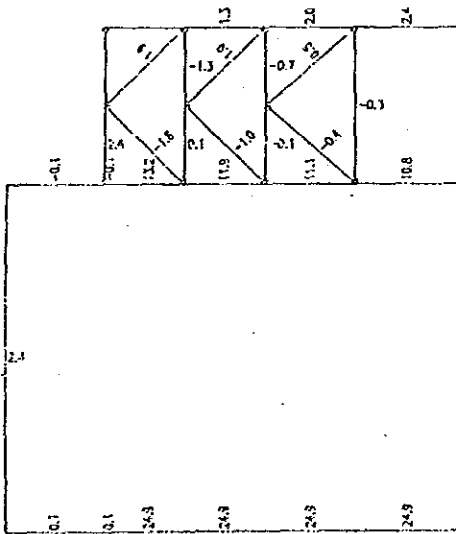
C.L.L. 荷重ケース 1 (10t-LINE) B.M.D. 曲げモーメント図



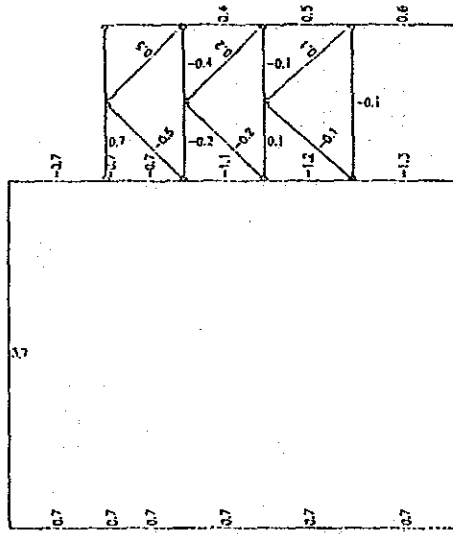
C.H.L. 荷重ケース 3 (10t-LINE) B.M.D. 曲げモーメント図



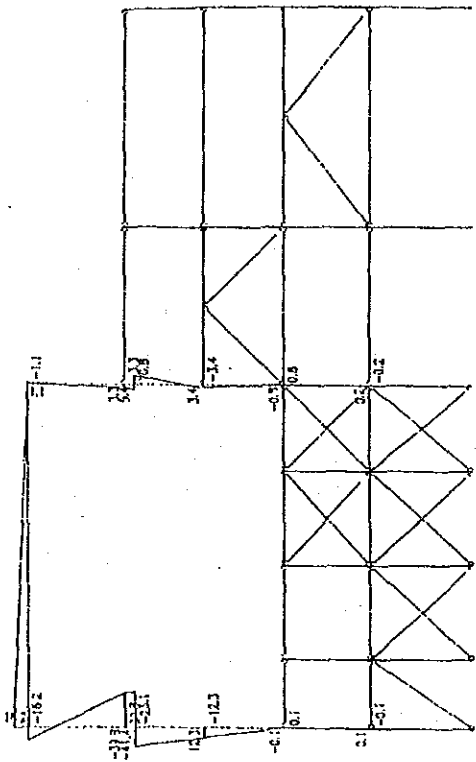
C.H.L. 荷重ケース 2 軸 A.F.D. 図
(104 - LINE)



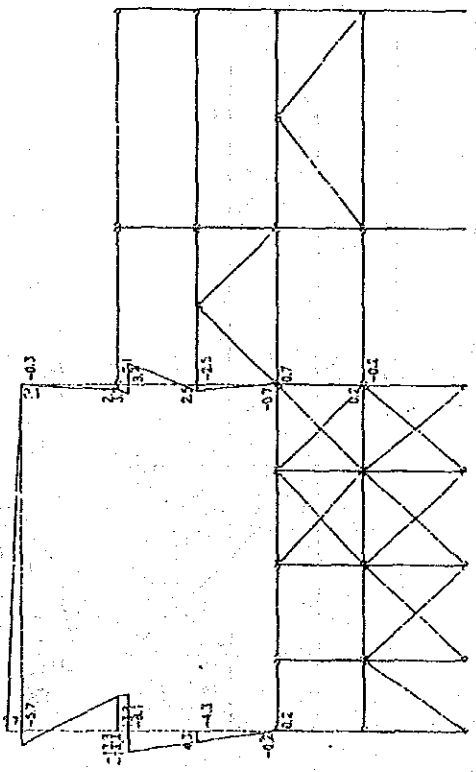
C.H.L. 荷重ケース 1 軸 A.F.D. 図
(104 - LINE)



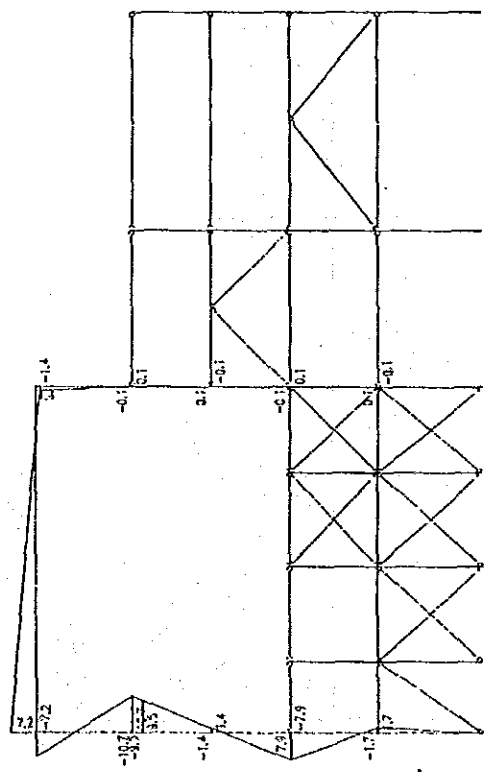
C.H.L. 荷重ケース 3 軸 A.F.D. 図
(104 - LINE)



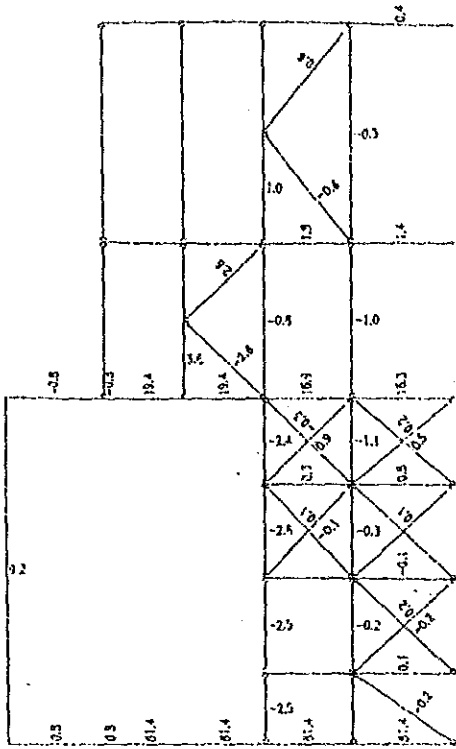
C.L.L. 2 B.M.D. 曲げモーメント図
(108 - LINE)



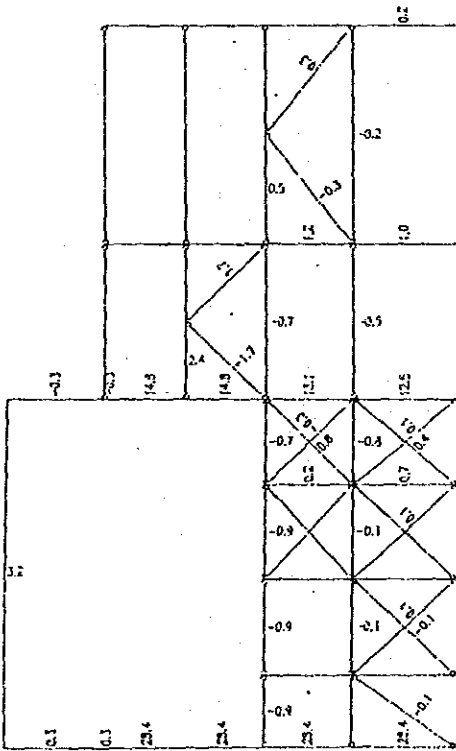
C.L.L. 1 B.M.D. 曲げモーメント図
(108 - LINE)



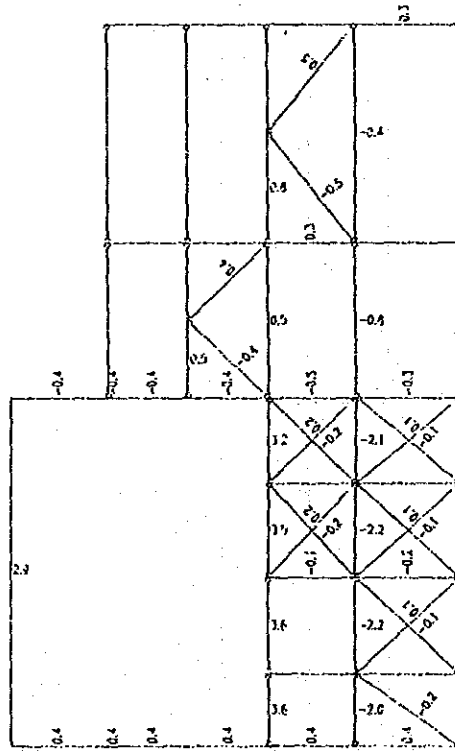
C.L.L. 3 B.M.D. 曲げモーメント図
(108 - LINE)



C.D.L. A.F.D. 荷重ゲージ 2 (108 - LINE) 図

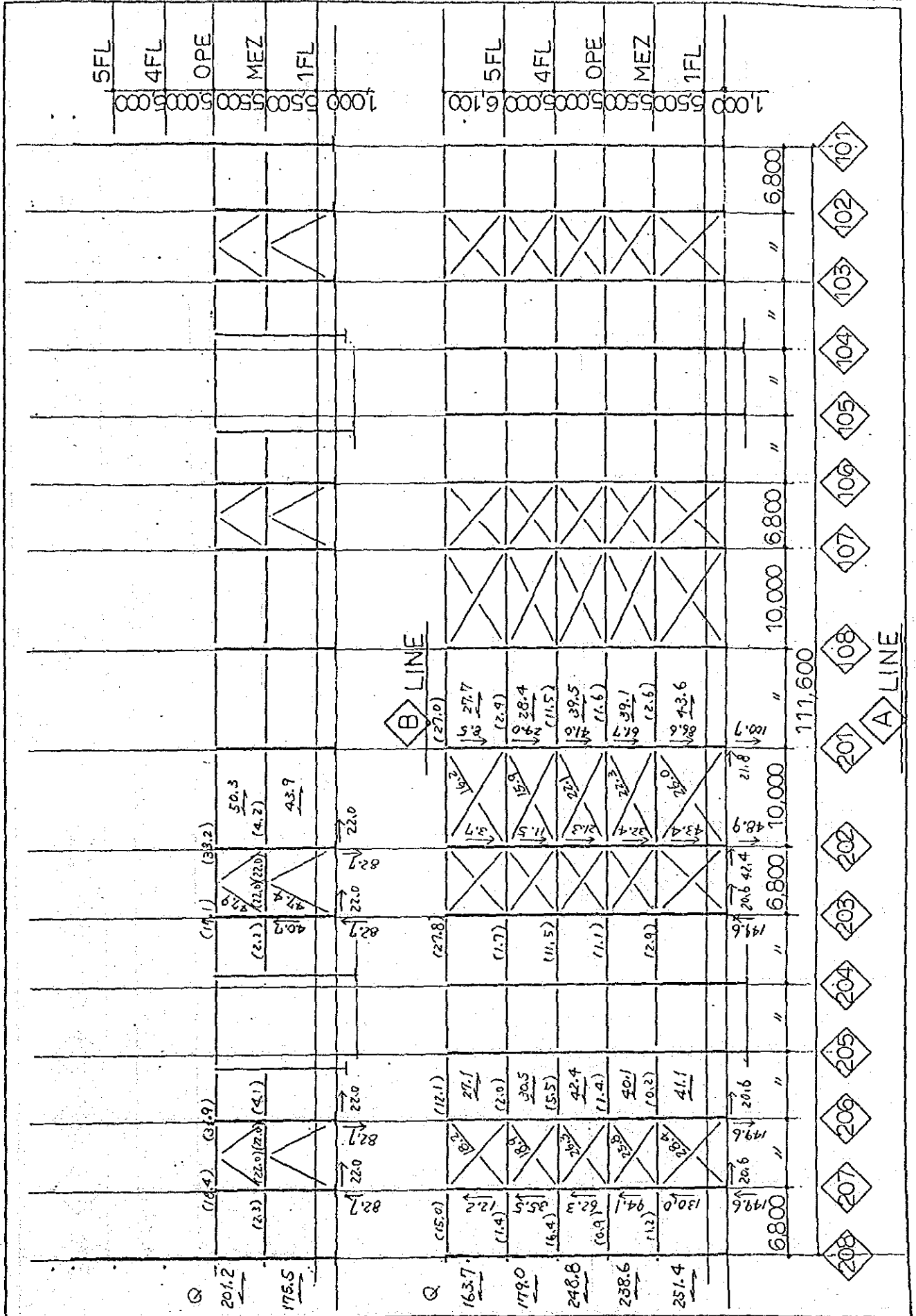


C.D.L. A.F.D. 荷重ゲージ 1 (108 - LINE) 図



C.H.L. A.F.D. 荷重ゲージ 3 (108 - LINE) 図

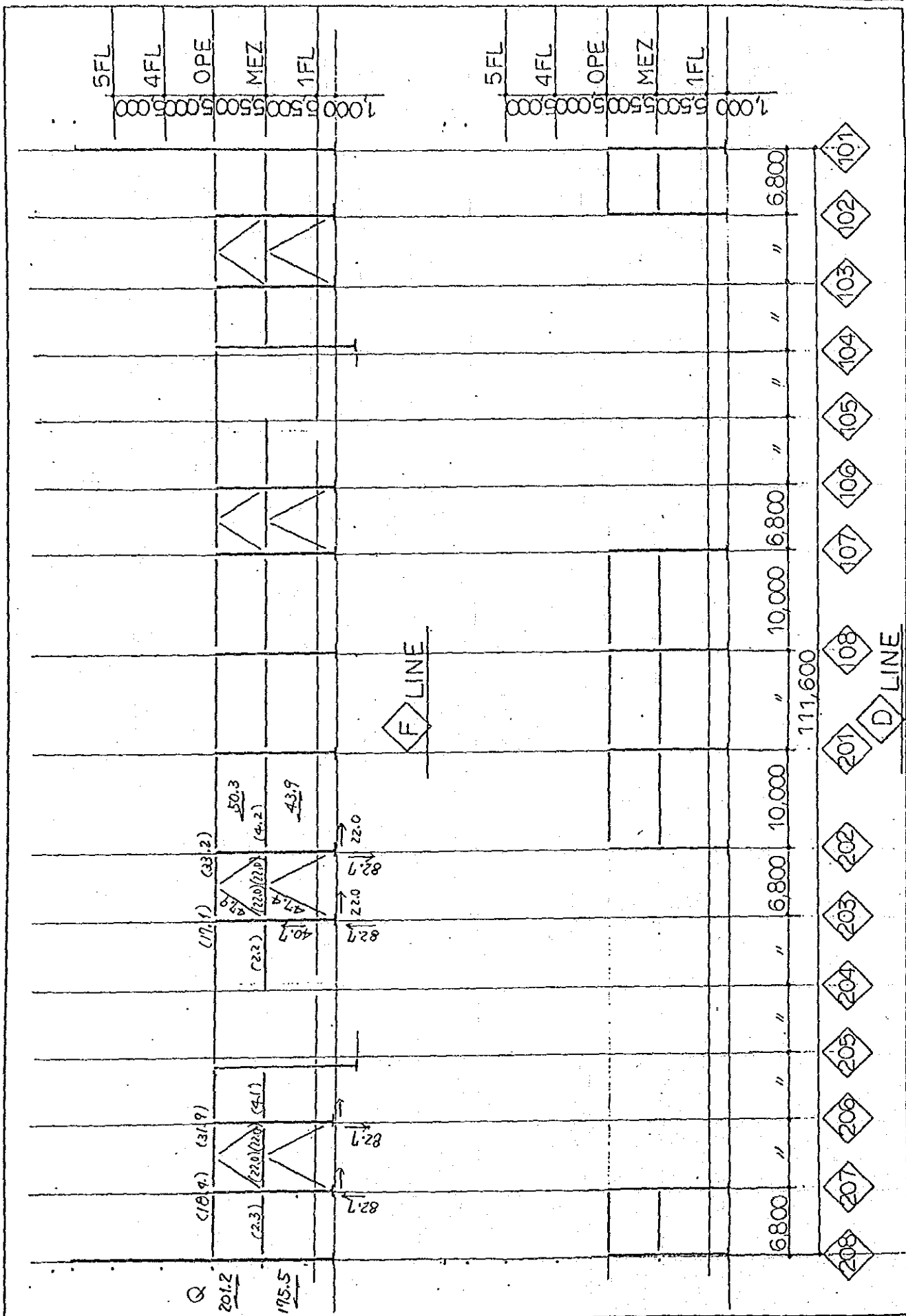
3.4 Stress of vertical bracing, Column and Girder



134

SFL
4FL
OPE
MEZ
1FL
1,000 5,500 5,500 1,000

5FL
4FL
OPE
MEZ
1FL
1,000 5,500 5,500 1,000



D LINE

Q
201.2
175.5

4. Design of Column
4.1 Design of Column

NOTE : VL --- VERTICAL LOAD
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CHL --- CRANE HORIZONTAL LOAD
PC --- PERMANENT CONDITIONS (VL+CLL)
TC --- TEMPORARY CONDITIONS (VL+CDL+CHL+SL)

TABLE FOR COLUMN STRESS (1)
[柱应力表]

LOCALIZATION	VL		CDL		CLL		SL		CHL		PC		TC		REMARKS							
	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t								
A-101 A-208	RF	36.7	-72.7	-7.5	0.2	-5.0	-2.8	0.7	-14.0	-8.1	0.0	0.0	0.3	0.3	32.4	-86.7	-15.6	41.1	-101.2	-10.0	X	
	OPF	70.5	-48.0	10.3	23.6	-0.1	0.0	62.7	-0.2	-0.1	-17.9	-36.0	-8.2	0.2	0.2	130.2	47.8	10.2	111.8	-90.9	20.0	Y
A-102 A-207	MEZ	96.8	-8.5	-1.3	23.0	0.1	0.0	66.0	0.2	0.0	-30.2	9.3	1.4	0.1	162.8	-8.3	-1.3	149.9	-91.2	-2.9		
	RF	53.8	-96.7	-9.7	0.2	-5.0	-2.8	0.7	-14.0	-8.1	13.2	-9.3	1.6	0.3	-6.1	54.5	-10.7	-17.8	58.0	-123.3	11.5	
A-103 A-206	OPF	106.4	59.3	12.7	23.6	-0.1	0.0	62.7	-0.2	-0.1	-15.4	-37.5	-8.5	0.2	174.1	59.1	72.6	145.2	103.7	22.7		
	MEZ	142.4	-10.8	-1.7	23.0	0.1	0.0	66.0	0.7	0.0	19.0	9.4	1.4	0.1	208.4	-10.6	-1.7	190.9	-21.6	3.3		
A-104 A-205	RF	53.8	-96.7	-9.7	0.2	-5.0	-2.8	0.7	-14.0	-8.1	-12.2	-4.7	1.6	0.3	-6.1	54.5	-10.6	-17.8	58.1	-123.7	11.5	
	OPF	106.4	59.2	12.7	25.0	-0.1	0.0	71.9	-0.1	0.0	-4.4	-37.7	-8.6	0.3	-7.0	178.3	59.1	12.7	155.5	103.8	22.8	
A-105 A-204	MEZ	142.4	-10.7	-1.6	24.0	0.0	0.0	68.8	0.1	0.0	-17.8	9.4	1.4	0.2	211.2	-10.6	-1.6	184.0	21.6	3.2		
	RF	53.8	-55.3	-2.0	0.1	-2.8	-2.4	0.3	-7.4	-6.9	8.0	-2.2	-0.0	-5.1	54.1	-62.7	-8.9	58.7	79.4	13.2		
A-105 A-204	5TH	61.8	-42.0	-2.0	24.9	-7.4	-0.3	71.5	-21.6	-1.0	-2.2	23.4	-1.1	0.7	133.3	-63.6	-3.0	88.2	82.8	6.2		
	MEZ	166.5	-12.8	-2.0	24.9	-2.3	-0.3	71.5	-6.6	-1.0	-2.2	24.4	3.8	0.7	178.0	-19.4	-3.0	132.9	42.5	6.6		
A-105 A-204	RF	53.8	-55.3	-2.0	0.1	-2.8	-2.4	0.3	-7.4	-6.9	0.0	-2.2	-9.0	0.7	54.1	-62.7	-8.9	52.4	79.6	13.2		
	5TH	61.8	-42.0	-2.0	24.9	-7.4	-0.3	71.5	-21.6	-1.0	-2.2	23.4	-1.1	0.7	133.3	-63.6	-3.0	88.2	82.8	6.2		
A-106 A-203	MEZ	107.5	-12.8	-2.0	24.9	-2.3	-0.3	71.5	-6.6	-1.0	-2.2	24.3	3.7	0.7	179.0	-19.4	-3.0	133.9	42.4	11.0		
	RF	53.8	-96.7	-9.7	0.2	-5.0	-2.8	0.7	-14.0	-8.1	12.2	-9.3	1.5	0.3	-6.1	54.5	-10.6	-17.8	58.0	-123.0	11.4	
A-106 A-203	OPF	107.4	54.2	12.7	25.0	-0.1	0.0	71.9	-0.1	0.0	-4.3	-37.5	-8.6	0.2	179.3	59.1	72.7	136.4	102.6	22.8		
	MEZ	150.2	-10.7	-1.7	24.0	0.0	0.0	68.8	0.1	0.0	-17.5	9.8	1.4	0.2	219.1	-10.6	-1.7	191.6	22.0	3.4		

NOTE : VL --- VERTICAL LOAD
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 PC --- PERMANENT CONDITIONS
 TC --- TEMPORARY CONDITIONS
 TABLE FOR COLUMN STRESS (2)
 (柱应力表)

LOCALIZATION	VL		CDL		CLL		SL		CHL		PC		TC		REMARKS					
	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t						
A-107 A-202	65.6	-11.7	-1.1	0.3	-3.2	0.8	-1.2	-9.2	-3.7	0.4	1.6	0.4	-7.2	-2.9	66.4	-127.9	-20.3	70.5	141.2	13.0
OPF	129.4	67.8	14.6	28.4	0.2	0.0	0.1	0.0	-3.7	0.4	-9.8	0.4	-7.9	-1.7	210.8	67.9	14.6	162.4	118.7	26.1
MEZ	173.5	-12.4	-1.9	28.4	0.0	0.0	0.1	0.0	-4.3	0.4	1.7	0.4	1.7	0.3	254.9	-12.5	-1.9	222.4	25.2	3.9
A-108 A-201	117.4	-127.0	-2.6	0.3	-5.7	0.8	-1.2	-9.2	-8.5	0.4	1.0	0.4	-7.2	-2.9	78.2	-143.2	-21.8	82.2	155.0	13.9
OPF	116.8	75.3	11.3	28.4	0.2	0.0	0.1	0.0	-6.7	0.4	-10.2	0.4	-7.9	-1.7	198.2	75.9	16.3	178.5	127.7	28.2
MEZ	133.1	-14.5	-2.2	28.4	0.0	0.0	0.1	0.0	-4.9	0.4	1.8	0.4	1.7	0.3	214.5	-14.6	-2.2	149.1	127.7	28.2
B-101 208	86.4	1.5	7.8	0.2	0.2	90.6	95.9
B-102 207	125.3	1.5	44.7	0.2	0.2	129.5	151.3
B-103 206	79.4	0.9	-10.7	0.2	0.2	81.9	114.0
B-104 205	35.1	0.0	0.9	0.0	0.0	35.1	35.1
B-105 204	36.1	0.0	0.0	0.0	0.0	36.1	36.1
B-106 203	96.3	0.9	40.7	0.2	0.2	98.8	131.0
B-107 202	143.6	0.0	-10.7	0.0	0.0	143.6	189.3
B-108 201	90.0	0.0	20.6	0.0	0.0	90.0	100.2
D-101 202	82.8	0.0	9.4	0.0	0.0	82.8	92.2
D-102 207	89.3	0.0	9.4	0.0	0.0	89.3	105.5
D-107 202	135.7	0.0	0.0	0.0	0.0	135.7	135.7
D-108 201	198.5	0.0	-21.7	0.0	0.0	198.5	220.2
	187.7	0.0	-21.7	0.0	0.0	187.7	209.4

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 PC --- PERMANENT CONDITIONS (VL+CLL)
 TC --- TEMPORARY CONDITIONS
 (VL+CDL+CHL+SL)

TABLE FOR COLUMN STRESS (3)
 [柱应力表]

LOCALIZATION	VL		CDL		CLL		SL		CHL		PC		TC		REMARKS
	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	
F-101 200	82.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	81.1	0.0	96.8	0.0	
F-102 207	128.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	128.4	0.0	169.1	0.0	
F-103 206	103.9	0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	104.5	0.0	145.1	0.0	
F-105 205	43.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	43.2	0.0	43.2	0.0	
F-106	75.4	0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	76.0	0.0	116.1	0.0	
F-107 202	160.6	0.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	161.4	0.0	201.3	0.0	
F-108	188.6	0.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	189.4	0.0	181.3	0.0	
F-201	199.7	0.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	180.5	0.0	193.2	0.0	
F-203	92.6	0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	93.2	0.0	133.3	0.0	
G-101 208	36.7	0.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	36.7	0.0	40.7	0.0	
G-102 209	59.9	0.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	72.0	0.0	66.5	0.0	
G-103 206	181.1	0.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	196.1	0.0	237.5	0.0	
G-104 207	53.0	0.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	53.0	0.0	57.6	0.0	
G-105 208	86.7	0.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	103.8	0.0	90.5	0.0	
G-106 209	300.9	0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	315.9	0.0	345.4	0.0	
G-107 206	53.8	0.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	53.8	0.0	57.7	0.0	
G-108 207	159.5	0.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	176.6	0.0	168.4	0.0	
G-109 208	373.7	0.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	386.3	0.0	431.6	0.0	

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 TC --- TEMPORARY CONDITIONS (VL+CDL+CHL+SL)

LOCATION	VL			CDL			CLL			SL			CHL			PC			TC			REMARKS	
	N, t	M, tmQ, t	t	N, t	M, tmQ, t	t	N, t	M, tmQ, t	t	N, t	M, tmQ, t	t	N, t	M, tmQ, t	t	N, t	M, tmQ, t	t	N, t	M, tmQ, t	t		
G-104 205	RF	53.8	83.1	9.9	-0.1	0.3	-0.3	0.3	0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
	5TH	152.5	19.4	5.2	13.2	7.3	2.1	17.5	9.3	2.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
	MER	365.0	2.8	0.4	10.8	-0.1	0.0	13.9	-0.2	0.0	-87.6	2.5	0.4	-1.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
G-105 204	RF	53.8	83.1	9.9	-0.1	0.3	-0.3	0.3	0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
	5TH	86.7	19.4	5.2	13.2	7.3	2.1	17.5	9.3	2.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
	MER	215.7	2.8	0.4	10.8	-0.1	0.0	13.9	-0.2	0.0	-58.8	2.5	0.4	-1.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
G-106 203	RF	53.8	71.2	7.3	-0.2	-0.4	0.2	-0.9	-1.4	0.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	5TH	79.9	22.7	6.2	13.0	7.4	2.2	17.1	9.6	2.9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	MER	273.2	2.7	0.4	9.9	-0.1	0.0	12.6	-0.1	0.0	-12.2	2.5	0.4	-0.6	-0.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
G-107 202	RF	65.0	89.9	9.3	-0.3	0.3	-0.8	-1.1	0.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	5TH	97.5	24.1	6.5	14.8	8.4	2.5	19.4	10.8	3.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	MER	381.5	3.3	0.5	12.5	-0.2	0.0	16.3	-0.2	0.0	-12.2	2.5	0.4	-0.3	-0.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
G-108 201	RF	77.4	112.6	15.3	-0.3	0.3	-0.8	-1.1	0.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	5TH	115.2	16.5	4.2	14.8	8.9	2.5	19.4	10.8	3.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
	MER	485.2	3.9	0.5	12.5	-0.2	0.0	16.3	-0.2	0.0	-16.9	2.9	0.4	-0.3	-0.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
H-101 208	RF	42.8	0.0
	5TH	79.5	1.7
	MER	87.3	0.0
H-102 209	RF	207.3	1.2
	5TH
	MER

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 TC --- TEMPORARY CONDITIONS
 TABLE FOR COLUMN STRESS (5)
 (柱底力表)

LOCATION	DIRLEV.	VL		CDL		CLL		SL		CHL		PC		TC		REMARKS
		N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	
H-103 206	4TH	140.9	1.3	2.0	38.0	0.3	162.9	207.2	386.6	378.8	207.2	386.6	378.8			
	ME8	280.9	2.3	3.5	76.0	0.6	384.4	766.6	757.6							
H-104 205	4TH	173.4	1.3	2.0	0.0	0.4	175.4	217.7	42.6							
	ME8	305.4	2.4	3.6	0.0	0.6	309.0	398.2	84.0							
H-105 209	4TH	99.8	1.3	2.0	38.0	0.4	101.8	127.8	0.0							
	ME8	237.8	2.4	3.6	76.0	0.6	235.4	337.5	0.0							
H-106	4TH	146.1	1.3	2.0	38.0	0.3	148.1	184.1	27.5							
	ME8	338.4	2.3	3.5	76.0	0.6	341.9	406.2	105.7							
H-107 202	4TH	202.8	0.0	0.0	47.6	0.0	202.8	250.4	47.6							
	ME8	571.0	1.0	1.4	137.1	0.0	522.4	638.1	73.6							
H-108 201	4TH	239.4	0.0	0.0	47.6	0.0	239.4	273.9	47.6							
	ME8	593.1	1.0	1.4	137.1	0.0	548.5	638.1	73.6							
H-203	4TH	176.5	1.3	2.0	38.0	0.3	178.5	214.5	27.5							
	ME8	371.8	2.3	3.5	76.0	0.6	375.3	477.5	105.7							
K-106	4TH	60.1	0.0	0.0	0.0	0.0	60.1	87.6	0.0							
	ME8	147.8	0.2	0.4	19.6	0.3	151.8	221.8	27.5							
K-107	4TH	133.7	0.0	0.0	0.0	0.0	133.7	405.7	0.0							
	ME8	288.1	0.2	0.4	19.6	0.3	288.5	405.7	27.5							
K-108	4TH	121.1	0.0	0.0	0.0	0.0	121.1	155.6	0.0							
	ME8	303.6	0.2	0.4	19.6	0.3	304.0	405.7	27.5							

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TC --- TEMPORARY CONDITIONS
(VL+CDL+CHL+SL)

TABLE FOR COLUMN STRESS (ϕ)
(柱应力表)

LOCATION TION	VL		CDL		CLL		SL		CHL		PC		TC		REMARKS
	N,t	M,tmQ,t	N,t	M,tmQ,t	N,t	M,tmQ,t	N,t	M,tmQ,t	N,t	M,tmQ,t	N,t	M,tmQ,t	N,t	M,tmQ,t	
K-201	4TH	113.7	34.5
			0.0	0.0	0.0	198.2
K-202	MEZ	287.8	86.0
			0.2	0.4	38.0	290.2
K-203	4TH	105.2	39.5
			0.0	0.0	0.0	105.2
	MEZ	248.7	86.0
			0.2	0.4	0.0	249.1
	4TH	55.3	0.0
			0.0	0.0	0.0	55.3
	MEZ	178.9	73.5
			0.2	0.4	73.5	179.3

NOTE: $\sigma b/fb < 1.0$
 $\sigma c/fc < 1.0$
 $\sigma b/fb + \sigma c/fc < 1.0$

DECISION OF COLUMN (I)
 [柱の断面算定]

LOCATION NO.	DIV NO.	DIR X Y	PERMANENT CONDITIONS			TEMPORARY CONDITIONS			MEMBER (A, I, Z)		A. STRESS			$\sigma b/fb$	$\sigma c/fc$	$\sigma b/fb + \sigma c/fc$
			N	M	Q	N	M	Q	i, η	$\lambda b, \lambda c$	fb, fc	$\sigma b/fb$	$\sigma c/fc$			
A-101 106	3		54.5	110.6	17.8	58.0	123.0	11.4	BH-700x350x28x40	9.4	46	1.6	1.07	0.12	0.76	
			201.5			201.5			BH-900x350x28x40	7.95	54	1.35	0.67	0.09		
			179.3	59.1	12.7	136.4	103.6	22.8	509.6	14,372	7.5	73	1.17	0.26		0.30
A-107 202	2		219.1	10.6	1.7	191.6	27.0	3.4	Ditto	9.15	71	1.6	0.07	0.43	0.46	
			280.3			280.3				7.5	87	1.02	0.04	0.42		
			78.2	143.2	21.8	82.2	155.0	13.9	BH-700x400x28x40	10.8	40	1.6	1.24	0.16		
G-101 106	3		210.8	67.9	14.6	162.4	118.7	26.1	BH-900x900x28x40	9.3	46	1.41	0.98	0.11	0.89	
			254.9	12.5	1.9	222.4	25.2	3.9	Ditto	10.5	52	1.6	0.42	0.38		
			10.8	40	1.6	1.24	0.16	10.5	52	1.6	0.42	0.38				
G-107 202	2		286.3	2.7	0.4	431.6	0.4	0.6	Ditto	10.5	62	1.6	0.08	0.46	0.95	
			431.6			431.6				8.8	74	1.16	0.05	0.40		
			53.5	83.4	10.4	55.2	138.6	29.8	BH-700x350x28x40	9.4	46	1.6	0.81	0.12		
G-107 202	2		179.6	28.7	8.0	171.2	81.3	22.3	BH-900x350x28x40	7.95	54	1.35	0.57	0.09	0.60	
			280.3			280.3				9.15	71	1.6	0.07	0.43		
			179.6	28.7	8.0	171.2	81.3	22.3	509.6	14,372	7.5	67	1.23	0.13		0.28
G-107 202	1		481.5	3.2	0.7	486.3	6.0	0.9	Ditto	9.15	71	1.6	0.02	0.76	0.76	
			114.8			114.8				7.5	87	1.02	0.01	0.75		
			77.4	112.6	15.3	81.6	189.7	40.9	BH-700x900x28x40	10.8	40	1.6	0.70	0.16		
B-101 103 106 108 201 202 208	1		143.6	0.0	0.0				H-300 ²						0.82	
			114.8			114.8				7.51	87	1.02	0.80	0.80		
			98.8	0.0	0.0											
B-102 107 202 207	1		193.9						H-350 ²						0.83	
			114.8			114.8				8.84	74	1.16	0.72	0.72		
			143.6	0.0	0.0											
D-101 208	1		89.3	0.0	0.0				H-300 ²						0.75	
			114.8			114.8				7.51	87	1.02	0.74	0.74		
			89.3	0.0	0.0											

NOTATION: DIV NO. --- DIVISION NUMBER OF COLUMN
 DIR --- DIRECTION, X OR Y
 N, M, Q --- AXIAL FORCE(t), BENDING MOMENT(tm), SHEAR FORCE(t)
 A, I, Z --- SECTION AREA (cm²), GEOMETRY MOMENT OF INERTIA (cm⁴), SECTION MODULUS (cm³)
 A. STRESS --- ALLOWABLE STRESS
 i, η --- CONSTANTS FOR DECISION OF ALLOWABLE BENDING STRESS
 i_{min} --- MINIMUM RADIUS OF SECTION (cm)
 $\lambda b, \lambda c$ --- BUCKLING COEFFICIENT FOR BENDING MOMENT AND AXIAL FORCE
 fb, fc --- ALLOWABLE STRESS FOR BENDING AND AXIAL FORCE (t/cm²)

NOTE: $\sigma b/fb < 1.0$
 $\sigma c/fc < 1.0$
 $\sigma b/fb + \sigma c/fc < 1.0$

DECISION OF COLUMN (2)
 [柱の断面算定]

LOCATION NO.	DIR X Y	PERMANENT CONDITIONS			TEMPORARY CONDITIONS			MEMBER (A, I, Z)	A. STRESS			σb σb fb	σc σc fc	$\sigma b/fb$ + $\sigma c/fc$	
		N	M	Q	N	M	Q		i, η imin	λb λc	f_b f_c				
D-102 207															
	1	198.5	0.0	0.0			H-350 ² 173.9	8.84	74	1.16		1.14 0.98	0.98		
F-101 103 106 203 206 208															
	1	104.5	0.0	0.0			H-300 ² 119.8	7.51	87	1.02		0.87 0.85	0.85		
F-102 107 202 207															
	1	189.4	0.0	0.0			H-350 ² 173.9	8.24	74	1.16		1.09 0.94	0.94		
H-101 208															
	2	42.8			70.3		H-300 ² 119.8	7.51	67	1.23		0.36 0.29	0.29		
	1	81.2			148.9		Diff ₀	7.51	87	1.02		1.24 0.81	0.81		
H-102 209															
	2	87.3					H-390 136.0	7.28	69	1.21		0.64 0.53	0.53		
	1	209.0					H-400 ² 210.7	10.1	64	1.26		0.96 0.76	0.76		
H-103 105 106 209 206															
	2	162.9					H-440 157.4	7.18	70	1.20		1.03 0.86	0.86		
	1	341.9					BH-450 ² 289.0	11.46	57	1.32		1.18 0.89	0.89		
H-107 108 201 202 203 205															
	2	239.4					BH-480x380x16x2L 223.76	8.38	60	1.30		1.07 0.82	0.82		
	1	594.5					BH-516x300 495.92	12.96	50	1.38		1.20 0.87	0.87		

NOTATION: DIV NO. --- DIVISION NUMBER OF COLUMN

DIR --- DIRECTION, X OR Y

N, M, Q --- AXIAL FORCE (t), BENDING MOMENT (tm), SHEAR FORCE (t)

A, I, Z --- SECTION AREA (cm²), GEOMETRY MOMENT OF INERTIA (cm⁴),
SECTION MODULUS (cm³)

A. STRESS --- ALLOWABLE STRESS

i, η --- CONSTANTS FOR DECISION OF ALLOWABLE BENDING STRESS

imin --- MINIMUM RADIUS OF SECTION (cm)

$\lambda b, \lambda c$ --- BUCKLING COEFFICIENT FOR BENDING MOMENT AND AXIAL FORCE

f_b, f_c --- ALLOWABLE STRESS FOR BENDING AND AXIAL FORCE (t/cm²)

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NOTE: $\sigma b/fb < 1.0$
 $\sigma c/fc < 1.0$
 $\sigma b/fb + \sigma c/fc < 1.0$

DECISION OF COLUMN (3)
 [柱の断面算定]

LOCATION NO.	DIR X Y	PERMANENT CONDITIONS			TEMPORARY CONDITIONS			MEMBER (A, I, Z)	A. STRESS			$\sigma b / fb$	$\sigma c / fc$	$\sigma b / fb + \sigma c / fc$	
		N	M	Q	N	M	Q		i, η	λb	fb				i, η
K-106 203															
	2		60.1					H-350 ² 193.9	8.84	57	1.32		0.35	0.27	0.27
	1		179.3					Ditto	8.84	74	1.16		1.03	0.89	0.89
K-107 202															
	2		133.7					H-440 157.4	7.18	70	1.20		0.85	0.71	0.71
	1		309.0					H-450 ² 289.0	11.46	57	1.32		1.05	0.80	0.80

NOTATION: DIV NO. --- DIVISION NUMBER OF COLUMN
 DIR --- DIRECTION, X OR Y
 N, M, Q --- AXIAL FORCE(t), BENDING MOMENT(tm), SHEAR FORCE(t)
 A, I, Z --- SECTION AREA (cm²), GEOMETRY MOMENT OF INERTIA (cm⁴), SECTION MODULUS (cm³)
 A. STRESS --- ALLOWABLE STRESS
 i, η --- CONSTANTS FOR DECISION OF ALLOWABLE BENDING STRESS
 i, η --- MINIMUM RADIUS OF SECTION (cm)
 $\lambda b, \lambda c$ --- BUCKLING COEFFICIENT FOR BENDING MOMENT AND AXIAL FORCE
 fb, fc --- ALLOWABLE STRESS FOR BENDING AND AXIAL FORCE (t/cm²)

4.2 Design of Column Base

NOTE : VL --- VERTICAL LOAD
 CHL --- CRANE HORIZONTAL LOAD (VL+CLL)
 CDL --- CRANE DEAD LOAD
 PC --- PERMANENT CONDITIONS (VL+CLL)
 CLL --- CRANE LIFTING LOAD
 TC --- TEMPORARY CONDITIONS
 SL --- SEISMIC LOAD
 (VL+CDL+CHL+SL)

TABLE FOR COLUMN BASE ()
 (柱底力表)

LOCALIZATION	DIRLEV.	VL		CDL		CLL		SL		CHL		PC		TC		REMARKS	
		N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t		
A-101	X																
208	Y	96.8		23.3	-0.2	62.0	-0.7	-3.0	1.4	0.1		0.3	183.8	-0.7	150	1.9	
102	X								20.6						215.3	20.6	
207	Y	142.4		23.3	-0.2	67.0	-0.7	-48.5	17.3	0.1		0.3	209.4	-0.7	-7.2	20.6	
103	X								20.6						216.1	20.6	
206	Y	142.4		24.1	-0.1	69.4	-0.4	-37.9	15.4	0.1		0.3	211.8	-0.4	-7.2	20.6	
104	X								3.8	0.7							
205	Y	106.5		27.9	-0.3	71.5	-1.0	-2.3				0.5	178.0	-1.0	133.0	3.6	
105	X																
204	Y	107.5		24.9	-0.3	71.5	-1.0	-2.2	3.7	0.7		0.5	179.0	-1.0	133.9	3.5	
106	X								20.6						387.9	20.6	
203	Y	150.3		24.1	-0.1	69.4	-0.4	-37.9	15.7	0.1		0.3	219.7	-0.4	0.7	20.6	
107	X								42.9						259.7	42.9	
202	Y	173.5		28.3	0.0	81.2	0.1	-45.3	18.6	0.3		0.4	254.7	0.1	124.6	42.9	
108	X								21.8						262.1	21.8	
201	Y	133.1		28.3	0.0	81.2	0.1	-32.8	21.1	0.3		0.4	244.3	0.1	32.4	21.8	
B-101																	
208		86.4		1.4	0.0	4.1	0.1	-11.2	17.5	0.1		0.1	90.5	0.1	98.9	17.4	
102									22.0						11.9	209.4	
207		125.3		1.4	0.0	4.1	0.1	7.3	15.7	0.1		0.1	129.4	0.1	42.6	22.9	
103									22.9						-10.1	X Direction	
206		79.4		0.9	0.0	2.5	0.0	-22.7	32.9	0.2				-11.9	163.0	23.9	
104									0.0						0.0	-3.3	10.1
205		35.1															
105													35.1				
204		36.1															
106									22.9						11.9	179.9	23.9
203		96.3		0.9	0.0	2.5	0.0	33.6	0.0	0.2		0.0	98.8	0.0	13.6	-10.1	
107									22.9						-11.9	226.3	23.9
202		143.6		0.0	0.0	-0.1	0.1	3.4	15.8	-0.1		0.1	143.5	0.1	60.9	10.1	
108									17.7								
201		90.8		0.0	0.0	-0.1	0.1	-9.7		-0.1		0.1	90.7	0.1	100.6	17.8	
									17.7								
		82.8		0.0	0.0	-0.1	0.1	-9.7		-0.1		0.1	82.7	0.1	92.6	17.8	

NOTE : VI --- VERTICAL LOAD
 CDL --- CRANE DEAD LOAD
 CLL --- CRANE LIFTING LOAD
 SL --- SEISMIC LOAD
 CHL --- CRANE HORIZONTAL LOAD
 PC --- PERMANENT CONDITIONS
 TC --- TEMPORARY CONDITIONS
 (VL+CDL+CHL+SL)

LOCATION	VI		CDL		CLL		SL		CHL		PC		TC		REMARKS
	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	N, t	M, tmQ, t	
G-101	181.1	-9.7	11.6	0.0	15.1	0.1	-81.2	60.3	-0.5	0.1	196.2	-9.6	70.1	70.1	Y Direction
202														60.3	
102															Y Direction
207														37.0	
103															Y Direction
206														15.1	
104															Y Direction
205														32.0	
105															Y Direction
204														11.1	
106															Y Direction
203														0.4	
107															Y Direction
202														36.9	
108															Y Direction
201														0.4	
H-101															Y Direction
208														53.9	
102															Y Direction
207														43.7	
103															Y Direction
206														20.6	
104															Y Direction
205														20.6	
105															Y Direction
204														20.6	
106															Y Direction
203														20.6	
107															Y Direction
202														20.6	
108															Y Direction
201														20.6	
203															Y Direction
														20.6	

DECISION OF COLUMN BASE (1)
 [柱脚の断面算定]

LOCATION		A-105 (101, 104, 105, 204, 205, 208)	A-103 (102, 106, 203, 206, 207)
COLUMN SIZE		BH-900x350 x 28x40	BH-900x350 x 28x40
DIRECTION		X	Y
PERMANENT CONDITIONS	M (tm)		0.5
	N (t)		179.0
	Q (t)		1.0
TEMPORARY CONDITIONS	M (tm)		1.9
	N (t)		133.9
	Q (t)		3.5
FIGURE			same as A-105
BASE PLATE (LxBxt)		1300x700 x 29	1300x700 x 29
CAP PLATE (THICK.)		22	22
RIB PLATE (HxBxt)		16	16
WING PLATE (HxBxt)		22	22
ANCHOR BOLT (n-Dφ)		8-30φ	8-30φ
SHEAR KEY (HxBxt)			
CONC.	e=M/N (cm)	$\frac{0.5 \times 10^2}{179} = 0.28 < \frac{130}{6} = 21.7$	
	$\sigma_c < f_c$ (kg/cm ²)	$\frac{179.0 \times 10^3}{65 \times 130} \left(1 + \frac{6 \times 0.28}{130}\right) = 21.5 < 70$	$\frac{211.8 \times 10^3}{65 \times 130} = 25.1 < 70$
ANCHOR BOLT	P/(n*A) < ft (t/cm ²)		$N/bdf = -0.0005$ $f_t = 0.13\%$
BASE PLATE	M=α wlx ² (tcm)	$0.09 \times 0.022 \times 32.5^2 = 2.1$ tcm	$M/bd^2 = 0.001$ $a_t = 110 \times 65 \times R/2 = 5.5 \text{ cm}^2$
	t > √(6XM/fb) (cm)	$\sqrt{6 \times 2.1 / 1.85} = 2.61 \rightarrow 2.9$	$0.09 \times 0.025 \times 32.5^2 = 2.4$ tcm $\sqrt{6 \times 2.4 / 1.85} = 2.79 \rightarrow 2.9$
CAP PLATE	M=α wlx ² (tcm)		$0.085 \times (2 \times 1.8 \times 7.06 / (20 \times 14)) \times 14^2 = 1.51$
	t > √(6XM/fb) (cm)		$\sqrt{6 \times 1.51 / 2.79} = 1.81 \rightarrow 2.2$
RIB PLATE	$\tau = \sigma cA / (txH)$ (t/cm ²)	$\frac{0.022 \times 32.5^2 \times 35}{1.6 \times 50} = 0.81 < 0.9$	$\frac{1.90 \times 7.06}{1.6 \times 50} = 0.15 < 1.35$
	H/t	$\frac{50}{1.6} = 31.3 < 71$	
	WELDING $\tau < f_s$ (t/cm ²)		
WING PLATE	$\tau = \sigma cA / (txH)$ (t/cm ²)	$\frac{0.022 \times 37.5^2 \times 15.0}{1.6 \times 50} = 0.15 < 0.9$	
	H/t	$\frac{50}{1.6} = 31.3 < 71$	
	WELDING $\tau < f_s$ (t/cm ²)		
ALLOWABLE STRESS	CONC. : fc (t/cm ²)		
	A. BOLT : ft (t/cm ²)		
	PLATE : fb (t/cm ²)		
	PLATE : fs (t/cm ²)		
REMARKS			$(130 \times 135 - 130 \times 15) \times 7 \times 1.5 / 1000 = 95.2 \tau$
NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm) e --- ECCENTRICITY σ c --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e P --- UP-LIFT FORCE FOR COLUMN BASE (t) α --- COEFFICIENT FOR BENDING MOMENT OF SLAB			

DECISION OF COLUMN BASE (2)
[柱脚の断面算定]

LOCATION		A-107.202	A-108.201
COLUMN SIZE		BH-900x400x28x40	
DIRECTION		X	Y
PERMANENT CONDITIONS	M (tm)	0.1	0.1
	N (t)	254.7	214.3
	Q (t)	0.1	0.1
TEMPORARY CONDITIONS	M (tm)	21.2	10.9
	N (t)	124.6	32.4
	Q (t)	42.4	21.8
FIGURE		Same as G-108	
BASE PLATE (LxBxt)		1300x700x35	
CAP PLATE (THICK.)		22	
RIB PLATE (HxBxt)		16	
WING PLATE (HxBxt)		22	
ANCHOR BOLT (n-Dφ)		8-35φ	
SHEAR KEY (HxBxt)			
CONC.	e=M/N (cm)		
	$\sigma < c < f_c$ (kg/cm ²)	$\frac{254.7 \times 10^3}{65 \times 130} = 30.1 < 70$	
ANCHOR BOLT	$P/(n \cdot A) < f_t$ (t/cm ²)	$N/bd = 0.0008$ } $P_t = 0.1916$ $M/bd^2 = 0.002$ } $A_t = 130 \times 65 \times 16 / 2 = 8 \text{ cm}^2$	
BASE PLATE	$M = \alpha w l x^2$ (tcm)	$0.09 \times 0.03 \times 32.5^2 = 2.85$	
	$t > \sqrt{6xM/f_b}$ (cm)	$\sqrt{6 \times 2.85 / 1.85} = 3.1 \rightarrow 3.5$	
CAP PLATE	$M = \alpha w l x^2$ (tcm)	$0.085 \times (2 \times 1.8 \times 8.0 / 20 \times 14) \times 14^2 = 1.71$	
	$t > \sqrt{6xM/f_b}$ (cm)	$\sqrt{6 \times 1.71 / 2.77} = 1.92 \rightarrow 2.2$	
RIB PLATE	$\tau = \sigma CA / (txH)$ (t/cm ²)	$\frac{0.03 \times 32.5 \times 35}{1.2 \times 50} = 0.57 < 0.9$	
	H/t		
WING PLATE	WELDING $\tau < f_s$ (t/cm ²)		
	$\tau = \sigma CA / (txH)$ (t/cm ²)	$\frac{0.03 \times 37.5 \times 12.5}{1.6 \times 50} = 0.18 < 0.9$	
ALLOWABLE STRESS	H/t		
	WELDING $\tau < f_s$ (t/cm ²)		
ALLOWABLE STRESS	CONC. : f_c (t/cm ²)		
	A. BOLT : f_t (t/cm ²)		
	PLATE : f_b (t/cm ²)		
	PLATE : f_s (t/cm ²)		
REMARKS			
NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm) e --- ECCENTRICITY σc --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e P --- UP-LIFT FORCE FOR COLUMN BASE (t) α --- COEFFICIENT FOR BENDING MOMENT OF SLAB			

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DECISION OF COLUMN BASE (3)
 [柱脚の断面算定]

LOCATION		B-108 (101, 201, 208)	B-107 (102, 202, 207)
COLUMN SIZE		H-300 ² x 10 x 15	H-350 ² x 12 x 19
DIRECTION		X	Y
PERMANENT CONDITIONS	M (tm)		3.6
	N (t)	90.7	143.5
	Q (t)	0.1	11.9
TEMPORARY CONDITIONS	M (tm)	5.4	3.0
	N (t)	100.6	60.9
	Q (t)	17.8	10.1
FIGURE			
BASE PLATE (LxBxt)		700 x 350 x 22	750 x 400 x 25
CAP PLATE (THICK.)		—	—
RIB PLATE (HxBxt)		12	16
WING PLATE (HxBxt)		—	—
ANCHOR BOLT (n-Dφ)		4-25φ	4-25φ
SHEAR KEY (HxBxt)		—	—
CONC.	e=M/N (cm)		
	$\sigma_c < f_c$ (kg/cm ²)	$\frac{90.7 \times 10^3}{70 \times 35} = 37.0 < 70$	
ANCHOR BOLT	P/(n*A) < ft (t/cm ²)	$\frac{N}{bDf} = 0.027$ } $P_t = 0.0$ $\frac{M}{bDf^2} = 0.0024$	
	M=α wlx ² (tcm)	$0.085 \times 0.037 \times 7.5^2 = 0.49$	
BASE PLATE	t > √(6XM/Fb) (cm)	$\sqrt{6 \times 0.49 / 1.85} = 1.26 \rightarrow 2.2$	
	M=α wlx ² (tcm)		
CAP PLATE	t > √(6XM/fb) (cm)		
	τ = σ cA/(txH) (t/cm ²)	$\frac{0.037 \times 12.5 \times 20}{1.2 \times 30} = 0.26 < 0.9$	
RIB PLATE	H/t		
	WELDING τ < fs (t/cm ²)		
WING PLATE	τ = σ cA/(txH) (t/cm ²)		
	H/t		
ALLOWABLE STRESS	CONC. : fc (t/cm ²)		
	A. BOLT : ft (t/cm ²)		
REMARKS	PLATE : fb (t/cm ²)		
	PLATE : fs (t/cm ²)		
NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm) e --- ECCENTRICITY σ c --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e P --- UP-LIFT FORCE FOR COLUMN BASE (t) α --- COEFFICIENT FOR BENDING MOMENT OF SLAB			

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DECISION OF COLUMN BASE (4)
 [柱脚の断面算定]

LOCATION		B-103 (106, 203, 206)	B-105 (104, 204, 205)
COLUMN SIZE		H-300 ² x 10 x 15	
DIRECTION		X	Y
PERMANENT CONDITIONS	M (tm)	3.6	0.0
	N (t)	98.8	36.1
	Q (t)	11.9	0.0
TEMPORARY CONDITIONS	M (tm)	3.0	
	N (t)	-3.3	
	Q (t)	10.1	
FIGURE			
BASE PLATE (LxBxt)		700 x 350 x 22	700 x 350 x 22
CAP PLATE (THICK.)		16	12
RIB PLATE (HxBxt)		12	12
WING PLATE (HxBxt)			
ANCHOR BOLT (n-Dφ)		4-25φ	4-25φ
SHEAR KEY (HxBxt)			
CONC.	e=M/N (cm)	$\frac{3.6 \times 10^2}{98.8} = 3.6 < \frac{35}{6}$	
	$\sigma_c < f_c$ (kg/cm ²)	$\frac{98.8 \times 10^3}{70 \times 35} \times (1 + \frac{6 \times 3.6}{35}) = 65.2 < 70$	
ANCHOR BOLT	P/(n*A) < ft (t/cm ²)	N/bdf = -0.0007, ft = 0.316 M/bd ² f = 0.0019, ft = 70 x 35 ft / 2 = 3.7 cm ²	
BASE PLATE	M = α wlx ² (tcm)	0.085 x 0.065 x 15 ² = 1.24	
	t > √(6XM/ft) (cm)	√(6 x 1.24 / 1.85) = 2.0 → 2.2	
CAP PLATE	M = α wlx ² (tcm)	0.085 x (2 x 3.7 x 1.5 / (12.5 x 20)) x 12.5 ² = 0.71	
	t > √(6XM/ft) (cm)	√(6 x 0.71 / 2.97) = 1.24 → 1.6	
RIB PLATE	τ = σ cA/(txH) (t/cm ²)	$\frac{0.065 \times 17.5 \times 15}{1.2 \times 30} = 0.47 < 0.9$	
	H/t		
WING PLATE	WELDING τ < fs (t/cm ²)		
	τ = σ cA/(txH) (t/cm ²)		
WING PLATE	H/t		
	WELDING τ < fs (t/cm ²)		
ALLOWABLE STRESS	CONC. : fc (t/cm ²)		
	A. BOLT : ft (t/cm ²)		
	PLATE : fb (t/cm ²)		
	PLATE : fs (t/cm ²)		
REMARKS			
NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm) e --- ECCENTRICITY σ c --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e P --- UP-LIFT FORCE FOR COLUMN BASE (t) α --- COEFFICIENT FOR BENDING MOMENT OF SLAB			

DECISION OF COLUMN BASE (5)
 [柱脚の断面算定]

LOCATION		D-101, 208	D-108(102, 107, 201, 202, 207)
COLUMN SIZE		H-300 ² X10 X15	H-350 ² X12 X19
DIRECTION		X	Y
PERMANENT CONDITIONS	M (tm)		0.2
	N (t)		89.8
	Q (t)		0.7
TEMPORARY CONDITIONS	M (tm)		9.1
	N (t)		76.0
	Q (t)		30.4
FIGURE		Same as B-108	Same as B-107
BASE PLATE (LxBxt)		700 X 350 X 22	750 X 400 X 25
CAP PLATE (THICK.)			
RIB PLATE (HxBxt)		12	16
WING PLATE (HxBxt)			
ANCHOR BOLT (n-Dφ)		4-25φ	4-25φ
SHEAR KEY (HxBxt)			
CONC.	e=M/N (cm)		
	$\sigma_c < f_c$ (kg/cm ²)		$\frac{198.5 \times 10^3}{75 \times 40} = 66.2 < 70$
ANCHOR BOLT	$P/(n \cdot A) < f_t$ (t/cm ²)		
BASE PLATE	$M = \alpha w l x^2$ (tcm)		$0.085 \times 0.066 \times 17.5^2 = 1.72$
	$t > \sqrt{6XM/f_b}$ (cm)		$\sqrt{6 \times 1.72 / 1.85} = 2.36 \rightarrow 2.5$
CAP PLATE	$M = \alpha w l x^2$ (tcm)		
	$t > \sqrt{6XM/f_b}$ (cm)		
RIB PLATE	$\tau = \sigma c A / (t \cdot x H)$ (t/cm ²)		$\frac{0.066 \times 17.5 \times 20}{1.6 \times 30} = 0.48 < 0.9$
	H/t		
WING PLATE	WELDING $\tau < f_s$ (t/cm ²)		
	$\tau = \sigma c A / (t \cdot x H)$ (t/cm ²)		
ALLOWABLE STRESS	H/t		
	WELDING $\tau < f_s$ (t/cm ²)		
REMARKS	CONC. : f_c (t/cm ²)		
	A. BOLT : f_t (t/cm ²)		
	PLATE : f_b (t/cm ²)		
	PLATE : f_s (t/cm ²)		
NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm)			
e --- ECCENTRICITY			
σ_c --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e			
P --- UP-LIFT FORCE FOR COLUMN BASE (t)			
α --- COEFFICIENT FOR BENDING MOMENT OF SLAB			

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DECISION OF COLUMN BASE (6)
 [柱脚の断面算定]

LOCATION		F-103 (101, 206, 208)	F-108 (102, 107, 201, 202, 207)
COLUMN SIZE		H-300 ² x 10 x 15	H-350 ² x 12 x 19
DIRECTION		X	Y
PERMANENT CONDITIONS	M (tm)		3.6
	N (t)		109.8
	Q (t)		11.9
TEMPORARY CONDITIONS	M (tm)	3.0	8.9
	N (t)	21.2	69.5
	Q (t)	10.1	29.8
FIGURE		same as B-108	same as B-107
BASE PLATE (LxBxt)		700x350x22	750x400x25
CAP PLATE (THICK.)			
RIB PLATE (HxBxt)		12	16
WING PLATE (HxBxt)			
ANCHOR BOLT (n-Dφ)		4-25φ	4-25φ
SHEAR KEY (HxBxt)			
CONC.	e=M/N (cm)	$\frac{3.6 \times 10^2}{109.8} = 3.4 < \frac{70}{6}$	
	$\sigma c < f_c$ (kg/cm ²)	$\frac{104.8 \times 10^3}{70 \times 35} \times \left(1 + \frac{6 \times 3.4}{70}\right) = 55.2 < 70$	
ANCHOR BOLT	P/(n*A) < ft (t/cm ²)		
BASE PLATE	M = α wlx ² (tcm)	0.085 × 0.055 × 12.5 ² = 0.73	
CAP PLATE	t > √(6XM/fb) (cm)	√(6 × 0.73 / 1.85) = 1.5 → 22	
WING PLATE	M = α wlx ² (tcm)		
RIB PLATE	t > √(6XM/fb) (cm)		
WELDING	τ = σ cA/(txH) (t/cm ²)		
	H/t		
	τ < fs (t/cm ²)		
WELDING	τ = σ cA/(txH) (t/cm ²)		
	H/t		
	τ < fs (t/cm ²)		
ALLOWABLE STRESS	CONC. : fc (t/cm ²)		
	A. BOLT : ft (t/cm ²)		
	PLATE : fb (t/cm ²)		
	PLATE : fs (t/cm ²)		
REMARKS			
NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm) e --- ECCENTRICITY σ c --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e P --- UP-LIFT FORCE FOR COLUMN BASE (t) α --- COEFFICIENT FOR BENDING MOMENT OF SLAB			

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DECISION OF COLUMN BASE (7)

[柱脚の断面算定]

LOCATION		F-104 (105, 204, 205)	F-203, 106
COLUMN SIZE		H-300 ² x 10 x 15	H-300 ² x 10 x 15
DIRECTION		X	Y
PERMANENT CONDITIONS	M (tm)		3.6
	N (t)	73.2	26.3
	Q (t)	0.0	11.9
TEMPORARY CONDITIONS	M (tm)		6.6
	N (t)		9.9
	Q (t)		22.0
FIGURE		Same as B-108	Same as B-108
BASE PLATE (LxBxt)		700 x 350 x 22	700 x 350 x 22
CAP PLATE (THICK.)			
RIB PLATE (HxBxt)		12	12
WING PLATE (HxBxt)			
ANCHOR BOLT (n-Dφ)		4-25φ	4-25φ
SHEAR KEY (HxBxt)			
CONC.	e=M/N (cm)		
	σ c < fc (kg/cm ²)		
ANCHOR BOLT	P/(n*A) < ft (t/cm ²)		
BASE PLATE	M=α wlx ² (tcm) t > √(6XM/FB) (cm)		
CAP PLATE	M=α wlx ² (tcm) t > √(6XM/fb) (cm)		
RIB PLATE	τ = σ cA/(txH) (t/cm ²)		
	H/t		
WING PLATE	WELDING τ < fs (t/cm ²)		
	τ = σ cA/(txH) (t/cm ²)		
ALLOWABLE STRESS	H/t		
	WELDING τ < fs (t/cm ²)		
ALLOWABLE STRESS	CONC. : fc (t/cm ²)		
	A. BOLT : ft (t/cm ²)		
	PLATE : fb (t/cm ²)		
	PLATE : fs (t/cm ²)		
REMARKS			
NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm)			
e --- ECCENTRICITY			
σ c --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e			
P --- UP-LIFT FORCE FOR COLUMN BASE (t)			
α --- COEFFICIENT FOR BENDING MOMENT OF SLAB			

DECISION OF COLUMN BASE (8)
 [柱脚の断面算定]

LOCATION		G-101, 208	G-103 (102, 105, 106 203, 204, 206, 207)		
COLUMN SIZE		BH-900x350 x 28 x 90		BH-900x350 x 28 x 90	
DIRECTION		X	Y	X	Y
PERMANENT CONDITIONS	M (tm)	7.8		7.5	
	N (t)	196.2		386.4	
	Q (t)	9.6		-15.0	
TEMPORARY CONDITIONS	M (tm)	35.1	30.2	16.0	5.6
	N (t)	274.4	99.9	450.0	339.0
	Q (t)	70.1	60.3	32.0	11.1
FIGURE		Same as A-105			
BASE PLATE (LxBxt)		1300x700x29		1300x700x45	
CAP PLATE (THICK.)		22		22	
RIB PLATE (HxBxt)		16		16	
WING PLATE (HxBxt)		22		22	
ANCHOR BOLT (n-Dφ)		8-30φ		8-30φ	
SHEAR KEY (HxBxt)					
CONC.	e=M/N (cm)			$\frac{7.5 \times 10^2}{386.4} = 1.94 < \frac{70}{6}$	
	$\sigma_c < f_c$ (kg/cm ²)			$\frac{386.4 \times 10^3}{130 \times 70} \times (1 + \frac{6 \times 1.94}{70}) = 49.5 < 70$	
ANCHOR BOLT	P/(n*A) < ft (t/cm ²)	N/bbt = 0.017 M/bbt ² = 0.0016		R _t = 0.0%	
BASE PLATE	M = α wlx ² (tcm) t > √(6XM/FB) (cm)			0.085 × 0.0495 × 35 ² = 5.15 √(6 × 5.15 / 1.7) = 4.26 → 4.5	
CAP PLATE	M = α wlx ² (tcm) t > √(6XM/FB) (cm)				
RIB PLATE	τ = σ cA/(txH) (t/cm ²)			$\frac{0.0495 \times 35 \times 25}{1.6 \times 50} = 0.76 < 0.9$	
	H/t			$\frac{50}{1.6} = 31.3 < 71$	
WING PLATE	WELDING τ < fs (t/cm ²)			$\frac{0.0495 \times 37.5 \times 17.5}{1.2 \times 50} = 0.54 < 0.9$	
	H/t			$\frac{50}{1.2} = 41.7 < 71$	
ALLOWABLE STRESS	CONC. : fc (t/cm ²)				
	A. BOLT : ft (t/cm ²)				
REMARKS	PLATE : fb (t/cm ²)				
	PLATE : fs (t/cm ²)				
NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm) e --- ECCENTRICITY σ c --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e P --- UP-LIFT FORCE FOR COLUMN BASE (t) α --- COEFFICIENT FOR BENDING MOMENT OF SLAB					

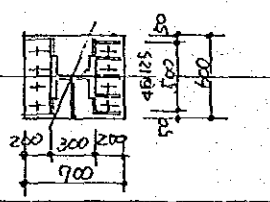
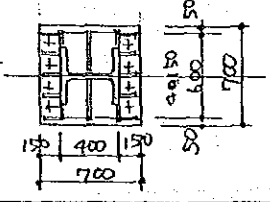
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DECISION OF COLUMN BASE (9)
[柱脚の断面算定]

LOCATION		G-104, 205	G-108 (107, 201, 202)		
COLUMN SIZE		BH-900x350x28x40		BH-900x400x28x40	
DIRECTION		X	Y	X	Y
PERMANENT CONDITIONS	M (tm)		0.0	15.2	
	N (t)		378.9	481.7	
	Q (t)		0.0	30.3	
TEMPORARY CONDITIONS	M (tm)		0.2	37.3	7.6
	N (t)		464.7	552.1	438.9
	Q (t)		0.4	74.5	15.2
FIGURE		Same as G-103			
BASE PLATE (LxBxt)		1300x700x45		1300x700x50	
CAP PLATE (THICK.)		22		22	
RIB PLATE (HxBxt)		16		16	
WING PLATE (HxBxt)		22		22	
ANCHOR BOLT (n-Dφ)		8-30φ		8-35φ	
SHEAR KEY (HxBxt)					
CONC.	e=M/N (cm)	$\frac{15.2 \times 10^2}{481.7} = 3.16 < \frac{70}{6}$			
	$\sigma_c < f_c$ (kg/cm ²)	$\frac{481.7 \times 10^3}{130 \times 70} \left(1 + \frac{6 \times 3.16}{70}\right) = 67.3 < 70$			
ANCHOR BOLT	$P/(n \cdot A) < f_t$ (t/cm ²)	—			
BASE PLATE	$M = \alpha w l x^2$ (tcm)	$0.065 \times 0.067 \times 35^2 = 2.0$			
	$t > \sqrt{6XM/f_b}$ (cm)	$\sqrt{6 \times 2.0 / 1.70} = 4.97 \rightarrow 5.0$			
CAP PLATE	$M = \alpha w l x^2$ (tcm)				
	$t > \sqrt{6XM/f_b}$ (cm)				
RIB PLATE	$\tau = \sigma c A / (t x H)$ (t/cm ²)	$\frac{0.067 \times 35 \times 35}{2.2 \times 50} = 0.75 < 0.9$			
	H/t	$\frac{35}{2.2} = 15.9 < 16$			
	WELDING $\tau < f_s$ (t/cm ²)				
WING PLATE	$\tau = \sigma c A / (t x H)$ (t/cm ²)	$\frac{0.067 \times 15 \times 37.5}{1.2 \times 50}$			
	H/t	$\frac{15}{1.2} = 12.5 < 16$			
	WELDING $\tau < f_s$ (t/cm ²)				
ALLOWABLE STRESS	CONC. : f_c (t/cm ²)				
	A. BOLT : f_t (t/cm ²)				
	PLATE : f_b (t/cm ²)				
	PLATE : f_s (t/cm ²)				
REMARKS					
NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm)					
e --- ECCENTRICITY					
σ_c --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e					
P --- UP-LIFT FORCE FOR COLUMN BASE (t)					
α --- COEFFICIENT FOR BENDING MOMENT OF SLAB					

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DECISION OF COLUMN BASE (10)
[柱脚の断面算定]

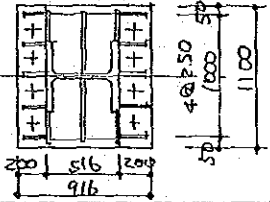
LOCATION		H-101, 208	H-102, 207
COLUMN SIZE		H-300 ² × 10 × 15	
DIRECTION		※ Y	※ X
PERMANENT CONDITIONS	M (tm)	2.9	0.0
	N (t)	81.2	209.0
	Q (t)	9.7	0.0
TEMPORARY CONDITIONS	M (tm)	14.0	13.1
	N (t)	205.3	-44.8
	Q (t)	53.4	43.7
FIGURE			
BASE PLATE (LxBxt)		700 × 600 × 22	700 × 700 × 25
CAP PLATE (THICK.)		19	—
RIB PLATE (HxBxt)		16	19
WING PLATE (HxBxt)		16	19
ANCHOR BOLT (n-Dφ)		8-30φ	8-25φ
CONC.	e=M/N (cm)	$\frac{16.0 \times 10^2}{205.3} = 7.8 < \frac{70}{6}$	
	$\sigma c < f_c$ (kg/cm ²)	$\frac{205.3 \times 10^3}{70 \times 60} \times (1 + \frac{6 \times 7.8}{70}) = 81.6 < 140$	$\frac{209.0 \times 10^3}{70 \times 70} = 42.7 < 70$
ANCHOR BOLT	P/(n*A) < ft (t/cm ²)	$N/bdf = 0.0059, P_c = 0.67$ $M/bd^2 = 0.0025, A_t = 70 \times 60 \times P_c/4 = 6.3 \text{ cm}^2$	
BASE PLATE	M=α wlx ² (tcm)	$0.085 \times 0.082 \times 12.5^2 = 1.09$	$0.085 \times 0.043 \times 20^2 = 1.46$
	t > √(6XM/fb) (cm)	$\sqrt{6 \times 1.09 / 2.77} = 1.5 \rightarrow 19$	$\sqrt{6 \times 1.46 / 1.85} = 2.10 \rightarrow 2.5$
CAP PLATE	M=α wlx ² (tcm)	$0.085 \times (2 \times 6.3 \times 1.8 / (2.5 \times 20)) \times 12.5^2 = 1.20$	
	t > √(6XM/fb) (cm)	$\sqrt{6 \times 1.2 / 2.77} = 1.6 \rightarrow 19$	
RIB PLATE	τ = σ cA/(txH) (t/cm ²)	$\frac{0.082 \times 20 \times 12.5}{1.2 \times 30} = 0.57 < 1.35$	$\frac{0.043 \times 20 \times 25}{1.9 \times 30} = 0.53 < 0.9$
	H/t		
WING PLATE	τ = σ cA/(txH) (t/cm ²)	$\frac{0.082 \times 15 \times 27.5}{1.2 \times 30} = 0.94 < 1.35$	$\frac{0.043 \times 12.5 \times 25}{1.9 \times 30} = 0.24 < 0.9$
	H/t		
ALLOWABLE STRESS	CONC. : fc (t/cm ²)		
	A. BOLT : ft (t/cm ²)		
	PLATE : fb (t/cm ²)		
	PLATE : fs (t/cm ²)		
REMARKS			
NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm) e --- ECCENTRICITY σ c --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e P --- UP-LIFT FORCE FOR COLUMN BASE (t) α --- COEFFICIENT FOR BENDING MOMENT OF SLAB			

DECISION OF COLUMN BASE (1)
 [柱脚の断面算定]

LOCATION		H-103 (105, 204, 206)		H-203 (104, 106, 205)	
COLUMN SIZE		H-450 ² x 16 x 25		H-516 x 500 x 22 x 40	
DIRECTION		X	FX	Y	Y
PERMANENT CONDITIONS	M (tm)	0.0		4.5	
	N (t)	284.4		375.3	
	Q (t)	0.0		15.0	
TEMPORARY CONDITIONS	M (tm)	6.2	6.2	20.9	11.9
	N (t)	408.5	155.6	499.1	247.4
	Q (t)	20.6	20.6	69.8	39.8
FIGURE					
BASE PLATE (LxBxt)		750 x 700 x 30		916 x 700 x 40	
CAP PLATE (THICK.)		19		22	
RIB PLATE (HxBxt)		19		22	
WING PLATE (HxBxt)		19		22	
ANCHOR BOLT (n-Dφ)		8-25φ		8-25φ	
SHEAR KEY (HxBxt)					
CONC.	e=M/N (cm)			$\frac{4.5 \times 10^2}{375.3} = 1.2 < \frac{916}{6}$	
	$\sigma c < f_c$ (kg/cm ²)			$\frac{375.3 \times 10^3}{916 \times 70} \times \left(1 + \frac{6 \times 1.2}{91.6}\right) = 63.1 < 70$	
ANCHOR BOLT	P/(n*A) < ft (t/cm ²)				
BASE PLATE	M=α wlx ² (tcm)			0.085 x 0.063 x 25.6 ² = 3.56	
	t > √(6XM/Fb) (cm)			√(6 x 3.56 / 1.85) = 3.40 → 4.0	
CAP PLATE	M=α wlx ² (tcm)				
	t > √(6XM/Fb) (cm)				
RIB PLATE	$\tau = \sigma cA / (txH)$ (t/cm ²)			$\frac{0.063 \times 25.8 \times 35}{22 \times 50} = 0.52 < 0.9$	
	H/t				
WING PLATE	WELDING $\tau < f_s$ (t/cm ²)				
	$\tau = \sigma cA / (txH)$ (t/cm ²)			$\frac{0.063 \times 12.5 \times 32.9}{22 \times 50} = 0.23 < 0.9$	
WING PLATE	H/t				
	WELDING $\tau < f_s$ (t/cm ²)				
ALLOWABLE STRESS	CONC. : f _c (t/cm ²)				
	A. BOLT : f _t (t/cm ²)				
	PLATE : f _b (t/cm ²)				
	PLATE : f _s (t/cm ²)				
REMARKS					

NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm)
 e --- ECCENTRICITY
 σc --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e
 P --- UP-LIFT FORCE FOR COLUMN BASE (t)
 α --- COEFFICIENT FOR BENDING MOMENT OF SLAB

DECISION OF COLUMN BASE (12)
 [柱脚の断面算定]

LOCATION		H-108 (107, 201, 202)	K-106, 203
COLUMN SIZE		H-516x500 x22x90	H-350 ² x12 x19
DIRECTION		X	≠ X
PERMANENT CONDITIONS			≠ Y
	M (tm)	15.2	4.5
	N (t)	599.5	179.1
	Q (t)	50.8	15.0
TEMPORARY CONDITIONS			Y
	M (tm)	34.5	4.0
	N (t)	819.5	372.7
	Q (t)	114.9	13.3
FIGURE			Same as B-107
BASE PLATE (LxBxt)		916x1100x30	750x900x25
CAP PLATE (THICK.)		—	—
RIB PLATE (HxBxt)		22	16
WING PLATE (HxBxt)		22	—
ANCHOR BOLT (n-Dφ)		8-25φ	4-25φ
SHEAR KEY (HxBxt)		—	—
CONC.	e=M/N (cm)	$\frac{15.2 \times 10^2}{599.5} = 2.56 < \frac{110}{6}$	
	$\sigma c < f_c$ (kg/cm ²)	$\frac{599.5 \times 10^3}{91.6 \times 110} \left(1 + \frac{6 \times 2.56}{110}\right) = 67.2 < 70$	
ANCHOR BOLT	P/(n*A) < ft (t/cm ²)		N/bDf = 0.004, p _t = 0.2d ₀ M/bD ² f = 0.0028, a _t = 75x40P _t /2 = 3mm ²
BASE PLATE	M = α wlx ² (tcm)	0.085 x 0.067 x 25.8 ² = 3.8	
	t > √(6XM/FB) (cm)	√(6 x 3.8 / 1.85) = 3.5 → 3.8	
CAP PLATE	M = α wlx ² (tcm)		
	t > √(6XM/FB) (cm)		
RIB PLATE	τ = σ cA/(txH) (t/cm ²)	$\frac{0.067 \times 25.8 \times 55}{22 \times 50} = 0.86 < 0.9$	
	H/t		
	WELDING		
	τ < fs (t/cm ²)		
WING PLATE	τ = σ cA/(txH) (t/cm ²)	$\frac{0.067 \times 30 \times 32.9}{2.2 \times 50} = 0.6 < 0.9$	
	H/t		
	WELDING		
	τ < fs (t/cm ²)		
ALLOWABLE STRESS	CONC. : fc (t/cm ²)		
	A. BOLT : ft (t/cm ²)		
	PLATE : fb (t/cm ²)		
	PLATE : fs (t/cm ²)		
REMARKS			
NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm)			
e --- ECCENTRICITY			
σ c --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e			
P --- UP-LIFT FORCE FOR COLUMN BASE (t)			
α --- COEFFICIENT FOR BENDING MOMENT OF SLAB			

DECISION OF COLUMN BASE (13)

[柱脚の断面算定]

LOCATION		K-107, 202	K-108, 201
COLUMN SIZE		H-450 ² x 16 x 25	H-450 ² x 16 x 25
DIRECTION		X	X
PERMANENT CONDITIONS	M (tm)	0.0	
	N (t)	288.5	307.0
	Q (t)	0.0	0.0
TEMPORARY CONDITIONS	M (tm)	6.2	6.2
	N (t)	386.2	190.2
	Q (t)	20.6	20.6
FIGURE		Same as H-103	Same as H-103
BASE PLATE (LxBxt)		750x700x30	750x700x30
CAP PLATE (THICK.)		—	—
RIB PLATE (HxBxt)		19	19
WING PLATE (HxBxt)		19	19
ANCHOR BOLT (n-Dφ)		8-25φ	8-25φ
SHEAR KEY (HxBxt)			
CONC.	e=M/N (cm)		
	$\sigma c < f_c$ (kg/cm ²)		$\frac{307 \times 10^3}{75 \times 70} = 57.9 < 90.0$
ANCHOR BOLT	$P/(n \cdot A) < f_t$ (t/cm ²)		
BASE PLATE	$M = \alpha w l x^2$ (tcm)		$0.085 \times 0.058 \times 225^2 = 2.50$
	$t > \sqrt{6xM/f_b}$ (cm)		$\sqrt{6 \times 2.5 / 1.85} = 2.85 \rightarrow 3.0$
CAP PLATE	$M = \alpha w l x^2$ (tcm)		
	$t > \sqrt{6xM/f_b}$ (cm)		
RIB PLATE	$\tau = \sigma c A / (t x H)$ (t/cm ²)		$\frac{0.058 \times 22.5 \times 25}{1.9 \times 30} = 0.8 < 0.9$
	H/t		
WING PLATE	WELDING		
	$\tau < f_s$ (t/cm ²)		
WING PLATE	$\tau = \sigma c A / (t x H)$ (t/cm ²)		$\frac{0.058 \times 22.5 \times 26.3}{1.9 \times 30} = 0.33 < 0.9$
	H/t		
ALLOWABLE STRESS	CONC. : f_c (t/cm ²)		
	A. BOLT : f_t (t/cm ²)		
STRESS	PLATE : f_b (t/cm ²)		
	PLATE : f_s (t/cm ²)		
REMARKS			
NOTATION: L, H, B, t, D --- LENGTH, HEIGHT, WIDTH, THICKNESS, DIAMETER (mm)			
e --- ECCENTRICITY			
σc --- COMPRESSIVE STRESS, N/(B*L) OR OTHER EQUATION DUE TO e			
P --- UP-LIFT FORCE FOR COLUMN BASE (t)			
α --- COEFFICIENT FOR BENDING MOMENT OF SLAB			